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Metallic materials — Charpy pendulum impact test —

Part 2: **Verification of testing machines**

Matériaux métalliques — Essai de flexion par choc sur éprouvette Charpy —

Partie 2: Vérification des machines d'essai (mouton-pendule)

Reference number ISO 148-2:2016(E)

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Foreword

ISO (the International Organization for Standardization) is a worldwide federation of national standards bodies (ISO member bodies). The work of preparing International Standards is normally carried out through ISO technical committees. Each member body interested in a subject for which a technical committee has been established has the right to be represented on that committee. International organizations, governmental and non-governmental, in liaison with ISO, also take part in the work. ISO collaborates closely with the International Electrotechnical Commission (IEC) on all matters of electrotechnical standardization.

The procedures used to develop this document and those intended for its further maintenance are described in the ISO/IEC Directives, Part 1. In particular the different approval criteria needed for the different types of ISO documents should be noted. This document was drafted in accordance with the editorial rules of the ISO/IEC Directives, Part 2 (see [www.iso.org/directives\)](http://www.iso.org/directives).

Attention is drawn to the possibility that some of the elements of this document may be the subject of patent rights. ISO shall not be held responsible for identifying any or all such patent rights. Details of any patent rights identified during the development of the document will be in the Introduction and/or on the ISO list of patent declarations received (see www.iso.org/patents).

Any trade name used in this document is information given for the convenience of users and does not constitute an endorsement.

For an explanation on the meaning of ISO specific terms and expressions related to conformity assessment, as well as information about ISO's adherence to the World Trade Organization (WTO) principles in the Technical Barriers to Trade (TBT) see the following URL: www.iso.org/iso/foreword.html.

The committee responsible for this document is ISO/TC 164, *Mechanical testing of metals*, Subcommittee SC 4, *Toughness testing — Fracture (F), Pendulum (P), Tear (T)*.

This third edition cancels and replaces the second edition (ISO 148-2:2008), which has been technically revised.

ISO 148 consists of the following parts, under the general title *Metallic materials — Charpy pendulum impact test*:

- *Part 1: Test method*
- *Part 2: Verification of testing machines*
- *Part 3: Preparation and characterization of Charpy V-notch test pieces for indirect verification of pendulum impact machines*

Introduction

The suitability of a pendulum impact testing machine for acceptance testing of metallic materials has usually been based on a calibration of its scale and verification of compliance with specified dimensions, such as the shape and spacing of the anvils supporting the specimen. The scale calibration is commonly verified by measuring the mass of the pendulum and its elevation at various scale readings. This procedure for evaluation of machines had the distinct advantage of requiring only measurements of quantities that could be traced to national standards. The objective nature of these traceable measurements minimized the necessity for arbitration regarding the suitability of the machines for material acceptance tests.

However, sometimes two machines that had been evaluated by the direct-verification procedures described above, and which met all dimensional requirements, were found to give significantly different impact values when testing test pieces of the same material.

This difference was commercially important when values obtained using one machine met the material specification, while the values obtained using the other machine did not. To avoid such disagreements, some purchasers of materials added the requirement that all pendulum impact testing machines used for acceptance testing of material sold to them are to be indirectly verified by testing reference test pieces supplied by them. A machine was considered acceptable only if the values obtained using the machine agreed, within specified limits, with the value furnished with the reference test pieces.

This part of ISO 148 describes both the original direct verification and the indirect verification procedures.

Metallic materials — Charpy pendulum impact test —

Part 2: **Verification of testing machines**

1 Scope

This part of ISO 148 covers the verification of pendulum-type impact testing machines, in terms of their constructional elements, their overall performance and the accuracy of the results they produce. It is applicable to machines with 2 mm or 8 mm strikers used for pendulum impact tests carried out, for instance, in accordance with ISO 148-1.

It can be applied to pendulum impact testing machines of various capacities and of different design.

Impact machines used for industrial, general or research laboratory testing of metallic materials in accordance with this part of ISO 148 are referred to as industrial machines. Those with more stringent requirements are referred to as reference machines. Specifications for the verification of reference machines are found in ISO 148-3.

This part of ISO 148 describes two methods of verification.

- a) The direct method, which is static in nature, involves measurement of the critical parts of the machine to ensure that it meets the requirements of this part of ISO 148. Instruments used for the verification and calibration are traceable to national or international standards.
- b) The indirect method, which is dynamic in nature, uses reference test pieces to verify points on the measuring scale for absorbed energy. The requirements for the reference test pieces are found in ISO 148-3.

A pendulum impact testing machine is not in compliance with this part of ISO 148 until it has been verified by both the direct and indirect methods and meets the requirements of [Clause](#page-11-1) 6 and [Clause](#page-18-1) 7.

This part of ISO 148 describes how to assess the different components of the total energy absorbed in fracturing a test piece. This total absorbed energy consists of

- the energy needed to fracture the test piece itself, and
- the internal energy losses of the pendulum impact testing machine performing the first half-cycle swing from the initial position.
- NOTE Internal energy losses are due to the following:
- air resistance, friction of the bearings of the rotation axis and of the indicating pointer of the pendulum which can be determined by the direct method (see [6.4.5\)](#page-14-0);
- shock of the foundation, vibration of the frame and pendulum for which no suitable measuring methods and apparatus have been developed.

2 Normative references

The following documents, in whole or in part, are normatively referenced in this document and are indispensable for its application. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

ISO 148-1, *Metallic materials — Charpy pendulum impact test — Part 1: Test method*

ISO 148-3, *Metallic materials — Charpy pendulum impact test — Part 3: Preparation and characterization of Charpy V-notch test pieces for indirect verification of pendulum impact machines*

3 Terms and definitions

For the purposes of this document, the following terms and definitions apply.

3.1 Definitions pertaining to the machine

3.1.1

anvil

portion of the machine that serves to properly position the test piece for impact with respect to the striker and the test piece supports, and supports the test piece under the force of the strike

3.1.2

base

part of the framework of the machine located below the horizontal plane of the supports

3.1.3

centre of percussion

point in a body at which, on striking a blow, the percussive action is the same as if the whole mass of the body were concentrated at the point

Note 1 to entry: When a simple pendulum delivers a blow along a horizontal line passing through the centre of percussion, there is no resulting horizontal reaction at the axis of rotation.

Note 2 to entry: See [Figure](#page-25-0) 4.

3.1.4

centre of strike

point on the striking edge of the pendulum at which, in the free hanging position of the pendulum, the vertical edge of the striker meets the upper horizontal plane of a test piece of half standard thickness (i.e. 5 mm) or equivalent gauge bar resting on the test piece supports

Note 1 to entry: See [Figure](#page-25-0) 4.

3.1.5

industrial machine

pendulum impact machine used for industrial, general or most research-laboratory testing of metallic materials

Note 1 to entry: Industrial machines are not used to establish reference values, unless they also meet the requirements of a reference pendulum (see ISO 148-3).

Note 2 to entry: Industrial machines are verified using the procedures described in this part of ISO 148.

3.1.6

reference machine

pendulum impact testing machine used to determine certified values for batches of *reference test pieces* ([3.3.4\)](#page-9-1)

Note 1 to entry: Reference machines are verified using the procedures described in ISO 148-3.

3.1.7

striker

portion of the pendulum that contacts the test piece

Note 1 to entry: The edge that actually contacts the test piece has a radius of 2 mm (the 2 mm striker) or a radius of 8 mm (the 8 mm striker).

Note 2 to entry: See [Figure](#page-22-0) 2.

3.1.8

test piece supports

portion of the machine that serves to properly position the test piece for impact with respect to the *centre of percussion* [\(3.1.3\)](#page-7-1) of the pendulum, the *striker* ([3.1.7\)](#page-7-2) and the *anvils* ([3.1.1](#page-7-3))

Note 1 to entry: See [Figure](#page-22-0) 2 and [Figure](#page-24-0) 3.

3.2 Definitions pertaining to energy

3.2.1

total absorbed energy

*K*T

total absorbed energy required to break a test piece with a pendulum impact testing machine, which is not corrected for any losses of energy

Note 1 to entry: It is equal to the difference in the *potential energy* [\(3.2.2\)](#page-8-1) from the starting position of the pendulum to the end of the first half swing during which the test piece is broken (see [6.3](#page-12-1)).

3.2.2 initial potential energy potential energy

*K*P

potential energy of the pendulum hammer prior to its release for the impact test, as determined by direct verification

Note 1 to entry: See [6.4.2.](#page-13-1)

3.2.3 absorbed energy

K

energy required to break a test piece with a pendulum impact testing machine, after correction for friction as defined in [6.4.5](#page-14-0)

Note 1 to entry: The letter V or U is used to indicate the notch geometry, which is *KV* or *KU*. The number 2 or 8 is used as a subscript to indicate striker radius, for example *KV*2.

3.2.4 calculated energy

*K*calc

energy calculated from values of angle, length and force measured during direct verification

3.2.5

nominal initial potential energy nominal energy

 $K_{\rm N}$

energy assigned by the manufacturer of the pendulum impact testing machine

3.2.6

indicated absorbed energy

 K_S

energy indicated by the display/dial of the testing machine, which may or may not need to be corrected for friction and air resistance to determine the *absorbed energy*, *K* [\(3.2.3\)](#page-8-2)

3.2.7

reference absorbed energy

*K*R

certified value of *absorbed energy* [\(3.2.3](#page-8-2)) assigned to the *reference test pieces* [\(3.3.4](#page-9-1)) used to verify the performance of pendulum impact machines

3.3 Definitions pertaining to test pieces

3.3.1 width

W

distance between the notched face and the opposite face

Note 1 to entry: In previous versions of the ISO 148 series (prior to 2016), the distance between the notched face and the opposite face was specified as "height". Changing this dimension to "width" makes ISO 148-2 consistent with the terminology used in other ISO fracture standards.

3.3.2

thickness

B

dimension perpendicular to the *width* [\(3.3.1\)](#page-9-2) and parallel to the notch

Note 1 to entry: In previous versions of the ISO 148 series (prior to 2016), the dimension perpendicular to the width that is parallel to the notch was specified as "width". Changing this dimension to "thickness" makes ISO 148-2 consistent with the terminology used in other ISO fracture standards.

3.3.3 length

L

largest dimension perpendicular to the notch

3.3.4

reference test piece

impact test piece used to verify the suitability of a pendulum impact testing machine by comparing the *indicated absorbed energy* [\(3.2.3](#page-8-2)) measured by that machine with the *reference absorbed energy* ([3.2.7](#page-8-3)) associated with the test pieces

Note 1 to entry: Reference test pieces are prepared in accordance with ISO 148-3.

4 Symbols and abbreviated terms

Table 1 — Symbols/abbreviated terms and their designations and units

Table 1 *(continued)*

Table 1 *(continued)*

Symbol/ abbreviated terma	Unit	Designation
v_B		Degrees of freedom corresponding to $u(B_V)$
v_{V}		Degrees of freedom corresponding to uy
$v_{\rm RM}$		Degrees of freedom corresponding to u_{RM}
See Figure 4. a		

5 Testing machine

A pendulum impact testing machine consists of the following parts (see [Figure](#page-24-0) 1 to Figure 3):

- a) foundation/installation;
- b) machine framework: the structure supporting the pendulum, excluding the foundation;
- c) pendulum, including the hammer;
- d) anvils and supports (see $Figure 2$ $Figure 2$ and $Figure 3$);
- e) indicating equipment for the absorbed energy (e.g. scale and friction pointer or electronic readout device).

6 Direct verification

6.1 General

Direct verification of the machine involves the inspection of the items a) to e) listed in [Clause](#page-11-2) 5.

Uncertainty estimates are required under [Clause](#page-11-1) 6 for direct verification measurements to harmonize the accuracy of the applied verification procedures. Uncertainty estimates required in [Clause](#page-11-1) 6 are not related to product standards or material property databases in any way.

The uncertainty of dial gauges, micrometres, callipers, and other commercial instrumentation used for the direct verification measurements shall be estimated once, by the producer.

Uncertainty of a method to measure a direct verification parameter is assessed as part of the method validation. Once method validation is completed, the uncertainty can be routinely used (provided the same method is followed, the same instrumentation is used, and the operators are trained).

6.2 Foundation/installation

6.2.1 The foundation to which the machine is fixed and the method(s) of fixing the machine to the foundation are of the utmost importance.

6.2.2 Inspection of the machine foundation can usually not be made once the machine has been installed; thus, documentation made at the time of installation shall be produced to provide assurance that the mass of the foundation is not less than 40 times that of the pendulum.

6.2.3 Inspection of the installed machine shall consist of the following.

a) Ensuring that the bolts are torqued to the value specified by the machine manufacturer. The torque value shall be noted in the document provided by the manufacturer of the machine (see [6.2.1](#page-11-3)). If other mounting arrangements are used or selected by an end user, equivalency shall be demonstrated.

b) Ensuring that the machine is not subject to external vibrations transmitted through the foundation at the time of the impact test.

NOTE This can be accomplished, for example, by placing a small container of water on any convenient location on the machine framework. The absence of ripples on the water surface during an impact test indicates that this requirement has been met.

6.3 Machine framework

6.3.1 Inspection of the machine framework (see [Figure](#page-21-0) 1) shall consist of determining the following items:

- a) free position of the pendulum;
- b) location of the pendulum in relation to the supports;
- c) transverse and radial play of the pendulum bearings;
- d) clearance between the hammer and the framework.

Machines manufactured after 1998 shall have a reference plane from which measurements can be made.

[Annex](#page-37-1) C is provided for information.

6.3.2 The axis of rotation of the pendulum shall be parallel to the reference plane to within 2/1 000. This shall be certified by the manufacturer.

6.3.3 The machine shall be installed so that the reference plane is horizontal to within 2/1 000.

For pendulum impact testing machines without a reference plane, the axis of rotation shall be established to be horizontal to within 4/1 000 directly or a reference plane shall be established from which the horizontality of the axis of rotation can be verified as described above.

6.3.4 When hanging free, the pendulum shall hang so that the striking edge is within 2,5 mm of the position where it would just touch the test specimen.

NOTE This condition can be determined using a gauge in the form of a bar that is approximately 55 mm in length and of rectangular section 7,5 mm by 12,5 mm (see [Figure](#page-24-0) 3).

6.3.5 The plane of swing of the pendulum shall be $90.0^{\circ} \pm 0.1^{\circ}$ to the axis of rotation ($u < 0.05^{\circ}$).

6.3.6 The striker shall make contact over the full thickness of the test piece.

One method of verifying this is to use a test piece having dimensions of $55 \text{ mm} \times 10 \text{ mm} \times 10 \text{ mm}$ that is tightly wrapped in thin paper (e.g. by means of adhesive tape) and a striking edge that is tightly wrapped in carbon paper with the carbon side outermost (i.e. not facing the striker). From its position of equilibrium, the pendulum is raised a few degrees, released so that it contacts the test piece, and prevented from contacting the test piece a second time. The mark made by the carbon paper on the paper covering the test piece should extend completely across the paper. This verification can be performed concurrently with that of checking the angle of contact between the striker and the test piece (see [6.4.8\)](#page-16-1).

6.3.7 The pendulum shall be located so that the centre of the striker and the centre of the gap between the anvils are coincident to within 0.5 mm $(u < 0.1$ mm).

6.3.8 Axial play in the pendulum bearings shall not exceed 0,25 mm (*u* < 0,05 mm) measured at the centre-of-rotation under a transverse force of approximately 4 % of the effective weight of the pendulum, F_g [see [Figure](#page-25-0) 4 b]], applied at the centre of strike.

6.3.9 Radial play of the shaft in the pendulum bearings shall not exceed 0,08 mm (*u* < 0,02 mm) when a force of 150 N ± 10 N is applied at a distance *l* perpendicular to the plane of swing of the pendulum.

NOTE The radial play can be measured, for example, by a dial gauge mounted on the machine frame at the bearing housing in order to indicate movement at the end of the shaft (in the bearings) when a force of about 150 N is applied to the pendulum perpendicularly to the plane of the swing.

6.3.10 It is recommended that the mass of the base of the machine framework be at least 12 times that of the pendulum.

6.4 Pendulum

6.4.1 The verification of the pendulum (including striker) shall consist of determining the following quantities:

- a) potential energy, $K_{\rm P}$;
- b) error in the indicated absorbed energy, *K*_S;
- c) velocity of the pendulum at the instant of impact;
- d) energy absorbed by friction;
- e) position of the centre of percussion (i.e. distance from the centre of percussion to the axis of rotation);
- f) radius of the striking edge of the striker;
- g) angle between the line of contact of the striker and the horizontal axis of the test piece.

6.4.2 The potential energy, K_{P} , shall not differ from the nominal energy, K_{N} , by more than ± 1 %. The potential energy, K_{P} , shall be determined as follows.

The moment of the pendulum is determined by supporting the pendulum at a chosen distance, *l*₂, from the axis of rotation by means of a knife edge on a balance or dynamometer in such a manner that the line through the axis of rotation that joins the centre of gravity of the pendulum is horizontal within 15/1 000 [see [Figure](#page-25-0) 4 a)] (*u* < 5/1 000).

The force, *F*, and the length, *l*2, shall each be determined to an accuracy of ±0,2 %. The moment, *M*, is the product of $F \cdot l_2$.

NOTE Length *l*2 can be equal to length *l*.

The angle of fall, α , shall be measured to an accuracy of $\pm 0.2^{\circ}$; this angle can be greater than 90°.

The potential energy, K_{P} , is then calculated by **[Formula](#page-13-2)** (1):

$$
K_{\rm P} = M \left(1 - \cos \alpha \right) \tag{1}
$$

6.4.3 The graduation marks on the scale corresponding approximately to values of absorbed energy of 0 %, 10 %, 20 %, 30 %, 50 % and 80 % of the nominal energy shall be verified.

For each of these graduation marks, the pendulum shall be supported so that the graduation mark is indicated by the pointer, and the angle of rise, β , then determined to $\pm 0.2^{\circ}$. The calculated energy is given by [Formula](#page-13-3) (2):

$$
K_{\text{calc}} = M \left(\cos \beta - \cos \alpha \right) \tag{2}
$$

NOTE 1 The measurement uncertainty of *l₂*, *F* and *β*, as specified, yields a mean total measurement uncertainty of K_{calc} of approximately ± 0.3 % of the full-scale value.

The difference between the indicated absorbed energy, K_S , and the calculated energy from the measured values shall not be greater than ± 1 % of the energy reading or ± 0.5 % of the nominal energy, K_N . In each case, the greater value is permitted, i.e.

$$
\left| \frac{K_{\text{calc}} - K_{\text{S}}}{K_{\text{S}}} \right| \cdot 100 \le 1 \text{ % at between 50 % and 80 % of the nominal energy, } K_{\text{N}} \tag{3}
$$
\n
$$
\left| \frac{K_{\text{calc}} - K_{\text{S}}}{K_{\text{N}}} \right| \cdot 100 \le 0,5 \text{ % at less than 50 % of the nominal energy, } K_{\text{N}} \tag{4}
$$

NOTE 2 Attention is drawn to the fact that the accuracy of the absorbed energy reading is inversely proportional to its value, and this is important when K is small in comparison with K_N .

NOTE 3 For machines with scales and readout devices that are corrected for energy losses, *K*_{calc} should be corrected in order to compare the results properly.

6.4.4 The velocity at impact can be determined from **[Formula](#page-14-1)** (5):

$$
v = \sqrt{2gl(1 - \cos \alpha)}\tag{5}
$$

where

 $\ddot{}$

g is the local acceleration of gravity known to 1 part in 1 000 or better, in m/s^2 .

The velocity at impact shall be 5 m/s to 5,5 m/s (*u* < 0,1 m/s); however, for machines manufactured prior to 1998, any value within the range of 4,3 m/s to 7 m/s is permissible and the value shall be stated in the report.

6.4.5 The energy absorbed by friction includes, but is not limited to, air resistance, bearing friction and the friction of the indicating pointer. These losses shall be estimated as follows.

6.4.5.1 To determine the loss caused by pointer friction, the machine is operated in the normal manner, but without a test piece in position, and the angle of rise, *β*1, or energy reading, *K*1, is noted as indicated by the pointer. A second test is then carried out without resetting the indication pointer and the new angle of rise, *β*2, or energy reading, *K*2, is noted. Thus, the loss due to friction in the indicating pointer during the rise is equal to as given by [Formula](#page-14-2) (6) :

$$
p = M\left(\cos\beta_1 - \cos\beta_2\right) \tag{6}
$$

when the scale is graduated in degrees, or as given by [Formula](#page-14-3) (7):

$$
p = K_1 - K_2 \tag{7}
$$

when the scale is graduated in energy units.

6.4.5.2 Determination of the losses caused by bearing friction and air resistance for one half swing is performed as follows.

After determining β_2 or K_2 in accordance with $6.4.5.1$, the pendulum is put into its initial position. Without resetting the indicating mechanism, release the pendulum without shock and vibration and permit it to swing 10 half swings. After the pendulum starts its eleventh half swing, move the indicating mechanism to about 5 % of the scale-range capacity and record the value as β_3 or K_3 . The losses by bearing friction and air resistance for one half swing are equal to as given by **[Formula](#page-15-0)** (8):

$$
p' = 1/10 M \left(\cos\beta_3 - \cos\beta_2\right) \tag{8}
$$

when the scale is graduated in degrees, or as given by [Formula](#page-15-1) (9):

$$
p' = 1/10(K_3 - K_2)
$$
 (9)

when the scale is graduated in energy units.

NOTE If it is required to take into account these losses in an actual test giving an angle of rise, *β*, the quantity as given by Formula $[10]$ can be subtracted from the value of the absorbed energy.

$$
p_{\beta} = p \frac{\beta}{\beta_1} + p' \frac{\alpha + \beta}{\alpha + \beta_2} \tag{10}
$$

Because $β_1$ and $β_2$ are nearly equal to $α$, [Formula](#page-15-3) (10) can be reduced to Formula (11):

$$
p_{\beta} = p \frac{\beta}{\alpha} + p' \frac{\alpha + \beta}{2\alpha} \tag{11}
$$

For machines graduated in energy units, the value of *β* can be calculated as given in [Formula](#page-15-4) (12):

$$
\beta = \arccos\left[\frac{1}{M(K_{\rm P} - K_{\rm T})}\right]
$$
\n(12)

6.4.5.3 The values of β_1 , β_2 , and β_3 , and the values of K_1 , K_2 , and K_3 shall be the mean values from at least two determinations. The total friction loss $p + p'$, so measured, shall not exceed 0,5 % of the nominal energy, K_N . If it does, and it is not possible to bring the friction loss within the tolerance by reducing the pointer friction, the bearings shall be cleaned or replaced.

6.4.6 The distance from the centre of percussion to the axis of rotation, l_1 , is derived from the period (time of swing) of the pendulum, and it shall be 0.995 *l* \pm 0.005 *l*. The measurement uncertainty of the calculated value of l_1 shall be <0,5 mm.

The distance can be determined by swinging the pendulum through an angle not exceeding 5° and measuring the time, *t*, of a complete swing in seconds.

*l*₁ is derived from [Formula](#page-15-5) (13):

$$
l_1 = \frac{g \cdot t^2}{4\pi^2} \tag{13}
$$

where

g is the acceleration of gravity, taken as equal to 9,81 m/s²;

 π^2 is taken as equal to 9,87.

Therefore, in metres, $l_1 = 0.2485 \cdot t^2$.

The value of *t* shall be determined to within 0,1 %.

With a pendulum having a period of approximately 2 s, this accuracy may be achieved as follows. Determine the time, *T*, of 100 complete swings, three times. An accurate measure of *t* is the average of the three values of *T* divided by 100, provided the quantity (*T*max − *T*min), which represents the repeatability, is not more than 0,2 s.

6.4.7 The dimensions of the striker shall be checked. Either of two types of striker may be used, the 2 mm striker or the 8 mm striker. The values for the radius of curvature and the angle of the tip for both types are shown in [Table](#page-22-1) 3.

The maximum width of that portion of the striker passing between the anvils shall be at least 10 mm but not greater than 18 mm ($u < 0.2 \text{ mm}$).

NOTE An example of a method of verifying the geometry of the striker is to make a replica for examination.

6.4.8 The angle between the line of contact of the striker and the horizontal axis of the test piece shall be 90° ± 2° (see [6.3.6](#page-12-2)) (*u* < 0,2°).

6.4.9 The mechanism for releasing the pendulum from its initial position shall operate freely and permit release of the pendulum without initial impulse, retardation or side vibration.

6.4.10 If the machine has a brake mechanism, means shall be provided to prevent the brake from being accidentally engaged. In addition, there shall be provision to disengage the brake mechanism, for example during the measurement of period and friction losses.

6.4.11 Machines with automated lifting devices shall be constructed so that direct verification can be performed.

6.5 Anvil and supports

6.5.1 Inspection of the anvils and supports should consist of determining the following items (see [Figure](#page-22-0) 2 and [Figure](#page-24-0) 3 and [Table](#page-22-1) 3):

- a) configuration of the supports;
- b) configuration of the anvils;
- c) distance between the anvils;
- d) taper of the anvils;
- e) radius of the anvils;
- f) clearance for the broken test piece to exit the machine.

6.5.2 The planes containing the support surfaces shall be parallel and the distance between them shall not exceed 0,1 mm (*u* < 0,05 mm). Supports shall be such that the axis of the test piece is parallel to the axis of rotation of the pendulum within $3/1000$ ($u < 1/1000$).

6.5.3 The planes containing the anvil surfaces facing the test piece shall be parallel and the distance between them shall not exceed 0,1 mm (*u* < 0,05 mm). The two planes containing the supports and the anvils shall be 90° ± 0,1° relative to each other (*u* < 0,05°). Additional requirements for the configuration of the anvils are given in [Table](#page-22-1) 3.

6.5.4 Sufficient clearance shall be provided to ensure that fractured test pieces are free to leave the machine with a minimum of interference and not rebound into the hammer before the pendulum

completes its swing. No part of the pendulum that passes between the anvils shall exceed 18 mm in width (*u* < 0,2 mm).

Hammers are often of one of two basic designs (see [Figure](#page-21-0) 1). When using the C-type hammer, the broken test pieces will not rebound into the hammer if the clearance at each end of the test piece is greater than 13 mm. If end stops are used to position test pieces, they shall be retracted prior to the instant of impact. When using the U-type hammer, means shall be provided to prevent the broken test pieces from rebounding into the hammer. In most machines using U-type hammers, shrouds (see [Figure](#page-24-0) 3) should be designed and installed with the following requirements:

- a) a thickness of approximately 1,5 mm;
- b) a minimum hardness of 45 HRC;
- c) a radius of at least 1,5 mm at the underside corners;
- d) a position in which the clearance between them and the hammer overhang does not exceed 1,5 mm.

In machines where the opening within the hammer permits a clearance between the ends of the test piece (resting in position ready to test) and the shrouds of at least 13 mm, the requirements of a) and d) need not apply.

6.6 Indicating equipment

6.6.1 The verification of the analogue indicating equipment shall consist of the following examinations:

- a) examination of the scale graduations;
- b) examination of the indicating pointer.

The scale shall be graduated in units of angle or of energy.

The thickness of the graduation marks on the scale shall be uniform and the width of the pointer shall be approximately equal to the width of a graduation mark. The indicating pointer shall permit a reading free from parallax.

The resolution, *r*, of the indicator is obtained from the ratio between the width of the pointer and the centre-to-centre distance between two adjacent scale-graduation marks (scale interval). The recommended ratios are 1:4, 1:5, or 1:10; a spacing of 2,5 mm or greater is required to estimate a tenth of a division on the scale.

The scale interval shall be at most 1 % of the nominal energy and shall permit an estimation of energy in increments of less than or equal to 0,25 % of the nominal energy.

6.6.2 The verification of digital indicating equipment shall ensure that the following requirements are met.

- The scale shall be graduated in units of angle or of energy.
- The resolution of the scale is considered to be one increment of the last active number of the digital indicator provided that the indication does not fluctuate by more than one increment. When the readings fluctuate by more than one increment, the resolution is taken to be equal to half the range of fluctuation.
- The resolution shall be less than or equal to 0.25% of the nominal energy.

7 Indirect verification by use of reference test pieces

7.1 Reference test pieces used

Indirect verification consists of verifying points on the measuring scale using reference test pieces. The following reference test pieces are used:

- a) for comparison between test results obtained with the machine under consideration and test results obtained with a particular reference machine or set of reference machines, or with an SI traceable K_R value obtained in full accordance with ISO 148-1;
- b) to monitor the performance of a machine over a period of time, without reference to any other machine.

7.2 Absorbed energy levels

The indirect verification shall be performed at a minimum of two absorbed energy levels within the range of use of the machine. A set for each energy level shall consist of at least five reference test pieces. The reference test piece absorbed energy levels shall be as close as possible to the upper and lower limits of the range of use, subject to the availability of reference test pieces for these absorbed energy levels.

When more than two reference test piece absorbed energy levels are used, the other level(s) should be distributed as uniformly as possible between the upper and lower limits subject to the availability of reference test pieces.

7.3 Requirements for reference test pieces

Reference test pieces shall be obtained from a reference material producer who has prepared the test pieces as specified in ISO 148-3. Whether or not test pieces that do not break shall be taken into account, the calculation of pendulum bias and repeatability is decided by the reference material producer.

7.4 Limited direct verification

A limited direct verification shall be performed before each indirect verification. This limited direct verification includes the following:

- a) inspection of the machine in accordance with $6.2.3$ a) and of the machine framework in accordance with [6.3.4](#page-12-3) and [6.3.6;](#page-12-2)
- b) inspection (visual at least) of the striker and anvils for excessive wear (see [Table](#page-22-1) 3);
- c) measurement of the distance between the anvils (see [Table](#page-22-1) 3);
- d) when the striker or supports or anvils are changed: measurement of items [6.3.4,](#page-12-3) [6.3.6](#page-12-2), [6.3.7](#page-12-4), [6.4.7](#page-16-2), [6.4.8](#page-16-1), [6.5.2,](#page-16-3) [6.5.3](#page-16-4) and [6.5.4;](#page-16-5)
- e) measurement of the losses due to bearing friction and air resistance;
- f) measurement of the loss due to pointer friction.

7.5 Bias and repeatability

7.5.1 Repeatability

*KV*₁, *KV*₂, ..., *KV*_{n</sup>V} are the absorbed energies of the *n*_V reference test pieces used for the indirect verification at a particular energy level. The repeatability of the machine under the particular controlled

conditions is characterized by *b*, the difference between the highest and lowest of the *n*_V *KV* values, as given by Formula [\(14\)](#page-19-1):

$$
b = KV_{\text{max}} - KV_{\text{min}} \tag{14}
$$

The maximum allowed repeatability values are given in [Table](#page-19-2) 2.

7.5.2 Bias

The bias of the machine under the particular controlled conditions is characterized by the number, as given by Formula [\(15\)](#page-19-3):

$$
B_{\rm V} = \overline{K}V_{\rm V} - K V_{\rm R} \tag{15}
$$

where

$$
\frac{1}{KV_V} = \frac{\sum KV_1 + \dots + KV_{n_V}}{n_V} \tag{16}
$$

Dimensions in joules

The maximum allowed bias values are given in [Table](#page-19-2) 2.

Table 2 — Maximum allowed values for repeatability and bias

8 Frequency of verification

8.1 A full direct verification followed by an indirect verification shall be performed at the time of installation and after moving the machine.

8.2 Indirect verifications, including a limited direct verification, shall be performed at intervals not exceeding 12 months. More frequent indirect verifications may be necessary based on the wear observed.

8.3 When anvils and/or striker are replaced, a direct verification in accordance with clauses describing the affected part(s) shall be performed. An indirect verification shall also be performed.

8.4 If the results of a first indirect verification are unsatisfactory and if limited corrective interventions on the instrument fail to lead to a satisfactory result of the repeated indirect verification, then a full direct verification shall be performed.

9 Verification report

9.1 General

The verification report shall include at least the following information:

- a) reference to this part of ISO 148, i.e. ISO 148-2;
- b) identification of the machine: manufacturer's name, model and serial number;
- c) radius of the striking edge;
- d) name of owner and address of place of installation;
- e) name or mark of organization performing the verification;
- f) date of the verification.

9.2 Direct verification

The following information on the direct verification of the machine shall be included:

- a) nominal energy of the pendulum;
- b) velocity of pendulum at impact;
- c) absorbed energy lost due to air resistance and friction.

9.3 Indirect verification

The following information on indirect verification of the machine shall be included:

- a) identification of the reference test pieces used in the indirect verification, including the reference values and the actual observed absorbed energy values for these test pieces;
- b) results of the indirect verification:
	- 1) repeatability;
	- 2) bias;
	- 3) a statement that the machine does or does not conform to the requirements of this part of ISO 148.

10 Uncertainty

A method for establishing the uncertainty of the indirect verification results is given in [Annex](#page-26-1) A. [Annex](#page-30-1) B gives methods for calculating measurement uncertainty for several of the measurements occurring in the direct verification.

c) Test machine

Key

-
- 2 pendulum bearings 10 foundation
- 3 friction pointer 11 C-type hammer
- 4 pendulum rod 12 edge of striker
- 5 machine framework 13 striker
-
-
-
- 1 scale 9 test-piece supports
	-
	-
	-
	-
- 6 base a Angle of striker.
- 7 anvil **b** Radius of striking edge.
- 8 test piece **c** Axis of rotation.

Figure 1 — Parts of a pendulum-type impact test machine

c) Overview

NOTE See [Table](#page-22-1) 3 for geometrical characteristics.

Numbera	Designation	Dimension			
1	Length of test piece	see ISO 148-1			
2	Width of test piece	see ISO 148-1			
3	Thickness of test piece	see ISO 148-1			
$\overline{4}$	Width of test piece minus depth of notch (ligament)	see ISO 148-1			
5	Angle of notch	see ISO 148-1			
6	Depth of notch	see ISO 148-1			
7	Notch root radius	see ISO 148-1			
See Figure 2. l a					

Table 3 — Geometrical characteristics

Table 3 *(continued)*

Key

-
-
-
-
- 5 width of test piece, *W*
- 1 anvils **1** anvils **1 anville 1 anville 1 anville 1 and 1 and**
- 2 standard size test piece 7 thickness of test piece, *B*
- 3 test piece supports 8 direction of pendulum swing
- 4 shroud 9 centre of strike

Figure 3 — Configuration of test piece supports and anvils of an industrial pendulum-type impact test machine

b) Designation of terms used to determine energy

Key

- 1 centre of percussion
- 2 centre of test piece
- 3 centre of strike of pendulum
- 4 centre of standard-size test piece
- ^a Angle of rise, *β*.
- ^b Angle of fall, *α*.
- c Axis of rotation.

Figure 4 — Determination of the initial potential energy

Annex A

(informative)

Measurement uncertainty of the result of the indirect verification of a Charpy pendulum impact machine

A.1 Overview and general requirements

A.1.1 General

This Annex provides a method for determining the uncertainty associated with the results of indirect verification tests of a Charpy pendulum impact machine. Other methods for assessing the uncertainty of these tests can be developed and are acceptable, if they meet the requirements of the GUM (see Reference [\[1](#page-43-1)]).

This Annex proposes a systematic approach, which leads to estimates for *B_V* (the bias of the machine) and *u*_V (the uncertainty of the overall indirect verification result). The values of these parameters are required for the calculation of the measurement uncertainty of the results of tests performed with the pendulum impact testing machine after the verification, as described in ISO 148-1.

NOTE ISO 148-1:2016, Annex E, also provides a general scheme of the metrological chain used to disseminate absorbed energy scales through indirect verification using reference test pieces.

A.1.2 Uncertainty disclaimer

Measurement uncertainty analysis is useful for identifying major sources of inconsistencies for measured results.

Product standards and material property databases based on this and earlier versions of this part of ISO 148 have an inherent contribution from measurement uncertainty. It is therefore inappropriate to apply further adjustments for measurement uncertainty and thereby risk failing product compliance. For this reason, the estimates of uncertainty derived by following this procedure are for information only, unless specifically instructed otherwise by the customer.

The test conditions and limits defined in this part of ISO 148 should not be adjusted to take account of uncertainties of measurement, unless specifically instructed otherwise by the customer. The estimated measurement uncertainties should not be combined with measured results to assess compliance to product specifications, unless specifically instructed otherwise by the customer. Instead, the indicated tolerances are to be interpreted as acceptance intervals.[[2](#page-43-2)] This approach assumes that measurements are made with a tacitly accepted maximum measurement uncertainty. Where possible, this maximum measurement uncertainty has been specified in the current version of the ISO 148 series. Measurement uncertainties of the measured values should be smaller than the indicated values.

A.2 Contributions to the uncertainty of the indirect verification result

A.2.1 Bias

The primary result of an indirect verification is the estimate of the instrument bias, B_V , as given by Formula (A.1):

$$
B_{\rm V} = \overline{KV}_{\rm V} - KV_{\rm R} \tag{A.1}
$$

where

 KV_V is the mean value of the reference test pieces broken during the indirect verification;

KV^R is the certified *KV* value of the reference test pieces.

The absolute value of B_V should meet the criteria set in [Clause](#page-18-1) 7.

A.2.2 Uncertainty of the bias value

The standard uncertainty of the bias value is equal to the combined standard uncertainties of the two terms in [Formula](#page-27-0) (A.1).

 u_{RM} , the standard uncertainty of the certified reference value, KV_{R} , is calculated from the expanded uncertainty, U_{RM} , indicated on the certificate of the reference test pieces, by dividing U_{RM} by the appropriate coverage factor (also indicated on the certificate).

The uncertainty associated with KV_V is calculated as given by Formula $(A.2)$:

$$
u(\overline{KV}_{V}) = \frac{s_V}{\sqrt{n_V}}
$$
 (A.2)

where

*s*_V is the standard deviation of the results of the *n*_V reference test pieces.

[7.2](#page-18-2) prescribes the use of at least five reference test pieces for the indirect verification.

NOTE [Formula](#page-27-1) $(A.2)$ shows that choosing a larger number n_V can be used to reduce the measurement uncertainty.

Therefore, $u(B_V)$, the standard uncertainty of B_V , is calculated as given by Formula (A.3):

$$
u(B_V) = \sqrt{\left(\frac{s_V}{\sqrt{n_V}}\right)^2 + u_{RM}^2}
$$
 (A.3)

A.3 Determining the combined uncertainty of the indirect verification result, \mathbf{u}_V

As a general rule, bias should be corrected for. However, due to wear of the anvils and hammer parts, it is difficult to obtain a perfectly stable bias value throughout the period between two indirect verifications. This is why the measured bias value is often considered an uncertainty contribution, to be combined with its own uncertainty to obtain the uncertainty of the indirect verification result, *u*_V, as given by **[Formula](#page-28-0)** (A.4):

$$
u_V = \sqrt{u^2 \left(B_V\right) + B_V^2} \tag{A.4}
$$

To correct the absorbed energy values measured with a pendulum impact testing machine, add a term equal to B_V . This requires that the bias value be firmly established and stable. Such a level of knowledge on the performance of a particular pendulum impact testing machine can only be achieved after a series of indirect verification and control chart tests, which should provide the required evidence about the stability of the instrument bias. Therefore, the practice is likely to be limited to reference pendulum impact testing machines.

A.4 Expanding the combined uncertainty

The value of u_V is used in ISO 148-1:2016, Annex E, as one of the contributions to the total measurement uncertainty. To expand a combined standard uncertainty, the degrees of freedom of the respective uncertainty contributions need to be combined into effective degrees of freedom. The degrees of freedom of *u*_V are calculated using the Welch-Satterthwaite approximation, as given by [Formula](#page-28-1) (A.5):

$$
v_V = \frac{u_V^4}{\frac{u^4}{v_B} + \frac{u_{RM}^4}{v_{RM}} + \frac{B_V^4}{v_B}}
$$
(A.5)

The value of v_B equals n_V – 1; the value of v_{BM} is taken from the reference materials' certificate.

The number of verification test pieces is at least five, but the heterogeneity of the samples is not insignificant. This is why the number of effective degrees of freedom is most often not large enough to use a coverage factor of *k* equal to 2. Other values of *k* may be used if interested parties are in agreement.

A.5 Examples of B_V and u_V calculation and reporting

This subclause presents an example of an indirect verification result and its analysis. The indirect verification is executed after a direct verification, using reference test pieces of three different energy levels. The results presented in [Table A.1](#page-28-2) are those obtained on reference test pieces with a certified *KV*R value of 123,8 J, and an expanded uncertainty of 3,4 J, with 30 degrees of freedom (values taken from the RM certificate).

Test results and data from certificates	Calculation of bias and uncertainty values		
Sample 1	$123,1$ J	KV_{V}	$119,4$ J
Sample 2	$116,1$ J	<i>S</i> V	$4,7$ J
Sample 3	112,8 J	nv	5
Sample 4	$123,6$ J	From Formula (A.2): $u\left\langle KV_{V}\right\rangle$	$2,1$ J
Sample 5	$121,3$ J		
From certificate: degrees of freedom, v_{RM}	30	From Formula $(A.1)$: By	-4.4

Table A.1 — Example *—* **Results of the indirect verification tests**

Test results and data from certificates	Calculation of bias and uncertainty values		
From certificate: expanded uncertainty at a confidence level of about 95 %, $U_{\rm RM}$	$3,4$ J	From Formula $(A.3)$: $u(B_V)$	2,7
Since v_{RM} > 10, u_{RM} , the standard uncertainty, can be calculated as $U_{\rm RM}/2$	$1,7$ J	From Formula $(A.4)$: uy	$5,2$ J
Degrees of freedom for five samples, v_B	4	From Formula $(A.5)$: v_V	

Table A.1 *(continued)*

The primary result of the indirect verification is good: the absolute value of the bias $(B_V = -4.4$ J) is below the upper threshold set in [Clause](#page-18-1) 7. The value of B_V needs to be combined with its uncertainty to obtain u_V , unless its value is well established, which we do not consider to be the case here. From [Formula](#page-28-1) $(A.5)$, the number of degrees of freedom corresponding to u_V is calculated to be 7. The verification results can be reported as shown in [Table](#page-29-0) A.2.

Table A.2 – Summary table of the result with expanded measurement uncertainty, $U|KV|$

A graphical representation of the example is given in [Figure](#page-29-1) A.1, together with the results obtained if the measured absorbed energy values are corrected for the measured bias. The uncertainty of the indirect verification is relatively large $(w = 5.2)$ as it consists of the combination of $u(B_V)$ and B_V . If the bias value had been better established, and the measured value corrected for its value, a considerably smaller uncertainty could have been obtained $[u(B_V) = 2.7]$].

Figure A.1 — Graphical representation of the default approach (left) with an uncorrected absorbed energy and the associated uncertainty, u_V , as well as the case where the measured **value is corrected for the bias (right), giving a smaller uncertainty,** $u(B_V)$

Annex B

(informative)

Measurement uncertainty of the results of the direct verification of a Charpy pendulum impact testing machine

B.1 General

Direct verification consists of a series of checks of geometrical and mechanical features of a pendulum impact testing machine. Deviation from the nominal values of these features contributes to the bias in the instrument with respect to the expected behaviour of a pendulum impact testing machine fulfilling the requirements of [Clause](#page-11-1) 6.

In theory, one can use a formula such as [Formula](#page-30-2) (B.1) for the estimation of *z*, the combined instrument bias:

$$
z = R + A + C + E + V + (l - l_1) + H + S \tag{B.1}
$$

where

- *R* is the bias in *K* (in energy units) due to bias in the radius of the edge of the striker;
- *A* is the bias in *K* (in energy units) due to bias in anvil and supports geometry;
- *C* is the bias in *K* (in energy units) due to bias of the centre of strike;
- *E* is the bias in *K* (in energy units) due to the energy calculation from measured angles;
- *V* is the bias in *K* (in energy units) due to bias in the impact velocity;
- (*l* − *l*1) is the bias in *K* (in energy units) due to bias in the difference between pendulum length and centre of percussion;
- *H* is the bias in *K* (in energy units) due to the correction for friction loss;
- *S* is the bias in *K* (in energy units) due to the bias in the energy read from an analogue or digital scale.

The effects of the factors $(R, A, C, E, V, l - l_1, H, S)$ on the absorbed energy are assumed to be small if they are within the tolerances required for direct verification of the machine (see [Clause](#page-11-1) 6) and if the pendulum impact test is performed according to the standard procedure (see ISO 148-1). However, there are uncertainties associated with the assessment of the individual factors contributing to *z*. Assuming that all quantities are independent, the combined standard uncertainty of *z* would be as given by [Formula](#page-30-3) (B.2):

$$
u_c(z) = \sqrt{u^2(R) + u^2(A) + u^2(C) + u^2(E) + u^2(V) + u^2(I - l_1) + u^2(H) + u^2(S)}
$$
(B.2)

Not all the elements from [Formula](#page-30-2) (B.1) and [Formula](#page-30-3) (B.2) can be reliably and quantitatively assessed. Instead, indirect verification of the instrument, with reference materials, is used to assess the bias in a pendulum and the associated uncertainty.

Nevertheless, it remains important to consider the reliability of the different steps in the mandatory direct verification. This is why this Annex discusses state-of-the-art methods to determine the

uncertainties associated with the results of a number of measurements performed during the direct verification of a Charpy pendulum impact machine.

Usually, the uncertainty of a certified value on the certificate of a certified reference material is specified for a confidence level of about 95 %. Therefore, the standard combined uncertainty, u_{RM} , has to be expanded using an appropriate coverage factor, *k*. The coverage factor to be used depends on the number of degrees of freedom associated with the combined uncertainty, which can be computed using the Welch-Satterthwaite approximation. For a typical case, the number of effective degrees of freedom is larger than 20 and a coverage factor of $k = 2$ can be used.

NOTE Other methods to assess the measurement uncertainties can be developed and are acceptable if they meet the requirements of the GUM (see Reference [\[1](#page-43-1)]).

The ultimate aim is to achieve a reliable estimate of the measurement uncertainty for the directly verified features so as to verify whether the sum of the deviation between the nominal and the measured value and the measurement uncertainty of this deviation is within the tolerances allowed by [Clause](#page-11-1) 6.

Uncertainty disclaimer note: Measurement uncertainty analysis is useful for identifying major sources of inconsistencies of measured results. Product standards and material property databases based on this and the previous version of this part of ISO 148 have an inherent contribution from measurement uncertainty. It is therefore inappropriate to apply further adjustments for measurement uncertainty and thereby risk failing product compliance. For this reason, the estimates of uncertainty derived by following this procedure are for information only, unless specifically instructed otherwise by the customer. The test conditions and limits defined in this part of ISO 148 should not be adjusted to take account of uncertainties of measurement, unless specifically instructed otherwise by the customer. The estimated measurement uncertainties should not be combined with measured results to assess compliance with product specifications, unless specifically instructed otherwise by the customer. Instead, the indicated tolerances are to be interpreted as acceptance intervals.[[2](#page-43-2)] This approach assumes that measurements are made with a tacitly accepted maximum measurement uncertainty. Where possible, this maximum measurement uncertainty has been specified in the current version of the ISO 148 series. Measurement uncertainties of the measured values should be smaller than the indicated values.

B.2 Uncertainty for particular instrument parameters

B.2.1 Centre of percussion

The pendulum is constructed in a way that makes pendulum length, *l*, equal to the distance between the centre of percussion and the axis of rotation, *l*1.

For the determination of l_1 , [Formula](#page-31-0) (B.3) is valid:

$$
l_1 = \frac{gt^2}{4\pi^2}
$$
 (B.3)

where

- l_1 is the distance between the position of the centre of percussion and the axis of rotation (reduced pendulum length), in metres;
- *t* is the average period of swing of pendulum from three measurements at 100, 50 or 25 swings.

The time measurement *T*, e.g. for 50 swings, is carried out manually or by a calibrated time-measuring device. In this example, a realistic measurement uncertainty of *u*(*T*) = 0,1 s will be used. The uncertainty of l_1 can then be calculated as given by **[Formula](#page-32-0)** $(B.4)$:

$$
u(l_1) = \frac{2gT}{(4\pi^2) \cdot 50^2} \cdot u(T)
$$
 (B.4)

The pendulum length, *l*, is measured with callipers. Because *l* can often not be measured directly, it is determined by three partial measurements *L*1, *L*2 and *L*3, which means:

$$
u(l) = \sqrt{u^2(L_1) + u^2(L_2) + u^2(L_3)}
$$
(B.5)

Callipers for smaller lengths (e.g. *L*1 and *L*3) usually have a measurement uncertainty of 0,1 mm. Callipers for the larger length (here *L*2) typically have a measurement uncertainty of 0,3 mm. In this case, the combined uncertainty $u(l) = 0.3$ mm.

NOTE These values are typically included on the calibration certificate of the instrument used.

The measurement uncertainty of the deviation of the position of the centre of percussion from the measured pendulum length, (*l* − *l*1), is calculated with the above-given uncertainties as given by [Formula](#page-32-1) (B.6):

$$
u(l - l_1) = \sqrt{u^2(l) + u^2(l_1)}
$$
 (B.6)

EXAMPLE See also [Table](#page-32-2) B.1.

For a measured pendulum length $l = 800,0$ mm, a measured T (50 swings) = 89,7 s, and the resulting calculated value for l_1 = 799,75 mm, and using the above uncertainties for length and time measurements, an uncertainty *u*(*l* − *l*1) of 1,07 mm is obtained. This shows that the measured (*l* − *l*1) is within the allowed tolerance (0,5 %), also taking into account measurement uncertainty.

B.2.2 Impact velocity

The impact velocity is calculated from the pendulum length and the fall angle and is a typical parameter of the testing machine. The permissible errors specified in this part of ISO 148 for the direct verification are relatively large. Since the relative uncertainties of the measurements needed to calculate impact velocity are very small, a specific calculation of the uncertainty of its value is not required.

B.2.3 Absorbed energy calculation

For the calculation of the absorbed energy, [Formula](#page-33-0) (B.7) measurement formulae is valid:

$$
KV = F \times l_2 \times (\cos \beta - \cos \alpha) \tag{B.7}
$$

where

- *KV* is the absorbed energy as calculated from measured fall and rise angles, in joules;
- *F* is the force exerted by the pendulum in the horizontal position on the force-proving device for distance l_2 , in newtons;
- *l*₂ is the distance between the point of application of force F and the axis of rotation, in metres;
- *β* is the angle of fall, in degrees;
- *α* is the angle of rise, in degrees.

The above parameters are not bound by certain nominal values or ranges in the standard. Therefore, there is no bias associated with these parameters, only a measurement uncertainty. The uncertainty of the energy calculated from the measured values is expressed as given in [Formula](#page-33-1) (B.8):

$$
u_1^2 = \left(\frac{\partial KV}{\partial F}\right)^2 u^2 \left(F\right) + \left(\frac{\partial KV}{\partial l_2}\right)^2 u^2 \left(l_2\right) + \left(\frac{\partial KV}{\partial \beta}\right)^2 u^2 \left(\beta\right) + \left(\frac{\partial KV}{\partial \alpha}\right)^2 u^2 \left(\alpha\right)
$$
(B.8)

From [Formula](#page-33-0) (B.7), the following can be derived:

$$
\frac{\partial KV}{\partial \alpha} = F \cdot l_2 \cdot \sin \alpha \tag{B.9}
$$

$$
\frac{\partial KV}{\partial \beta} = -F \cdot l_2 \cdot \sin \beta \tag{B.10}
$$

$$
\frac{\partial KV}{\partial F} = l_2 \cdot \left(\cos\beta - \cos\alpha\right) \tag{B.11}
$$

$$
\frac{\partial KV}{\partial l_2} = F \cdot \left(\cos \beta - \cos \alpha\right) \tag{B.12}
$$

With respect to the individual uncertainty contributions:

$$
u(F) = \sqrt{u^2(F_{\text{ftd}}) + u^2(t) + u^2(S) + u^2(D)}
$$
 (B.13)

ISO 148-2:2016(E)

where

$$
u(t) = \frac{\delta \cdot a_{\text{temp}}}{\sqrt{3}} \tag{B.14}
$$

where

δ is the temperature coefficient of the working standard (given by the manufacturer);

 a_{temp} is the deviation from the reference temperature.

$$
u(S) = \frac{a_{\text{stab}}}{\sqrt{3}}\tag{B.15}
$$

where

 a_{stab} is the long-term stability of the working standard;

$$
u(D) = a_{\text{int-dev}} \tag{B.16}
$$

where

 $a_{int-dev}$ is the interpolation deviation of the working standard;

$$
u(l_2) = \frac{\Delta l_2}{l_2} \tag{B.17}
$$

where

Δ*l*² is the uncertainty of the distance measurement between the point of application of the force and the axis of rotation.

NOTE A minimum estimate for Δ*l*2 can be taken from the certificate of the instrument used to measure *l*2.

EXAMPLE See also [Table](#page-35-0) B.2.

a) **Force**

Measurement uncertainty of the force transducer: $U_{\text{ftd}} = 0.12 \% (k = 2)$

Long-term stability of the force transducer: $a_{stab} = 0.05$ %

Temperature coefficient of the force transducer: δ = 0,01 %

Deviation from the reference temperature: $a_{temp} = 5.0$ °C

Measurement uncertainty due to linear interpolation of the force exerted by the pendulum on the force-proving device: $a_{\text{int-dev}} = 0.05\%$

Force exerted by the pendulum on the force-proving device at a 750,1 mm length of the pendulum: *F* = 206,70 N

The combined contributions to the force uncertainty reach 0,1 %. For a force *F* of 206,70 N, the combined standard uncertainty, *u*(*F*), is therefore 0,21 N.

b) **Pendulum length**

Uncertainty of the distance measurement: $l_2 = 0.3$ mm

Length of the pendulum: $l = l_2 = 750,1$ mm

The uncertainty of the distance *l*2, over which the force measurement is carried out, can be applied with $\Delta l_2 = \pm 0.3$ mm for careful use.

c) **Angles**

Uncertainty of the angle measurement: $\Delta \alpha = \Delta \beta = 0.2^\circ$; rise angle: *β* = 120°; fall angle: *α* = 160°

Care should be taken to convert degrees into radians and millimetres into metres prior to applying the formulae.

Quantity	Estimated value	Uncertainty		Standard		Contribution
		Value	Distribution type	uncertainty	Sensitivity coefficient	to uncertainty of KV
F	206.7 N	0,21 N	Normal	0.21 _N	$0,33$ J/N	$0,07$ J
L	750,1 mm	$0,3 \text{ mm}$	Rectangular	$0,17$ mm	91 J/m	$0,016$ J
β	120°	0.2°	Rectangular	0.12°	134 J/rad	$0,27$ J
α	160°	0.2°	Rectangular	0.12°	53 J/rad	$0,11$ J
Combined measurement uncertainty						$0,30$ J
Expanded measurement uncertainty using $k = 2$ for a 95 % confidence level						0,6

Table B.2 — Budget of measurement uncertainty for the absorbed energy calculation

B.2.4 Absorbed energy readings from an analog or a digital scale

S is the bias in the scale mechanism; it indicates the difference between the reading of an absorbed energy from the instrument analog scale or a digital value displayed on the instrument PC, and the calculated energy. *S* can be deduced for a particular pendulum using the results of direct verification, as given by [Formula](#page-35-1) (B.18):

$$
S = K_S - K_{\text{calc}} \tag{B.18}
$$

where

- *S* is the deviation of the indicated energy;
- K_S is from the calculated energy, K_{calc} , both in joules.
- The effective uncertainty, $u(S)$, is calculated as given by [Formula](#page-35-2) (B.19) and Formula (20):

$$
u(S) = \sqrt{u^2(K_S) + u^2(K_{\text{calc}})}
$$
(B.19)

where

$$
u(K_{\rm S}) = \frac{a}{2 \cdot \sqrt{3}}\tag{B.20}
$$

where

- *a* is the resolution of the scale (i.e. the smallest distinguishable difference between two measured values).
- EXAMPLE See also [Table](#page-36-0) B.3.

Value read from analog scale: $K_S = 68.0$ J

Resolution of the indicator: $a = 0.5$ J

Energy value calculated from measured angles: K_{calc} = 68,17 J

Uncertainty of the energy calculated from measured angles: $u(K_{\text{calc}}) = 0.38$ J

Quantity	Estimated value	Uncertainty		Standard		Contribution
		Value	Distribution type	uncertainty	Sensitivity coefficient	to uncertainty of S
K_{S}	$68,0$ J	$0,5$ J	Rectangular	$0,14$ J		$0,14$ J
K_{calc}	68,17 J	$0,3$ J	Normal	$0,3$ J		$0,3$ J
	$0,33$ J					
Expanded measurement uncertainty using $k = 2$ for a 95 % confidence level						0,7

Table B.3 — Measurement uncertainty of the deviation of the indicated absorbed energy

Annex C

(informative)

Direct method of verifying the geometric properties of pendulum impact testing machines using a jig

C.1 Field of application

This Annex describes a direct method for verifying the geometric properties of pendulum impact testing machines using a jig.

The properties which can be verified are the following:

- position of the striker in the plane of symmetry of the anvils;
- horizontality of the axis of rotation of the pendulum;
- perpendicularity between the arm of the pendulum and the axis of rotation;
- alignment of the striker and the arm of the pendulum;
- perpendicularity between the plane of the striker and the test piece.

This method may be applied to all machines and, in particular, to machines without a reference plane on the framework.

C.2 Jig

The shape and the dimensions of the jig are specified in [Figure](#page-38-0) C.1. The jig has two ends (A and B) corresponding to two positions of use (A and B).

C.3 Procedure

Before using the jig, the following two properties should be verified using a level:

- the horizontality of the plane of the supports;
- the perpendicularity between the plane of the anvils and the plane of the supports.

The jig should be used in the two positions (A and B). As shown in [Figure](#page-39-0) C.2, passing from position A to position B corresponds to the striker travelling 30 mm.

[Figure](#page-40-0) C.3 and [Figure](#page-41-0) C.4 illustrate the way in which to use the jig for verifying the properties defined in [C.1](#page-37-2).

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Dimensions in millimetres

Key

1 end A of jig

2 end B of jig

EXAMPLE X46Cr13 (55 HRC), 100Cr6 (62 HRC).

NOTE 1 Material: Stainless steel or steel with improved corrosion resistance, with low thermal expansion.

NOTE 2 All the dimensional tolerances should be ± 0.2 mm unless otherwise specified.

Figure C.1 — Jig

Dimensions in millimetres

- 3 end A of jig
- 4 end B of jig

Figure C.2 — Change of position from A to B corresponding to the striker travelling 30 mm

Key

- 1 test piece
- 2 jig
- 3 end A
- 4 end B
- a Pendulum axis.
- b Plane of swing of the pendulum perpendicular to the longitudinal axis of the test piece.
- c Plane of swing of the pendulum not perpendicular to the longitudinal axis of the test piece.

Figure C.3 — Example of application of the jig illustrated in [Figure](#page-38-0) C.1

In [Figure](#page-40-0) C.3:

- a) the plane of swing of the pendulum is not perpendicular to the longitudinal axis of the test piece (right-hand figures);
- b) the error is characterized by the fact that the striking edge is in contact with the sides of the jig: top left and bottom right parts of end A of the jig.

Key

- 1 test piece
- 2 jig
- a Pendulum axis.
- b Plane of symmetry of the hammer in the plane of swing of the pendulum.
- c Plane of symmetry of the hammer not in the plane of swing of the pendulum.

Figure C.4 — Example of application of the jig illustrated in [Figure](#page-38-0) C.1

In [Figure](#page-41-0) C.4:

- a) the plane of symmetry of the hammer is not in the plane of swing of the pendulum (right-hand figures);
- b) the error is characterized by the fact that the striking edge is in contact with the sides of the jig: top left and bottom right parts of end A of the jig;

c) the error is characterized by the fact that the striking edge is not in contact with the bottom of the V of the jig.

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