#### BS ISO 20765-2:2015



### **BSI Standards Publication**

# Natural gas — Calculation of thermodynamic properties

Part 2: Single-phase properties (gas, liquid, and dense fluid) for extended ranges of application



BS ISO 20765-2:2015

#### National foreword

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## Natural gas — Calculation of thermodynamic properties —

#### Part 2:

Single-phase properties (gas, liquid, and dense fluid) for extended ranges of application

Gaz naturel — Calcul des propriétés thermodynamiques —

Partie 2: Propriétés des phases uniques (gaz, liquide, fluide dense) pour une gamme étendue d'applications



BS ISO 20765-2:2015 **ISO 20765-2:2015(E)** 



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Cont	tents		Page
Forew	ord		v
1	Scope		1
2	Norm	ative references	2
3		s and definitions	
4		nodynamic basis of the method	
•	4.1	Principle	4
	4.2	The fundamental equation based on the Helmholtz free energy	4
		4.2.1 Background 4.2.2 The Helmholtz free energy	
		4.2.3 The reduced Helmholtz free energy	
		4.2.4 The reduced Helmholtz free energy of the ideal gas	6
		4.2.5 The pure substance contribution to the residual part of the reduced Helmholtz free energy	
		4.2.6 The departure function contribution to the residual part of the reduced Helmholtz free energy	
		4.2.7 Reducing functions	
	4.3	Thermodynamic properties derived from the Helmholtz free energy 4.3.1 Background	8
		4.3.1 Background  4.3.2 Relations for the calculation of thermodynamic properties in the	ð
		homogeneous region	9
5	Metho	od of calculation	11
	5.1	Input variables	
	5.2	Conversion from pressure to reduced density	
	5.3	Implementation	
6		es of application	13
	6.1 6.2	Pure gases	
	6.3	Binary mixtures Natural gases	
7		tainty of the equation of state	
/	7.1	Background	
	7.2	Uncertainty for pure gases	
		7.2.1 Natural gas main components	
		7.2.2 Secondary alkanes	
	7.0	7.2.3 Other secondary components	
	7.3 7.4	Uncertainty for binary mixtures	
	7.4	7.4.1 Uncertainty in the normal and intermediate ranges of applicability of	
		natural gas	24
		beyond this range	25
	7.5	Uncertainties in other properties	25
	7.6	Impact of uncertainties of input variables	
8	Repor	rting of results	25
Annex	A (noi	rmative) <b>Symbols and units</b>	27
Annex	B (noi	rmative) The reduced Helmholtz free energy of the ideal gas	29
Annex	C (nor	rmative) Values of critical parameters and molar masses of the pure components	35
Annex	D (no	rmative) The residual part of the reduced Helmholtz free energy	36
Annex	E (nor	mative) The reducing functions for density and temperature	48
		ormative) Assignment of trace components	

## BS ISO 20765-2:2015 **ISO 20765-2:2015(E)**

Annex G (informative) Examples	.57
Bibliography	.60

#### Foreword

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The committee responsible for this document is ISO/TC 193, *Natural Gas*, Subcommittee SC 1, *Analysis of Natural Gas*.

ISO 20765 consists of the following parts, under the general title *Natural gas — Calculation of thermodynamic properties*:

- Part 1: Gas phase properties for transmission and distribution applications
- Part 2: Single-phase properties (gas, liquid, and dense fluid) for extended ranges of application
- Part 3: Two-phase properties (vapour-liquid equilibria)

### Natural gas — Calculation of thermodynamic properties —

#### Part 2:

## Single-phase properties (gas, liquid, and dense fluid) for extended ranges of application

#### 1 Scope

This part of ISO 20765 specifies a method to calculate volumetric and caloric properties of natural gases, manufactured fuel gases, and similar mixtures, at conditions where the mixture may be in either the homogeneous (single-phase) gas state, the homogeneous liquid state, or the homogeneous supercritical (dense-fluid) state.

NOTE 1 Although the primary application of this document is to natural gases, manufactured fuel gases, and similar mixtures, the method presented is also applicable with high accuracy (i.e., to within experimental uncertainty) to each of the (pure) natural gas components and to numerous binary and multi-component mixtures related to or not related to natural gas.

For mixtures in the gas phase and for both volumetric properties (compression factor and density) and caloric properties (for example, enthalpy, heat capacity, Joule-Thomson coefficient, and speed of sound), the method is at least equal in accuracy to the method described in Part 1 of this International Standard, over the full ranges of pressure p, temperature T, and composition to which Part 1 applies. In some regions, the performance is significantly better; for example, in the temperature range 250 K to 275 K (–10 °F to 35 °F). The method described here maintains an uncertainty of  $\leq 0.1$  % for volumetric properties, and generally within 0,1 % for speed of sound. It accurately describes volumetric and caloric properties of homogeneous gas, liquid, and supercritical fluids as well as those in vapour-liquid equilibrium. Therefore its structure is more complex than that in Part 1.

NOTE 2 All uncertainties in this document are expanded uncertainties given for a 95 % confidence level (coverage factor k = 2).

The method described here is also applicable with no increase in uncertainty to wider ranges of temperature, pressure, and composition for which the method of Part 1 is not applicable. For example, it is applicable to natural gases with lower content of methane (down to 0,30 mole fraction), higher content of nitrogen (up to 0,55 mole fraction), carbon dioxide (up to 0,30 mole fraction), ethane (up to 0,25 mole fraction), and propane (up to 0,14 mole fraction), and to hydrogen-rich natural gases. A practical usage is the calculation of properties of highly concentrated  $\rm CO_2$  mixtures found in carbon dioxide sequestration applications.

The mixture model presented here is valid by design over the entire fluid region. In the liquid and dense-fluid regions the paucity of high quality test data does not in general allow definitive statements of uncertainty for all sorts of multi-component natural gas mixtures. For saturated liquid densities of LNG-type fluids in the temperature range from 100 K to 140 K ( $-280\,^{\circ}$ F to  $-208\,^{\circ}$ F), the uncertainty is  $\leq (0.1-0.3)\,^{\circ}$ %, which is in agreement with the estimated experimental uncertainty of available test data. The model represents experimental data for compressed liquid densities of various binary mixtures to within  $\pm (0.1-0.2)\,^{\circ}$ % at pressures up to 40 MPa ( $5800\,^{\circ}$ psia), which is also in agreement with the estimated experimental uncertainty. Due to the high accuracy of the equations developed for the binary subsystems, the mixture model can predict the thermodynamic properties for the liquid and dense-fluid regions with the best accuracy presently possible for multi-component natural gas fluids.

#### 2 Normative references

The following documents, in whole or in part, are normatively referenced in this document and are indispensable for its application. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

ISO 7504, Gas Analysis — Vocabulary

ISO 14532, Natural gas — Vocabulary

ISO 20765-1, Natural gas — Calculation of thermodynamic properties — Part 1: Gas phase properties for transmission and distribution applications

ISO 80000-5:2007, Quantities and units — Part 5: Thermodynamics

#### 3 Terms and definitions

For the purposes of this document, the terms and definitions in ISO 80000-5:2007 and/or ISO 20765-1, ISO 7504, ISO 14532, and the following apply.

NOTE 1 See Annex A for the list of symbols and units used in this part of ISO 20765.

NOTE 2 Figure 1 is a schematic representation of the phase behaviour of a typical natural gas as a function of pressure and temperature. The positions of the bubble and dew lines depend upon the composition. This phase diagram may be useful in understanding the definitions below.

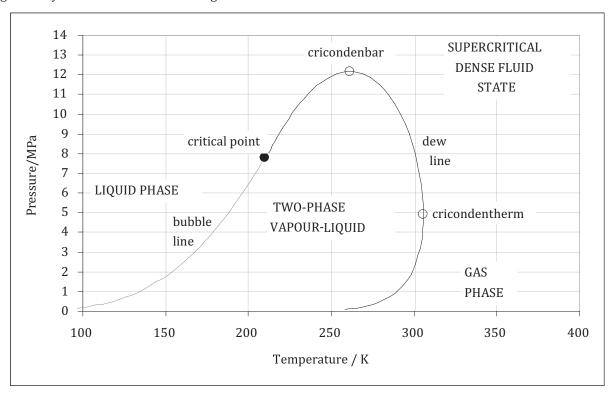


Figure 1 — Phase diagram for a typical natural gas

### 3.1 bubble pressure

pressure at which an infinitesimal amount of vapour is in equilibrium with a bulk liquid for a specified temperature

#### 3.2

#### bubble temperature

temperature at which an infinitesimal amount of vapour is in equilibrium with a bulk liquid for a specified pressure

Note 1 to entry: The locus of bubble points is known as the bubble line.

Note 2 to entry: More than one bubble temperature may exist at a specific pressure. Moreover, more than one bubble pressure may exist at a specified temperature, as explained in the example given in 3.6.

#### 3.3

#### cricondenbar

maximum pressure at which two-phase separation can occur

#### 3.4

#### cricondentherm

maximum temperature at which two-phase separation can occur

#### 3.5

#### critical point

unique saturation point along the two-phase vapour-liquid equilibrium boundary where both the vapour and liquid phases have the same composition and density

Note 1 to entry: The critical point is the point at which the dew line and the bubble line meet.

Note 2 to entry: The pressure at the critical point is known as the critical pressure and the temperature as the critical temperature.

Note 3 to entry: A mixture of given composition may have one, more than one, or no critical points. In addition, the phase behaviour may be quite different from that shown in <u>Fig. 1</u> for mixtures (including natural gases) containing, e.g., hydrogen or helium.

#### 3.6

#### dew pressure

pressure at which an infinitesimal amount of liquid is in equilibrium with a bulk vapour for a specified temperature

Note 1 to entry: More than one dew pressure may exist at the specified temperature. For example, isothermal compression at 300 K with a gas similar to that shown in Figure 1: At low pressure the mixture is a gas. At just above 2 MPa (the dew pressure), a liquid phase initially forms. As pressure increases more liquid forms in the two-phase region, but a further increase in pressure reduces the amount of liquid (retrograde condensation) until at about 8 MPa where the liquid phase disappears at the upper dew pressure, and the mixture is in the dense gas phase. In the two-phase region, the overall composition is as specified, however the coexisting vapour and liquid will have different compositions.

#### 3.7

#### dew temperature

temperature at which an infinitesimal amount of liquid is in equilibrium with a bulk vapour for a specified pressure

Note 1 to entry: More than one dew temperature may exist at a specified pressure, similar to the example given in 3.6.

Note 2 to entry: The locus of dew points is known as the dew line.

#### 3.8

#### supercritical state

dense phase region above the critical point (often considered to be a state above the critical temperature and pressure) within which no two-phase separation can occur

#### 4 Thermodynamic basis of the method

#### 4.1 Principle

The method is based on the concept that natural gas or any other type of mixture can be completely characterized in the calculation of its thermodynamic properties by component analysis. Such an analysis, together with the state variables of temperature and density, provides the necessary input data for the calculation of properties. In practice, the state variables available as input data are generally temperature and pressure, and it is thus necessary to first iteratively determine the density using the equations provided here.

These equations express the Helmholtz free energy of the mixture as a function of density, temperature, and composition, from which all other thermodynamic properties in the homogeneous (single-phase) gas, liquid, and supercritical (dense-fluid) regions may be obtained in terms of the Helmholtz free energy and its derivatives with respect to temperature and density. For example, pressure is proportional to the first derivative of the Helmholtz energy with respect to density (at constant temperature).

NOTE These equations are also applicable in the calculation of two-phase properties (vapour-liquid equilibria). Additional composition-dependent derivatives are required and are presented in Part 3 of this International Standard.

The method uses a detailed molar composition analysis in which all components present in amounts exceeding 0,000 05 mole fraction (50 ppm) are specified. For a typical natural gas, this might include alkane hydrocarbons up to about  $C_7$  or  $C_8$  together with nitrogen, carbon dioxide, and helium. Typically, isomers for alkanes  $C_6$  and higher may be lumped together by molar mass and treated collectively as the normal isomer.

For some fluids, additional components such as  $C_9$ ,  $C_{10}$ , water, and hydrogen sulfide may be present and need to be taken into consideration. For manufactured gases, hydrogen, carbon monoxide, and oxygen may also be present in the mixture.

More precisely, the method uses a 21-component analysis in which all of the major and most of the minor components of natural gas are included (see <u>Clause 6</u>). Any trace component present but not identified as one of the 21 specified components may be assigned appropriately to one of these 21 components (see <u>Annex F</u>).

#### 4.2 The fundamental equation based on the Helmholtz free energy

#### 4.2.1 Background

The GERG-2008 equation [1] was published by the Lehrstuhl für Thermodynamik at the Ruhr-Universität Bochum in Germany as a new wide-range equation of state for the volumetric and caloric properties of natural gases and other mixtures. It was originally published in 2007[2] and later updated in 2008.[1] The new equation improves upon the performance of the AGA-8 equation [3] for gas phase properties and in addition is applicable to the properties of the liquid phase, to the dense-fluid phase, to the vapour-liquid phase boundary, and to properties for two-phase states. The ranges of temperature, pressure, and composition to which the GERG-2008 equation of state applies are much wider than the AGA-8 equation and cover an extended range of application. The Groupe Européen de Recherches Gazières (GERG) supported the development of this equation of state over several years.

The GERG-2008 equation is explicit in the Helmholtz free energy, a formulation that enables all thermodynamic properties to be expressed analytically as functions of the free energy and of its derivatives with respect to the state conditions of temperature and density. There is generally no need for numerical differentiation or integration within any computer program that implements the method.

#### 4.2.2 The Helmholtz free energy

The Helmholtz free energy a of a fluid mixture at a given mixture density  $\rho$ , temperature T, and molar composition  $\bar{x}$  can be expressed as the sum of  $a^{\rm o}$  describing the ideal gas behaviour and  $a^{\rm r}$  describing the residual or real-gas contribution, as follows:

$$a(\rho, T, \overline{x}) = a^{0}(\rho, T, \overline{x}) + a^{r}(\rho, T, \overline{x}) \tag{1}$$

#### 4.2.3 The reduced Helmholtz free energy

The Helmholtz free energy is often used in its dimensionless form  $\alpha = a/(RT)$  as

$$\alpha(\delta, \tau, \overline{x}) = \alpha^{0}(\rho, T, \overline{x}) + \alpha^{r}(\delta, \tau, \overline{x}) \tag{2}$$

In this equation, the reduced (dimensionless) mixture density  $\delta$  is given by

$$\delta = \frac{\rho}{\rho_{\rm r}(\bar{x})} \tag{3}$$

and the inverse reduced (dimensionless) mixture temperature  $\tau$  is given by

$$\tau = \frac{T_{\rm r}(\overline{x})}{T} \tag{4}$$

where

 $\rho_r$  and  $T_r$  are reducing functions for the mixture density and mixture temperature (see <u>4.2.7</u>) depending on the molar composition of the mixture only.

The residual part  $\alpha^r$  of the reduced Helmholtz free energy is given by

$$\alpha^{r}(\delta,\tau,\overline{x}) = \alpha_{0}^{r}(\delta,\tau,\overline{x}) + \Delta\alpha^{r}(\delta,\tau,\overline{x})$$
(5)

In this equation, the first term on the right-hand side  $\alpha_0^r$  describes the contribution of the residual parts of the reduced Helmholtz free energy of the pure substance equations of state, which are multiplied by the mole fraction of the corresponding substance, and calculated at the reduced mixture variables  $\delta$  and  $\tau$  (see equation (8)). The second term  $\Delta \alpha^r$  is the departure function, which is the double summation over all binary specific and generalized departure functions developed for the respective binary mixtures (see equation (10)).

#### 4.2.4 The reduced Helmholtz free energy of the ideal gas

The reduced Helmholtz free energy  $\alpha^0$  represents the properties of the ideal-gas mixture at a given mixture density  $\rho$ , temperature T, and molar composition  $\overline{x}$  according to

$$\alpha^{0}(\rho, T, \bar{x}) = \sum_{i=1}^{N} x_{i} \left[\alpha_{0i}^{0}(\rho, T) + \ln x_{i}\right]$$
(6)

In this equation, the term  $\sum x_i \ln x_i$  is the contribution from the entropy of mixing, and  $\alpha_{oi}^{o}(\rho,T)$  is the dimensionless form of the Helmholtz free energy in the ideal-gas state of component i, as given by

$$\alpha_{0i}^{o}(\rho,T) = \ln\left(\frac{\rho}{\rho_{c,i}}\right) + \frac{R^{*}}{R} \left[ n_{0i,1}^{o} + n_{0i,2}^{o} \frac{T_{c,i}}{T} + n_{0i,3}^{o} \ln\left(\frac{T_{c,i}}{T}\right) + \sum_{k=4,6} n_{0i,k}^{o} \ln\left|\sinh\left(\theta_{0i,k}^{o} \frac{T_{c,i}}{T}\right)\right| - \sum_{k=5,7} n_{0i,k}^{o} \ln\left|\cosh\left(\theta_{0i,k}^{o} \frac{T_{c,i}}{T}\right)\right| \right]$$

$$(7)$$

where

 $\rho_{c,i}$  and  $T_{c,i}$  are the critical parameters of the pure components (see Annex C).

The values of the coefficients  $n_{0i,k}^0$  and the parameters  $9_{0i,k}^0$  for all 21 components are given in Annex B.

NOTE 1 The method prescribed is taken without change from the method prescribed in Part 1 of this International Standard. The user should however be aware of significant differences that result inevitably from the change in definition of the inverse reduced temperature  $\tau$  between Part 1 and Part 2.

NOTE 2  $R = 8,314\,472\,\text{J}\cdot\text{mol}^{-1}\cdot\text{K}^{-1}$  was the internationally accepted standard for the molar gas constant [4] at the time of development of the equation of state. Equation (7) results from the integration of the equations for the ideal-gas heat capacities taken from [5], where a different molar gas constant was used than the one adopted in the mixture model presented here. The ratio  $R^*/R$  with  $R^*=8,314\,51\,\text{J}\cdot\text{mol}^{-1}\cdot\text{K}^{-1}$  takes into account this difference and therefore leads to the exact solution of the original equations for the ideal-gas heat capacity.

#### 4.2.5 The pure substance contribution to the residual part of the reduced Helmholtz free energy

The contribution of the residual parts of the reduced Helmholtz free energy of the pure substance equations of state  $\alpha_0^r$  to the residual part of the reduced Helmholtz free energy of the mixture is

$$\alpha_0^{\mathrm{r}}(\delta,\tau,\overline{x}) = \sum_{i=1}^{N} x_i \,\alpha_{0i}^{\mathrm{r}}(\delta,\tau) \tag{8}$$

where

 $\alpha_{oi}^{\rm r}(\delta,\tau)$  is the residual part of the reduced Helmholtz free energy of component i (i.e., the residual part of the respective pure substance equation of state listed in Table 2) and is given by

$$\alpha_{oi}^{r}(\delta,\tau) = \sum_{k=1}^{K_{Pol,i}} n_{oi,k} \, \delta^{d_{oi,k}} \, \tau^{t_{oi,k}} + \sum_{k=K_{Pol,i}+1}^{K_{Pol,i}+K_{Exp,i}} n_{oi,k} \, \delta^{d_{oi,k}} \, \tau^{t_{oi,k}} \, e^{-\delta^{c_{oi,k}}}$$
(9)

The equations for  $\alpha_{0i}^{r}$  use the same basic structure as further detailed in Annex <u>D.2</u>. The values of the coefficients  $n_{0i,k}$  and the exponents  $d_{0i,k}$ ,  $t_{0i,k}$  and  $c_{0i,k}$  for all 21 components are given in Annex <u>D.2.2</u>.

## 4.2.6 The departure function contribution to the residual part of the reduced Helmholtz free energy

The purpose of the departure function is to further improve the accuracy of the mixture model in the description of thermodynamic properties in addition to fitting the parameters of the reducing functions (see 4.2.7) when sufficiently accurate experimental data are available to characterize the properties of the mixture. The departure function  $\Delta \alpha^r$  of the multi-component mixture is the double summation over all binary specific and generalized departure functions developed for the binary subsystems and is given by

$$\Delta \alpha^{\mathrm{r}}(\delta, \tau, \overline{x}) = \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} \Delta \alpha_{ij}^{\mathrm{r}}(\delta, \tau, \overline{x})$$
(10)

with

$$\Delta \alpha_{ij}^{r}(\delta, \tau, \bar{x}) = x_i x_j F_{ij} \alpha_{ij}^{r}(\delta, \tau)$$
(11)

In this equation, the function  $\alpha^{r}_{ij}(\delta,\tau)$  is the part of the departure function  $\Delta\alpha^{r}_{ij}(\delta,\tau,\overline{x})$  that depends only on the reduced mixture variables  $\delta$  and  $\tau$ , as given by

$$\alpha_{ij}^{\mathbf{r}}(\delta,\tau) = \sum_{k=1}^{K_{\text{Pol},ij}} n_{ij,k} \, \delta^{d_{ij,k}} \, \tau^{t_{ij,k}}$$

$$+ \sum_{k=K_{\text{Pol},ij}+1}^{K_{\text{Pol},ij}+K_{\text{Exp},ij}} n_{ij,k} \, \delta^{d_{ij,k}} \, \tau^{t_{ij,k}} e^{-\eta_{ij,k}(\delta-\varepsilon_{ij,k})^2 - \beta_{ij,k}(\delta-\gamma_{ij,k})}$$

$$(12)$$

where

 $\alpha_{ij}^{\rm r}(\delta,\tau)$  was developed either for a specific binary mixture (a binary specific departure function with binary specific coefficients and exponents) or for a group of binary mixtures (generalized departure function with a uniform structure for the group of binary mixtures).

#### a) Binary specific departure functions

Binary specific departure functions were developed for the binary mixtures of methane with nitrogen, carbon dioxide, ethane, propane, and hydrogen, and of nitrogen with carbon dioxide and ethane. For a binary specific departure function, the adjustable factor  $F_{ij}$  in equation (11) equals unity.

#### b) Generalized departure function

A generalized departure function was developed for the binary mixtures of methane with n-butane and isobutane, of ethane with propane, n-butane, and isobutane, of propane with n-butane and isobutane, and of n-butane with isobutane. For each mixture in the group of generalized binary mixtures, the

parameter  $F_{ij}$  is fitted to the corresponding binary specific data (except for the binary system methanen-butane, where  $F_{ij}$  equals unity).

#### c) No departure function

For all of the remaining binary mixtures, no departure function was developed, and  $F_{ij}$  equals zero, i.e.,  $\Delta \alpha_{ij}^{\rm r}(\delta,\tau,\overline{x})$  equals zero. For most of these mixtures, however, the parameters of the reducing functions for density and temperature were fitted to selected experimental data (see <u>4.2.7</u> and <u>6.2</u>).

The values of the coefficients  $n_{ij,k}$ , the exponents  $d_{ij,k}$  and  $t_{ij,k}$ , and the parameters  $\eta_{ij,k}$ ,  $\varepsilon_{ij,k}$ ,  $\beta_{ij,k}$ , and  $\gamma_{ij,k}$  for all binary specific and generalized departure functions considered in the mixture model described here are given in Annex D.3, Table D.4. The number of digits given in these tables is as presented in the source publication; the effect of truncation is not obvious and all of the digits shall be used in all calculations. The non-zero  $F_{ij}$  parameters are listed in Table D.5.

NOTE Compared to the reducing functions for density and temperature, the departure function is in general of minor importance for the residual behaviour of the mixture since it only describes an additional small residual deviation to the real mixture behaviour. The development of such a function was, however, necessary to fulfil the high demands on the accuracy of the mixture model presented here in the description of the thermodynamic properties of natural gases and other mixtures.

#### 4.2.7 Reducing functions

The reduced mixture variables  $\delta$  and  $\tau$  are calculated from equations (3) and (4) by means of the composition-dependent reducing functions for the mixture density and temperature

$$\frac{1}{\rho_{r}(\bar{x})} = \sum_{i=1}^{N} \sum_{j=1}^{N} x_{i} x_{j} \beta_{v,ij} \gamma_{v,ij} \frac{x_{i} + x_{j}}{\beta_{v,ij}^{2} x_{i} + x_{j}} \left[ \frac{1}{8} \right] \left( \frac{1}{\rho_{c,i}^{1/3}} + \frac{1}{\rho_{c,j}^{1/3}} \right)^{3}$$

$$(13)$$

$$T_{r}(\overline{x}) = \sum_{i=1}^{N} \sum_{j=1}^{N} x_{i} x_{j} \beta_{T,ij} \gamma_{T,ij} \frac{x_{i} + x_{j}}{\beta_{T,ij}^{2} x_{i} + x_{j}} (T_{c,i} \cdot T_{c,j})^{0.5}$$
(14)

These functions are based on quadratic mixing rules and are reasonably connected to physically well-founded mixing rules. The binary parameters  $\beta_{v,ij}$  and  $\gamma_{v,ij}$  in equation (13) and  $\beta_{T,ij}$  and  $\gamma_{T,ij}$  in equation (14) are fitted to data for binary mixtures subject to the conditions  $\beta_{ij}=1/\beta_{ji}$  and  $\gamma_{ij}=\gamma_{ji}$ . The values of the binary parameters for all binary mixtures are listed in Table E.1 of Annex E. The critical parameters  $\rho_{C,i}$  and  $T_{C,i}$  of the pure components are given in Annex C.

NOTE The binary parameters of equations (13) and (14) were fitted based on the deviations between the behaviour of the real mixture (determined by experimental data) and the one resulting from ideal combining rules (with  $\beta$  and  $\gamma$  set to 1) for the critical parameters of the pure components. In those cases where sufficient experimental data are not available, the parameters of equations (13) and (14) are either set to unity or modified (calculated) in such a manner that the critical parameters of the pure components are combined in a different way, which proved to be more suitable for certain binary subsystems (see also Annex E.1).

#### 4.3 Thermodynamic properties derived from the Helmholtz free energy

#### 4.3.1 Background

The thermodynamic properties in the homogeneous gas, liquid, and supercritical regions of a mixture are related to derivatives of the Helmholtz free energy with respect to the reduced mixture variables  $\delta$  and  $\tau$ , as summarized in the following section (see <u>Table 1</u>). All of the thermodynamic properties may

be written explicitly in terms of the reduced Helmholtz free energy  $\alpha$  and its various derivatives. The required derivatives  $\alpha_{\tau}$ ,  $\alpha_{\tau\tau}$ ,  $\alpha_{\delta}$ ,  $\alpha_{\delta\delta}$ , and  $\alpha_{\delta\tau}$  are defined as follows:

$$\alpha_{\tau} = \left(\frac{\partial \alpha}{\partial \tau}\right)_{\delta, \overline{x}} \quad \alpha_{\tau\tau} = \left(\frac{\partial^{2} \alpha}{\partial \tau^{2}}\right)_{\delta, \overline{x}} \quad \alpha_{\delta} = \left(\frac{\partial \alpha}{\partial \delta}\right)_{\tau, \overline{x}} \quad \alpha_{\delta\delta} = \left(\frac{\partial^{2} \alpha}{\partial \delta^{2}}\right)_{\tau, \overline{x}} \quad \alpha_{\delta\tau} = \left(\frac{\partial}{\partial \tau}\left(\frac{\partial \alpha}{\partial \delta}\right)_{\tau, \overline{x}}\right)_{\delta, \overline{x}}$$

$$(15)$$

Each derivative is the sum of an ideal-gas part (see <u>Annex B</u>) and a residual part (see <u>Annex D</u>). The following substitutions help to simplify the appearance of the relevant relationships:

$$\alpha_{1} = \left(\frac{\partial(\delta 2\alpha_{\delta})}{\partial \delta}\right)_{\tau, x_{i}} = 2\delta\alpha_{\delta} + \delta^{2}\alpha_{\delta\delta} = 1 + 2\delta\alpha_{\delta}^{r} + \delta^{2}\alpha_{\delta\delta}^{r}$$

$$\tag{16}$$

$$\alpha_{2} = -\tau^{2} \left( \frac{\partial}{\partial \tau} \left( \frac{\delta \alpha_{\delta}}{\tau} \right) \right)_{\delta, x_{i}} = \delta \alpha_{\delta} - \delta \tau \alpha_{\delta \tau} = 1 + \delta \alpha_{\delta}^{r} - \delta \tau \alpha_{\delta \tau}^{r}$$

$$(17)$$

Detailed expressions for  $\alpha_{\tau}$ ,  $\alpha_{\tau\tau}$ ,  $\alpha_{\delta}$ ,  $\alpha_{\delta\delta}$ ,  $\alpha_{\delta\tau}$ ,  $\alpha_{1}$ , and  $\alpha_{2}$  can be found in Annexes B and D.

NOTE In addition to the derivatives of  $\alpha$  with respect to the reduced mixture variables  $\delta$  and  $\tau$ , composition derivatives of  $\alpha$  and of the reducing functions for density and temperature are required for the calculation of vapour-liquid equilibrium (VLE) properties as described in Part 3 of this International Standard.

#### 4.3.2 Relations for the calculation of thermodynamic properties in the homogeneous region

The relations between common thermodynamic properties and the reduced Helmholtz free energy  $\alpha$  and its derivatives are summarized in <u>Table 1</u>. The first column of this table defines the thermodynamic properties. The second column gives their relation to the reduced Helmholtz free energy  $\alpha$  of the mixture. In equations (26), (28), (29), (30), and (31), the basic expressions for the properties  $c_p$ , w,  $\mu_{\text{JT}}$ ,  $\phi$ , and  $\kappa$  have been additionally transformed, such that values of properties already derived can be used to simplify the subsequent calculations. This approach is useful for applications where several or all of the thermodynamic properties are to be determined.

In equations (22) to (27), the relations for the thermodynamic properties represent the molar quantities (i.e., quantity per mole, lower case symbols). Specific quantities (i.e., quantity per kilogram, represented normally by upper case symbols) are obtained by dividing the molar variables (e.g., v, u, s, h, g,  $c_v$ , and  $c_p$ ) by the molar mass M.

The molar mass M of the mixture is derived from the composition  $x_i$  and the molar masses  $M_i$  of the pure substances, as follows

$$M(\bar{x}) = \sum_{i=1}^{N} x_i \cdot M_i \tag{18}$$

The mass-based density *D* is given by

$$D = \rho M \tag{19}$$

NOTE 1 Values of the molar masses  $M_i$  of the pure substances are given in Annex C and are taken from [6]; these values are not identical with those given in ISO 20765-1 and ISO 6976:1995. [7] However, they are identical with the most recent values adopted by the international community of metrologists. In these equations, R is the molar gas constant; consequently R/M is the specific gas constant.

NOTE 2 See Annex B.1 for information on reference states for enthalpy and entropy.

Table 1 — Definitions of common thermodynamic properties and their relation to the reduced Helmholtz free energy  $\alpha$ 

Property and definition	Relation to $\alpha$ and its derivatives	
Pressure $p = -(\partial a / \partial v)_{T,\overline{x}}$	$\frac{p}{\rho RT} = 1 + \delta \alpha_{\delta}^{r}$	(20)
Compression factor $Z = p/(\rho RT)$	$Z = 1 + \delta \alpha_{\delta}^{r}$	(21)
Internal energy $u = a + Ts$	$\frac{u}{RT} = \tau \alpha_{\tau}$	(22)
Entropy $s = -(\partial a / \partial T)_{v, \overline{x}}$	$\frac{s}{R} = \tau \alpha_{\tau} - \alpha$	(23)
Isochoric heat capacity $c_v = (\partial u / \partial T)_{v,\overline{x}}$	$\frac{c_{v}}{R} = -\tau^{2} \alpha_{\tau\tau}$	(24)
Enthalpy $h = u + pv$	$\frac{h}{RT} = 1 + \delta \alpha_{\delta}^{r} + \tau \alpha_{\tau}$	(25)
Isobaric heat capacity $c_p = (\partial h / \partial T)_{p,\overline{x}}$	$\frac{c_p}{R} = -\tau^2 \alpha_{\tau\tau} + \frac{\alpha_2^2}{\alpha_1}$	(26)
Gibbs free energy $g = h - Ts$	$\frac{g}{RT} = 1 + \delta \alpha \frac{r}{\delta} + \alpha$	(27)
Speed of sound $w^2 = (1/M)(\partial p/\partial \rho)_{S,\overline{X}}$	$\frac{w^2M}{RT} = \alpha_1 - \frac{\alpha_2^2}{\tau^2 \alpha_{\tau\tau}} = Z\kappa = \alpha_1 \frac{c_p}{c_v}$	(28)
Joule-Thomson coefficient $\mu = (\partial T / \partial p)_{h,\overline{x}}$	$\mu R \rho = \frac{\alpha_2 - \alpha_1}{\alpha_2^2 - \tau^2 \alpha_{\tau \tau} \alpha_1} = \frac{R}{c_p} \left( \frac{\alpha_2}{\alpha_1} - 1 \right)$	(29)
Isothermal throttling coefficient $\phi = (\partial h / \partial p)_{T,\overline{x}}$	$\phi \rho = 1 - \frac{\alpha_2}{\alpha_1}$	(30)
Isentropic exponent $\kappa = -(v/p)(\partial p/\partial v)_{S,\overline{X}}$	$\kappa = \frac{\alpha_1}{1 + \delta \alpha_{\delta}^{r}} \left( 1 - \frac{\alpha_2^2}{\tau^2 \alpha_{\tau \tau} \alpha_1} \right) = \frac{\alpha_1}{Z} \frac{c_p}{c_v}$	(31)
Second virial coefficient $B = \lim_{\rho \to 0} (\partial Z / \partial \rho)_{T, \overline{X}}$	$B\rho_{\rm r} = \lim_{\delta \to 0} (\alpha_{\delta}^{\rm r})$	(32)
Third virial coefficient $C = \lim_{\rho \to 0} (\partial^2 Z / \partial \rho^2)_{T, \overline{X}} / 2$	$C\rho_{\rm r}^2 = \lim_{\delta \to 0} (\alpha_{\delta\delta}^{\rm r})$	(33)

#### 5 Method of calculation

#### 5.1 Input variables

The method presented in this standard uses reduced density, inverse reduced temperature, and molar composition as the input variables. If the mass-based density D is available as input, then  $\rho$  is obtained directly as  $\rho = D/M$ , where  $M(\bar{x})$  is the molar mass given by equation (18). For given values of the molar density  $\rho$ , temperature T, and molar composition  $\bar{x}$ , the reduced mixture variables  $\delta$  and  $\tau$  can be calculated from equations (3) and (4) using the reducing functions for density and temperature given by equations (13) and (14).

More often, however, absolute pressure, temperature, and molar composition are available as the input variables. As a consequence, it is usually necessary to first evaluate the reduced density  $\delta$  and the inverse reduced temperature  $\tau$  from the available inputs. The conversion from temperature to inverse reduced temperature is given by equation (4). Section 5.2 explains how to obtain the reduced density given pressure and temperature.

The composition in mole fractions is required for the following 21 components: methane, nitrogen, carbon dioxide, ethane, propane, n-butane, isobutane (2-methylpropane), n-pentane, isopentane (2-methylbutane), n-hexane, n-heptane, n-octane, n-nonane, n-decane, hydrogen, oxygen, carbon monoxide, water, hydrogen sulfide, helium, and argon. For natural gases and similar (multi-component) mixtures, the allowable ranges of mole fraction are defined in <u>6.3</u>. The sum of all mole fractions shall be unity.

NOTE 1 If the sum of all mole fractions is not unity within the limit of analytical resolution, then the composition is either faulty or incomplete. The user should not proceed until the source of this problem has been identified and eliminated.

NOTE 2 If the mole fractions of heptanes, octanes, nonanes, and decanes are unknown, then the use of a composite C6+ fraction may be applicable for density calculation, where the composite components are split into individual fractions. VLE calculations (including dew points) obtain the best results when all components in the mixture are quantified. The composite component simplification (C6+) may have a higher uncertainty since even small amounts of heptanes, octanes, nonanes, decanes, and higher hydrocarbons have a significant influence on the phase behaviour of the mixture. The user should carry out a sensitivity analysis in order to test whether a particular approximation of this type is suitable for the intended purpose.

NOTE 3 Composition given in volume or mass fractions will need to be converted to mole fractions using the method given in ISO 14912.[8]

#### 5.2 Conversion from pressure to reduced density

The combination of the relations for the reduced mixture variables  $\delta$  and  $\tau$  (equations (3) and (4)) and equation (21) results in the following expression

$$\frac{p\tau}{\delta \rho_{r}(\bar{x})RT_{r}(\bar{x})} = Z(\delta, \tau, \bar{x}) = 1 + \delta \alpha_{\delta}^{r}(\delta, \tau, \bar{x})$$
(34)

where

$$\alpha_{\delta}^{r}(\delta,\tau,\bar{x}) = \left(\frac{\partial\alpha^{r}}{\partial\delta}\right)_{\tau,\bar{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\partial\alpha_{0i}^{r}(\delta,\tau)}{\partial\delta}\right)_{\tau} + \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left(\frac{\partial\alpha_{ij}^{r}(\delta,\tau)}{\partial\delta}\right)_{\tau}$$
(35)

If the input variables are available as pressure, temperature, and molar composition, equation (34) may be solved for the reduced molar density  $\delta$ . The derivatives of  $\alpha_{oi}^{r}(\delta,\tau)$  with respect to  $\delta$  and the coefficients and exponents involved (see equation (9)) are given in Annex D.2. The derivatives of  $\alpha_{ij}^{r}(\delta,\tau)$  with respect to  $\delta$  and the coefficients and exponents involved (see equations (11) and (12)) are given in Annex D.3. Information on the reducing functions is given in Annex E.

## BS ISO 20765-2:2015 **ISO 20765-2:2015(E)**

The solution of equation (34) requires any suitable numerical method, where, in practice, a standard form of equation-of-state density-search algorithm may be the most convenient and satisfactory. Such algorithms usually use an initial estimate of the density (e.g., the ideal-gas approximation for low density gaseous states) and proceed to calculate the pressure p. In an iterative procedure, density values are changed with decreasing increments to find the optimal density that reproduces the known value of pressure to within a pre-established level of agreement. A suitable criterion in the present case is that the pressure p calculated from the iteratively determined reduced density  $\delta$  shall reproduce the input value of p at least to within 1 part in  $10^6$ . The user must be careful to determine that the calculated state is stable since multiple roots can exist.

#### 5.3 Implementation

Once the independent variables reduced density  $\delta$ , inverse reduced temperature  $\tau$ , and molar composition  $\overline{x}$  of the mixture are known, the reduced Helmholtz free energy and the other thermodynamic properties (see Table 1) can be calculated. Equation (2) formulates the reduced Helmholtz free energy as  $\alpha = \alpha^0 + \alpha^r$ . The relations for the ideal-gas part  $\alpha^0$  are given in equations (6) and (7). The relations for the residual part  $\alpha^r$ , which is formulated as a function of the reduced density  $\delta$ , the inverse reduced temperature  $\tau$ , and the molar composition  $\overline{x}$ , are specified in equations (5) and (8) to (12) so as to give the following expression for  $\alpha$ :

$$\alpha(\delta, \tau, \overline{x}) = \sum_{i=1}^{N} x_{i} \frac{R^{*}}{R} \left[ n_{0i,1}^{0} + n_{0i,2}^{0} \frac{T_{c,i}}{T} + n_{0i,3}^{0} \ln \left( \frac{T_{c,i}}{T} \right) + \sum_{k=4,6} n_{0i,k}^{0} \ln \left| \sinh \left( \theta_{0i,k}^{0} \frac{T_{c,i}}{T} \right) \right| \right]$$

$$- \sum_{k=5,7} n_{0i,k}^{0} \ln \left| \cosh \left( \theta_{0i,k}^{0} \frac{T_{c,i}}{T} \right) \right| \right] + \sum_{i=1}^{N} x_{i} \left[ \ln \left( \frac{\rho}{\rho_{c,i}} \right) + \ln x_{i} \right]$$

$$+ \sum_{i=1}^{N} x_{i} \left[ \sum_{k=1}^{K_{Pol,i}} n_{0i,k} \delta^{d_{0i,k}} \tau^{t_{0i,k}} + \sum_{k=K_{Pol,i}+1}^{K_{Pol,i}+K_{Exp,i}} n_{0i,k} \delta^{d_{0i,k}} \tau^{t_{0i,k}} e^{-\delta^{c_{0i,k}}} \right]$$

$$+ \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left[ \sum_{k=1}^{K_{Pol,ij}} n_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} + \sum_{k=K_{Pol,ij}+1}^{K_{Pol,ij}+K_{Exp,ij}} n_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} e^{-\eta_{ij,k}(\delta-\varepsilon_{ij,k})^{2} - \beta_{ij,k}(\delta-\gamma_{ij,k})} \right]$$

$$(36)$$

For all 21 components, the values of the coefficients  $n_{0i,k}^0$  and the parameters  $\mathcal{G}_{0i,k}^0$  of the ideal-gas part of the reduced Helmholtz free energy are given in Annex B. The values of the coefficients  $n_{0i,k}$  and exponents  $d_{0i,k}$ ,  $t_{0i,k}$ , and  $c_{0i,k}$  in the contribution to the residual parts of the pure substance equations of state are listed in Annex D. The values of the coefficients  $n_{ij,k}$ , the exponents  $d_{ij,k}$  and  $t_{ij,k}$ , and the parameters  $F_{ij}$ ,  $\eta_{ij,k}$ ,  $\varepsilon_{ij,k}$ ,  $\beta_{ij,k}$ , and  $\gamma_{ij,k}$  in the departure functions for all relevant binary mixtures are given in Annex D.

Derivatives of  $\alpha$  with respect to the reduced mixture variables  $\delta$  and  $\tau$  that are needed for the calculation of the various thermodynamic properties may be obtained from Annexes B and D. Annex B lists the derivatives of the ideal-gas part of the reduced Helmholtz free energy,  $a^o$ , with respect to the reduced density and inverse reduced temperature of the mixture. Derivatives of the contribution of the residual parts of the pure substance equations of state to the reduced residual Helmholtz free energy of the mixture,  $a^r$ , with respect to the reduced mixture variables  $\delta$  and  $\tau$  may be obtained from Annex D. Derivatives of the contribution of the departure functions for binary mixtures to  $a^r$  with respect to the reduced mixture variables  $\delta$  and  $\tau$  are also given in Annex D.

#### 6 Ranges of application

#### 6.1 Pure gases

The temperature and pressure ranges of validity for the pure fluid equations of state are listed in Table 2. For these ranges, the equations have been verified by experimental data. The lower temperatures correspond to rounded values of the triple point temperatures of the substances. For the main components, the equations are valid for temperatures up to at least 600 K (620 °F) and pressures from a perfect vacuum state up to 300 MPa (43 500 psia). For the secondary alkanes, the equations are valid for temperatures up to at least 500 K (440 °F) and pressures up to at least 35 MPa (5075 psia). For the other secondary components, the temperatures range up to at least 400 K (260 °F) (water up to 1273 K, 1830 °F) and pressures range up to at least 100 MPa (14 500 psia). The extrapolation yields reasonable results at temperatures and pressures far beyond the listed (tested) ranges of validity.

Table 2 — Validity range and references for the 21 components in the mixture modela

Pure substance	Reference	Tested range	of validity	Number
		Temperature	Pressure	of terms
		T/K	p <sub>max</sub> /MPa	
Main components				'
Methane	Klimeck (2000) <sup>[9]</sup>	90 - 623	300	24
Nitrogen	Klimeck (2000)[9]	63 – 700	300	24
Carbon dioxide	Klimeck (2000) <sup>[9]</sup>	90b - 900	300	22
Ethane	Klimeck (2000)[9]	90 - 623	300	24
Secondary alkanes				
Propane	Span & Wagner (2003)[10]	85 - 623	100	12
n-Butane	Span & Wagner (2003)[10]	134 - 693	70	12
Isobutane	Span & Wagner (2003) <sup>[10]</sup>	113 - 573	35	12
n-Pentane	Span & Wagner (2003)[10]	143 - 573	70	12
Isopentane	Lemmon & Span (2006)[11]	112 - 500	35	12
n-Hexane	Span & Wagner (2003)[10]	177 - 548	100	12
n-Heptane	Span & Wagner (2003)[10]	182 – 523	100	12
n-Octane	Span & Wagner (2003)[10]	216 - 548	100	12
n-Nonane	Lemmon & Span (2006)[11]	219 - 600	800	12
n-Decane	Lemmon & Span (2006) <sup>[11]</sup>	243 – 675	800	12
Other secondary com	ponents			
Hydrogen <sup>c</sup>	Kunz et al. (2007) <sup>[1]</sup>	14 - 700	300	14
Oxygen	Span & Wagner (2003)[10]	54 – 500e	100	12
Carbon monoxide	Lemmon & Span (2006)[11]	68 - 400	100	12
Water	Kunz et al. (2007)[1]	273 - 1 273	100	16
Hydrogen sulfide	Lemmon & Span (2006)[11]	187 - 760	170	12
Helium <sup>d</sup>	Kunz et al. (2007)[1]	2,2 - 573	100	12
Argon	Span & Wagner (2003)[10]	83 - 520	100	12

The tabulated references document the equations for the residual part of the Helmholtz free energy of the pure substances. Equations for the isobaric heat capacity in the ideal-gas state taken from reference [5] were used to derive the Helmholtz free energy of the ideal gas for all components. Ranges listed are for volumetric properties only (density, compressibility factor, and vapour pressure). Components with a different lower limit for the ideal heat capacity than that listed in the table are:  $n-C_5$ ,  $i-C_5$ ,  $n-C_6$  and  $n-C_7$  (200 K), and  $H_2$  and  $O_2$  (100 K). See the publication of Jaeschke and Schley [5] for additional information on the limits of caloric properties.

#### 6.2 Binary mixtures

Table 3 summarizes the available data for volumetric and caloric properties of all binary mixtures considered in this method. Almost half of the data come from the 15 mixtures with binary specific or

b The equation can be extrapolated from the triple-point temperature of 216 K down to 90 K for the vapour phase.

The equation given in Kunz et al. [1] represents equilibrium-hydrogen; volumetric properties of normal-hydrogen and para-hydrogen are nearly the same as those for e-hydrogen. The heat capacity equation for the ideal gas reported by Jaeschke and Schley [5] is for n-H2. The difference in heat capacities between e-H2 and n-H2 are significant below about 200 K.

d Represents helium-4. The lower temperature limit of the equation of state is the lambda point at which helium I transitions to helium II for the saturated liquid.

The upper limit of the oxygen equation has been increased to 500 K based on recent validation.

generalized departure functions. The majority of the data (approximately 65 %) describe the ppT relation, approximately 25 % are vapour-liquid equilibrium state points, and about 9 % are caloric properties.

#### a) Specific departure functions

Table 4 lists the binary mixtures for which either binary specific or generalized departure functions were developed with sufficiently accurate and extensive data sets. Binary specific departure functions are available for the binary mixtures of methane with nitrogen, carbon dioxide, ethane, propane, and hydrogen, and of nitrogen with carbon dioxide and ethane (see also figure 2).

The experimental data used to test the equations developed for the binary mixtures cover a region from about 80 K (-315 °F) to 700 K (800 °F) at pressures up to 70 MPa (10150 psia) or more (e.g., 100 MPa (14500 psia) for methane–carbon dioxide and 750 MPa (110 000 psia) for methane–nitrogen). The data cover compositions from nearly 0 to 1 for all mixtures listed above; see [ $\underline{2}$ ] for further details.

#### b) Generalized departure function

A generalized departure function was developed for eight binary mixtures of secondary alkanes (see <u>Table 4</u>). The generalized departure function is used for the binary mixtures of methane with n-butane and isobutane, ethane with propane, n-butane, and isobutane, propane with n-butane and isobutane, and n-butane with isobutane.

In addition to the experimental data for these binary mixtures, selected volumetric and caloric properties for the well-measured mixtures methane–ethane and methane–propane were also used for the development of the generalized departure function. The final structure is mostly based on accurate and comprehensive data sets for the three binary systems methane–ethane, methane–propane, and methane–n-butane, with more than 6200 selected data points (about 1000 data points for the methane–n-butane binary system). The temperature range of the experimental data used to test the equations for the secondary binary alkane mixtures covers roughly 95 K to 600 K (–290 °F to 620 °F) at pressures up to 35 MPa (5075 psia) (70 MPa for methane–n-butane). The composition ranges from nearly 0 to 1 for all mixtures listed above; see [2] for further details.

#### c) No departure functions

No departure function was developed for the remaining binary mixtures (see Figure 2). These mixtures are either characterized by limited data, which do not permit the development of binary specific or generalized departure functions, or are of minor importance for the description of the thermodynamic properties of multi-component natural gases due to the small mole fractions of the respective components. Fitting the parameters of the reducing functions for density and temperature to selected experimental data is sufficient and yields good results for most of the binary mixtures. When sufficient experimental data are not available, the parameters of the reducing functions are either set to unity or modified (calculated) in such a manner that the critical parameters of the pure components are combined in a different way, which proved to be more suitable for certain binary subsystems (see also Annex E.1).

For the temperature, pressure, and composition ranges covered by the experimental data used to test the equations developed for the binary mixtures without a departure function, i.e., by adjusting the reducing functions for density and temperature only, see [2].

Table 3 — Summary of the available data for volumetric and caloric properties of binary mixtures

Data type	Number of data points	Temperature range	Pressure range	Composition range
		T/K	p/MPa	X
Density	51 442	66,9 - 800	0,00 - 1 027	0,00 - 1,00
Isochoric heat capacity	1 236	101 – 345	67,1 <i>P</i> – 902 <i>P</i>	0,01 - 0,84

Listed separately due to a different data format. Saturated liquid (and vapour) densities may also be tabulated as ordinary  $p\rho T$ .

Density in kg·m<sup>-3</sup> instead of pressure.

 Table 3 (continued)

Data type	Number of data points	Temperature range	Pressure range	Composition range
		T/K	p/MPa	X
Density	51 442	66,9 – 800	0,00 - 1 027	0,00 - 1,00
Speed of sound	2 819	157 – 450	0,00 - 1 971	0,01 - 0,96
Isobaric heat capacity	1 072	100 - 424	0,00 - 52,9	0,09 - 0,93
Enthalpy differences	1 804	107 - 525	0,00 - 18,4	0,05 - 0,90
Excess molar enthalpy	177	221 - 373	0,8 - 15,0	0,01 - 0,98
Saturated liquid density <sup>a</sup>	460	95,0 – 394	0,03 - 22,1	0,00 - 1,00
VLE data	20 161	15,5 – 700	0,00 - 422	0,00 - 1,00
Total	79 171	15,5 - 800	0,00 - 1 971	0,00 - 1,00

<sup>&</sup>lt;sup>a</sup> Listed separately due to a different data format. Saturated liquid (and vapour) densities may also be tabulated as ordinary  $p\rho T$ .

Table 4 — Binary mixtures for which specific or generalized departure functions were developed

Binary mixture	Type of departure function	Number of terms	Type of terms <sup>a</sup>
Methane-Nitrogen	Binary specific	9	P (2), E (7)
Methane–Carbon dioxide	Binary specific	6	P (3), E (3)
Methane-Ethane	Binary specific	12	P (2), E (10)
Methane-Propane	Binary specific	9	P (5), E (4)
Methane-n-Butane	Generalized	10	P
Methane-Isobutane	Generalized	10	P
Methane-Hydrogen	Binary specific	4	P
Nitrogen–Carbon dioxide	Binary specific	6	P (2), E (4)
Nitrogen-Ethane	Binary specific	6	P (3), E (3)
Ethane-Propane	Generalized	10	P
Ethane-n-Butane	Generalized	10	P
Ethane–Isobutane	Generalized	10	P
Propane-n-Butane	Generalized	10	P
Propane-Isobutane	Generalized	10	P
n-Butane-Isobutane	Generalized	10	P

<sup>&</sup>lt;sup>a</sup> "P" indicates polynomial terms, and "E" indicates terms composed of a polynomial and exponential expression according to equation (12). The numbers in parentheses indicate the respective number of terms Kpol and Kexp.

 $<sup>\</sup>rho$  Density in kg·m<sup>-3</sup> instead of pressure.

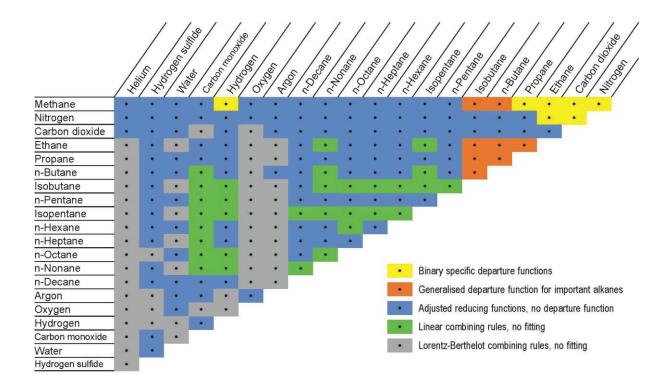


Figure 2 — Overview of the 210 binary combinations that result from the 21 natural gas components for the development of the GERG-2008 equation of state. The diagram shows the different formulations developed for the binary mixtures.

#### 6.3 Natural gases

Table 5 — Ranges of applicationa

	Normal range Full ra		
Pressure (absolute)	0 < p / MPa ≤ 35	0 < p / MPa ≤ 70	
Temperature	$90 \le T / K \le 450$	$60 \le T / K \le 700$	
Pressure (absolute)	0 < p / psia ≤ 5075	0 < p / psia ≤ 10150	
Temperature	-298 ≤ T / °F ≤ 350	-352 ≤ T / °F ≤ 800	

a Ranges listed are for volumetric properties only (density, compressibility factor, and vapour pressure). Other properties have a lower temperature limit of approximately 200 K. See the publication of Jaeschke and Schley[5] for additional information on the limits of the caloric equations.

Pipeline quality natural gas is generally described as a mixture of various components with mole fractions that fall within the ranges given in the third column of <u>Table 6</u>. The method described in ISO 20765-1 applies only to pipeline quality natural gases at temperatures and pressures found in transmission and distribution operations. The method given in this standard not only applies to these conditions but also to an intermediate composition range of natural gas as given in the fourth column of <u>Table 6</u>. Accurate and extensive experimental data sets are available within the composition ranges listed in the table and have been used to validate the quality of the method presented here (see <u>7.4</u>). Possible trace components of natural gases, and details of how to deal with these, are discussed in <u>Annex F</u>. This method is not applicable where the total of all trace components exceeds 0,000 5 mole fraction.

Beyond the intermediate quality range is the full range that covers all compositions for all mixture components, i.e., from 0 to 1. Due to limited data, results obtained for multi-component mixtures in the full range of application should be carefully assessed. The method can also be used for temperatures

above 700 K (800 °F) and pressures above 70 MPa (10150 psia); however, due to the limited data it is difficult to estimate the uncertainty of the method (see <u>7</u>).

Table 6 — Mole fraction ranges for components of intermediate and pipeline quality natural gas

		Mole fractions for		
i	Component	Pipeline quality range	Intermediate quality range	
1	Methane	$0.7 \le x_{\text{CH4}} \le 1.00$	$0.3 \le x_{\text{CH4}} \le 1.00$	
2	Nitrogen	$0 \le x_{\rm N2} \le 0.20$	$0 \le x_{\rm N2} \le 0.55$	
3	Carbon dioxide	$0 \le x_{\rm CO2} \le 0.20$	$0 \le x_{CO2} \le 0.30$	
4	Ethane	$0 \le x_{\rm C2H6} \le 0.10$	$0 \le x_{\text{C2H6}} \le 0.25$	
5	Propane	$0 \le x_{\text{C3H8}} \le 0.035$	$0 \le x_{\text{C3H8}} \le 0.14$	
6 + 7	n-Butane + Isobutane <sup>a</sup>	$0 \le x_{\text{C4H10}} \le 0.015$	$0 \le x_{\text{C4H10}} \le 0.06$	
8 + 9	n-Pentane + Isopentanea	$0 \le x_{\text{C5H12}} \le 0.005$	$0 \le x_{\text{C5H}12} \le 0.005$	
10	n-Hexane	$0 \le x_{\text{C6H}14} \le 0.001$	$0 \le x_{\text{C6H}14} \le 0.002$	
11	n-Heptane	$0 \le x_{\text{C7H16}} \le 0,0005$	$0 \le x_{\text{C7H16}} \le 0.001$	
12 + 13 + 14	Octane+Nonane+Decanea	$0 \le x_{C8+} \le 0.0005$	$0 \le x_{C8+} \le 0.0005$	
15	Hydrogen	$0 \le x_{\rm H2} \le 0.10$	$0 \le x_{\rm H2} \le 0.40$	
16	Oxygen	$0 \le x_{02} \le 0,000 \ 2$	$0 \le x_{02} \le 0.02$	
17	Carbon monoxide	$0 \le x_{\rm CO} \le 0.03$	$0 \le x_{\rm CO} \le 0.13$	
18	Water	$0 \le x_{\text{H2O}} \le 0,000\ 15$	$0 \le x_{\rm H2O} \le 0.000$ 2	
19	Hydrogen sulfide	$0 \le x_{\text{H2S}} \le 0,000\ 2$	$0 \le x_{\text{H2S}} \le 0.27$	
20	Helium	$0 \le x_{\text{He}} \le 0.005$	$0 \le x_{\text{He}} \le 0.005$	
21	Argon	$0 \le x_{\rm Ar} \le 0.0002$	$0 \le x_{\rm Ar} \le 0.0005$	

a Indicates the sum of the mole fractions of the components may not exceed the specified value.

NOTE 1 The method described in this part of the standard is applicable even for the individual pure components with high accuracy (i.e., to within experimental uncertainty). The tested ranges of application for temperature and pressure are given in  $\underline{\text{Table 2}}$ .

NOTE 2 The accurate description of the thermodynamic properties of multi-component mixtures by the GERG-2008 equation of state is based on the accurate and wide-ranging equations for the binary subsystems, which were developed with experimental data that generally cover the entire composition range. Therefore, it can be expected that even multi-component natural gases of very unusual composition will be accurately described.

#### 7 Uncertainty of the equation of state

#### 7.1 Background

The uncertainties in this document are given at a 95 % confidence level (coverage factor k = 2). When ranges of uncertainties are given, the upper uncertainty value should be used unless further comparisons are made with the information given in references [1] and [2] to verify that the lower uncertainty value is valid for a particular application.

#### 7.2 Uncertainty for pure gases

#### 7.2.1 Natural gas main components

The estimated uncertainties in calculated density and speed of sound for the natural gas main components methane, nitrogen, carbon dioxide, and ethane are summarized in <u>Table 7</u>. For methane, more details are given in <u>Figures 3</u> and <u>4</u>; details for nitrogen, carbon dioxide, and ethane are given in [1],[2]. The

estimated uncertainties in gas phase density and speed of sound range from 0,03 % to 0,05 % over wide ranges of temperature (e.g., up to 450 K, 350 °F) and at pressures up to 30 MPa (4 350 psia). In the liquid phase at pressures up to 30 MPa (4350 psia), the estimated uncertainties in density range from 0,05 % to 0,1 %. At higher temperatures or pressures, the estimated uncertainties in calculated speed of sound are generally higher than in calculated density because of less accurate data.

#### 7.2.2 Secondary alkanes

The estimated uncertainties in calculated density, speed of sound, and isobaric heat capacity for the secondary alkanes propane, n-butane, isobutane, n-pentane, isopentane, n-hexane, n-heptane, n-octane, n-nonane, and n-decane are summarized in  $\underline{\text{Table 8}}$ . For calculated densities, an uncertainty of 0,2 % was estimated, whereas calculations of speed of sound and isobaric heat capacity have estimated uncertainties between 1 % and 2 %.

Table 7 — Uncertainties of the equations of state for the natural gas main components methane, nitrogen, carbon dioxide, and ethane with regard to different thermodynamic properties.

	Density	Speed of sound
Gas phase <sup>a</sup>		
<i>p</i> ≤ 30 MPa (4350 psia)	0,03 % - 0,05 %	0,03 % - 0,05 %b
<i>p</i> > 30 MPa <sup>c</sup> (4350 psia)	0,1 % - 0,5 %	0,5 %d
Liquid phase		
<i>p</i> ≤ 30 MPa (4350 psia)	0,05 % - 0,1 %	0,5 % – 1,5 % <sup>e</sup>
p > 30 MPa <sup>c</sup> (4350 psia)	0,5 %f	1,5 %g
Saturated liquid	0,05 %h	
Saturated vapour	0,05 %	

- <sup>a</sup> for temperatures up to 450 K (350 °F); uncertainties at higher temperatures are given in <u>Figures 3</u> and <u>4</u> for methane and in Reference [2] for nitrogen, carbon dioxide, and ethane.
- b This uncertainty range is not valid over the entire temperature and pressure ranges specified in the table; further details are given in Figure 4 for methane and in Reference [2] for nitrogen, carbon dioxide, and ethane.
- States at pressures p > 100 MPa (14500 psia) are not included in the table due to their limited technical relevance, but can be obtained from Figures 3 and 4 for methane and from Reference [2] for nitrogen, carbon dioxide, and ethane.
- for methane; (1-2) % for nitrogen, (0,5-1) % for carbon dioxide, and  $\geq 2$  % for ethane; see Reference [2].
- for methane; 1 % ( $p \le 20$  MPa, 2900 psia) for nitrogen, (0,5 1) % for carbon dioxide, and (1 2) % for ethane; see Reference [2].
- f for methane, nitrogen, and ethane; 1 % for carbon dioxide.
- g for methane;  $\geq 2$  % (p > 20 MPa, 2900 psia) for nitrogen, (0,5 1) % for carbon dioxide, and  $\geq 2$  % for ethane; see Reference [2].
- for methane, nitrogen, and carbon dioxide; 0,1 % for ethane.

Table 8 — Uncertainties of the equations of state for propane, n-butane, isobutane, n-pentane, isopentane, n-hexane, n-heptane, n-octane, n-nonane, n-decane, hydrogen sulfide, oxygen, and argon for various thermodynamic properties

Pressure range			Uncert	ainty in		
	$\rho(T,p)$	w(T,p)	$c_p(T,p)$	$p_s(T)$	$\rho'(T)$	$\rho''(T)$
<i>p</i> ≤ 30 MPa (4350 psia) <sup>a</sup>	0,2 %b	(1 – 2) % <sup>c</sup>	(1 – 2) % <sup>c</sup>	0,2 %d	0,2 %	0,4 %d,e
p > 30 MPa (4350 psia) <sup>f</sup>	0,5 %	2 %	2 %	-	-	-

a Larger uncertainties exist in the extended critical region.

States at pressures p > 100 MPa (14 500 psia) are not included due to their limited technical relevance.

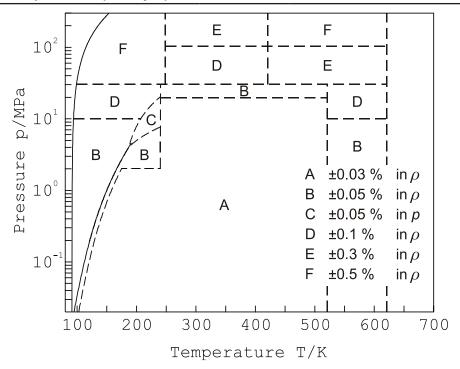


Figure 3 — Uncertainty diagram for densities of methane calculated from the GERG-2008 equation of state

b In the extended critical region,  $\Delta p/p$  is used instead of  $\Delta \rho/\rho$ .

<sup>1 %</sup> at gaseous and gas-like supercritical states, 2 % at liquid and liquid-like supercritical states.

d Larger relative uncertainties may result for small vapour pressures and their corresponding saturated vapour densities.

e Combination of the uncertainties of the gas densities and vapour pressures; experimental data with uncertainties of this level are available for only a few substances.

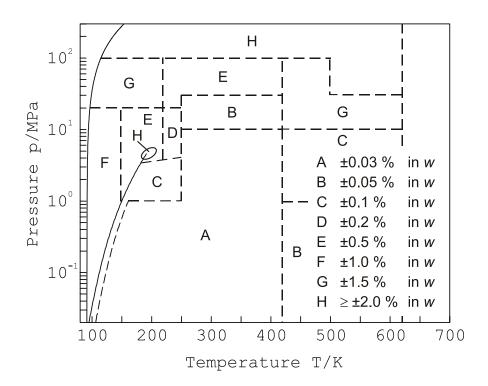


Figure 4 — Uncertainty diagram for speeds of sound of methane calculated from the GERG-2008 equation of state

#### 7.2.3 Other secondary components

For oxygen and argon, the estimated uncertainties in calculated density, speed of sound, and isobaric heat capacity are as stated in <u>Table 8</u> for the secondary alkanes.

The estimated uncertainties in calculated density for the other secondary components, namely hydrogen, carbon monoxide, hydrogen sulfide, and helium, at supercritical temperatures and for pressures up to 30 MPa (4350 psia) are less than 0,2 % and at higher pressures less than 0,5%. In general, higher uncertainties may occur in the liquid phase and for other thermodynamic properties.

For hydrogen at temperatures above 270 K (26 °F) and pressures up to 30 MPa (4350 psia), the uncertainty in calculated density is less than 0,1 %. At pressures above 30 MPa, the uncertainty in density is slightly higher, approximately (0,2 – 0,3) %. The equations for hydrogen and helium are designed to be valid at low (absolute) temperatures, which occur in their sub- and supercritical regions. This is of particular importance for the mixture model presented here since the reduced temperature range  $1,2 \le T/T_c \le 1,8$  corresponds to the region where the highest accuracy in the description of thermodynamic properties of typical natural gases is demanded.

For water, calculated liquid densities and vapour pressures have an estimated uncertainty of 0.2%. Other properties have higher uncertainties as detailed in [1],[2].

#### 7.3 Uncertainty for binary mixtures

The most accurate binary mixture data were used for the development of the GERG-2008 equation of state. However, in regions where these data are not available, less accurate data were also taken into account for the development and assessment of the equation. Experimental data for multi-component mixtures were used for the validation of the quality of the equation only. The total uncertainties of the most accurate experimental binary and multi-component mixture data with respect to selected thermodynamic properties are listed in <u>Table 9</u>. The tabulated values represent the lowest uncertainties possible that can be achieved by the mixture model presented here. The corresponding experimental results are based on modern measurement techniques, which fulfil present quality standards. In contrast to the experimental uncertainties given for pure fluid properties measured using state-of-the-art techniques, the experimental

uncertainties estimated for the properties of mixtures measured with the same apparatuses are, in general, higher due to the contribution of the uncertainty in the mixture composition.

Over wide ranges of temperature, pressure, and composition, the uncertainties tabulated below are mostly valid for those binary systems where binary specific departure functions were developed, see 6.2 and Table 4. Due to limited experimental data (e.g., accurate speed of sound measurements are available for only a few binary systems), the uncertainties are partly valid for the remaining binary mixtures, including those binary systems for which a generalized departure function was developed, see 6.2 and [1].

General estimates of the uncertainties of the GERG-2008 equation of state in the description of selected thermodynamic properties are given in <u>Table 10</u>. The different binary mixtures are distinguished by adjusted reducing functions with a binary specific departure function, adjusted reducing functions with a generalized departure function, or only adjusted reducing functions (without a departure function). Uncertainty values are given for different pressures, temperatures, and (approximate) reduced temperature ranges.

Table 9 — Relative experimental uncertainties of the most accurate binary and multicomponent mixture data

Data type	Property	Relative uncertainty				
Density (gas phase)	$\Delta \rho/ ho$	≤ (0,05 - 0,1) %				
Density (liquid phase)	$\Delta \rho/ ho$	≤ (0,1 - 0,3) %				
Isochoric heat capacity	$\Delta c_{\nu}/c_{\nu}$	≤ (1 − 2) %				
Speed of sound (gas phase)	$\Delta w/w$	≤ (0,05 – 0,1) %				
Isobaric heat capacity	$\Delta c_p/c_p$	≤ (1 − 2) %				
Enthalpy differences (gas phase)	$\Delta(\Delta h)/\Delta h$	≤ (0,2 - 0,5) %				
Saturated liquid density $\Delta \rho'/\rho' \leq (0.1 - 0.2) \%$						
VLE data $\Delta p_{\rm S}/p_{\rm S}$ $\leq$ (1 – 3) %						
NOTE $\Delta h$ indicates a difference between two state points, $h(T_2,p_2) - h(T_1,p_1)$ .						

Table 10 — Uncertainty of the GERG-2008 equation of state in the description of selected volumetric and caloric properties of different binary mixturesa

Mixture region <sup>b</sup>	Adjusted reducing functions with a		Only adjusted reducing	
	binary specific departure function	generalized departure function	functions (no depar- ture function)	
Gas phase 0 – 30 MPa (4 350 psia) $1,2 \le T/T_r \le 1,4$	$\Delta \rho/\rho \le 0.1 \%$	$\Delta \rho/\rho \leq (0.1-0.2) \%$	$\Delta \rho / \rho \le (0.5 - 1) \%$	
Gas phase 0 – 30 MPa (4 350 psia) $1.4 \le T/T_r \le 2.2$	$\Delta \rho/\rho \le 0.1 \%$	$\Delta \rho/\rho \le 0.1 \%$	$\Delta \rho/\rho \leq (0,3-0,5) \%$	
Gas phase 0 – 20 MPa (2 900 psia) $1,2 \le T/T_r \le 1,4$	$\Delta w/w \le 0.1 \%$	$\Delta w/w \le 0.5 \%$	$\Delta w/w \le 1 \%$	
Gas phase 0 – 20 MPa (2 900 psia) $1.4 \le T/T_r \le 2.2$	$\Delta w/w \le 0.1 \%$	$\Delta w/w \le 0.3 \%$	$\Delta w/w \le 0.5 \%$	
Saturated liquid state 100 K $\leq$ T $\leq$ 140 K (-280 °F $\leq$ T $\leq$ -208 °F)	$\Delta \rho' / \rho' \le (0.1 - 0.2) \%$	$\Delta \rho' / \rho' \le (0, 2 - 0, 5) \%$	$\Delta\rho'/\rho' \leq (0,5-1) \%$	
Liquid phase 0 – 40 MPa (5 800 psia) $T/T_r \le 0.7$	$\Delta \rho / \rho \le (0,1-0,3) \%$	$\Delta \rho/\rho \le (0.2-0.5) \%$	$\Delta \rho/\rho \leq (0.5-1)\%$	

Table 10 (continued)

<u> </u>	Only adjusted reducing	
binary specific generalized departure function		functions (no depar- ture function)

<sup>&</sup>lt;sup>a</sup> The relative uncertainty in isobaric and isochoric heat capacity is estimated to be less than (1 - 2) % in the homogeneous gas, liquid, and supercritical regions independent of the type of binary equation.

From Table 10 it is evident that binary systems with a binary specific departure function generally have the lowest uncertainty for the different properties as compared to the other binary systems with either a generalized departure function or only modified reducing parameters. Gas phase densities and speeds of sound have uncertainties of  $\leq 0.1$  % for binary mixtures with a binary specific departure function. The relative uncertainty in isobaric and isochoric heat capacity is estimated to be less than (1-2) % in the homogeneous gas, liquid, and supercritical regions independent of the type of developed binary equation.

#### 7.4 Uncertainty for natural gases

The GERG-2008 wide-range equation of state for natural gases and other (multi-component and binary) mixtures, consisting of the components listed in Table 6, is valid in the gas phase, in the liquid phase, in the supercritical region, and for vapour-liquid equilibrium states. For natural gases and similar mixtures, normal and full ranges were defined. The extrapolation to temperatures and pressures yields reasonable results even far beyond the full range of application. The estimated uncertainties for the different ranges of application, as described below, are based on the representation of the available experimental data for various thermodynamic properties of natural gases and other multi-component mixtures by the GERG-2008 equation of state as summarized in Table 11.

In general, there are no restrictions in the composition range of binary and multi-component mixtures. But, since the estimated uncertainty of the GERG-2008 equation of state is based on the experimental data used for the development and evaluation of the equation, the uncertainty is mostly unknown for composition ranges not covered by experimental data. The data situation allows for well-founded uncertainty estimates only for selected properties and parts of the fluid  $p\rho T$  surface.

Most of the available experimental data for multi-component mixtures describe the  $p\rho T$  relation of natural gases and similar mixtures in the gas phase. The majority of these data cover the temperature range 270 K  $\leq T \leq$  350 K at pressures up to 30 MPa<sup>[1],[2]</sup> and were measured for pipeline quality natural gas. There are a number of additional experimental data available that define the composition range of intermediate quality natural gas, e.g., measurements on rich natural gases with comparatively high content of carbon dioxide, ethane, propane, and n-butane; see Table 6 for the composition ranges defined for pipeline quality and intermediate quality natural gases. As mentioned in 6.3, pipeline quality natural gases are a subset of the intermediate quality natural gases.

Table 11 — Summary of the available data for volumetric and caloric properties of natural gases and other multi-component mixtures

		Covered	Maximum	
Data type <sup>a</sup>	Number of data points	Temperature	Pressure	number of
	data points	T/K	p/MPa	components
Density	21769	91,0 – 573	0,03 - 99,9	18
Speed of sound	1337	213 - 414	0,00 - 70,0	13
Isobaric heat capacity	325	105 - 350	0,5 - 30,0	8
Enthalpy differences	1166	105 - 422	0,2 - 16,5	10

Further data not included in Table 11 were used to validate the quality of the GERG-2008 equation of state, e.g., recent dew-point measurements for a number of different natural gases and other multi-component mixtures, see [1], [2].

<sup>&</sup>lt;sup>b</sup> For a typical lean natural gas or liquid natural gas, temperatures of 140 K (–208 °F), 250 K (–10 °F), 300 K (80 °F), and 350 K (170 °F) correspond to reduced temperatures  $T/T_{\rm r}(\bar{x}) = 1/\tau$  of about: 0,7; 1,3; 1,5; and 1,8.

		Covered	Maximum	
Data type <sup>a</sup>	Number of data points	Temperature	Pressure	number of
		T/K	p/MPa	components
Saturated liquid density	124	105 – 251	0,04 - 3,2	8
VLE data	2284	77,8 – 450	0,1 - 27,6	4
Total	27005	77,8 – 573	0,00 - 99,9	18

Further data not included in <u>Table 11</u> were used to validate the quality of the GERG-2008 equation of state, e.g., recent dew-point measurements for a number of different natural gases and other multi-component mixtures, see [1], [2].

#### 7.4.1 Uncertainty in the normal and intermediate ranges of applicability of natural gas

The normal range of applicability of natural gas covers the temperature range 90 K  $\leq$   $T \leq$  450 K for pressures up to 35 MPa, see <u>6.3</u>, <u>Table 5</u>. This range corresponds to the use of the equation in both standard and advanced technical applications using natural gases and similar mixtures, e.g., pipeline transport, natural gas storage, and improved processes with liquefied natural gas. Estimated uncertainties for the composition subsets "pipeline quality natural gas" and "intermediate quality natural gas" are summarized in <u>Table 12</u>.

Table 12 — Uncertainty of the GERG-2008 equation of state in the description of selected volumetric and caloric properties

Pipeline quality natural gas	Temperature region	Pressure region	Uncertainty
Density, gas phase	$250 \le T/K \le 450$	<i>p</i> /MPa ≤ 35	$\Delta \rho / \rho \le 0.1 \%$
Density, liquid phase	$100 \le T/K \le 140$	<i>p</i> /MPa ≤ 40	$\Delta \rho / \rho \le (0.1 - 0.5) \%$
Saturated liquid density	$100 \le T/K \le 140$		$\Delta \rho' / \rho' \le (0.1 - 0.3) \%$
Speed of sound, gas phase	$250 \le T/\mathrm{K} \le 270$	<i>p</i> /MPa ≤ 12	$\Delta w/w \le 0.1 \%$
	$270 \le T/\mathrm{K} \le 450$	<i>p</i> /MPa ≤ 20	$\Delta w/w \le 0.1 \%$
	$250 \le T/\mathrm{K} \le 270$	12 ≤ <i>p</i> /MPa ≤ 20	$\Delta w/w \le (0.2 - 0.3) \%$
	$250 \le T/\mathrm{K} \le 450$	20 ≤ <i>p</i> /MPa ≤ 30	$\Delta w/w \le (0.2 - 0.3) \%$
Enthalpy differences (gas)	$250 \le T/K \le 350$	<i>p</i> /MPa ≤ 20	$\Delta(\Delta h)/\Delta h \le (0,2-0,5) \%$
Enthalpy differences, liquid pha	se		$\Delta(\Delta h)/\Delta h \le (0.5 - 1.0) \%$
Isobaric and isochoric heat capa	cities, gas and liquid phase	S	$\Delta c_p/c_p$ or $\Delta c_v/c_v \approx (1-2)\%$
Intermediate rangea	Molar mass	Pressure region	Uncertainty
Density, gas phase	$M \le 26 \text{ kg} \cdot \text{kmol}^{-1}$	<i>p</i> /MPa ≤ 30	$\Delta \rho / \rho \le 0.1 \%$
	<i>M</i> > 26 kg⋅kmol <sup>-1</sup>	<i>p</i> /MPa ≤ 30	$\Delta \rho / \rho \le (0.1 - 0.3) \%^{b}$

For rich natural gases, i.e., for natural gas mixtures that contain comparatively large amounts of carbon dioxide, ethane, propane, and further secondary alkanes, the tested temperature range is  $280 \text{ K} \le T \le 350 \text{ K}$  (44 °F  $\le T \le 70$  °F).

#### 7.4.1.1 Pipeline quality natural gas

Density data in the gas phase for pipeline quality natural gases are described by the equation with an uncertainty of  $\Delta\rho/\rho \le 0.1$  % over the temperature range 250 K  $\le T \le 450$  K (-10 °F  $\le T \le 350$  °F) and for pressures up to 35 MPa (5075 psia). The uncertainty in speed of sound is likewise less than 0.1%. However, due to limited experimental data, this uncertainty is restricted to pressures below 20 MPa (2900 psia), and at temperatures below 270 K (26 °F) it is restricted to pressures below 12 MPa

<sup>&</sup>lt;sup>b</sup> For mixtures with molar masses  $M > 30 \text{ kg} \cdot \text{kmol}^{-1}$  and compositions within the limits stated in <u>Table 6</u>, the upper uncertainty value in density is estimated to be 0,5 %.

(1740 psia). The most accurate liquid or saturated liquid density data are described to within 0,1 % to 0,3 %, which is in agreement with the estimated experimental uncertainty of the measurements.

#### 7.4.1.2 Intermediate quality natural gas

This quality range of natural gases expands the composition range of the pipeline quality natural gases. The intermediate composition range is almost identical to the composition range covered by the available experimental natural gas and similar multi-component mixture data, including several data sets for natural gases containing synthetic mixtures, ternary mixtures of natural gas main components, and rich natural gases. Rich natural gases contain large amounts of carbon dioxide (up to 0,20 mole fraction), ethane (up to 0,18), propane (up to 0,14), n-butane (up to 0,06), n-pentane (0,005), and n-hexane (0,002).

For mixtures that fall within the intermediate composition range defined in Table 6, the estimated uncertainty in gas phase density is  $\leq 0.1$  % for molar masses  $M \leq 26$  kg·kmol<sup>-1</sup>; see equation (18) for the calculation of the molar mass from the given mixture composition. For mixtures with molar masses M > 26 kg·kmol<sup>-1</sup>, the uncertainty in gas phase density is 0.1 % to 0.3 %. For other thermodynamic properties, well-founded estimates of uncertainty cannot be given due to the limited data situation.

NOTE 1 For rich natural gases, the lower temperature limit is increased because dew point temperatures are considerably higher for these types of mixtures, which contain comparatively large amounts of carbon dioxide, ethane, propane, and the further secondary alkanes.

NOTE 2 Within the mole fraction limits defined for pipeline quality natural gas, the molar mass of any mixture will always be lower than  $26 \text{ kg} \cdot \text{kmol}^{-1}$ .

#### 7.4.2 Uncertainty in the full range of applicability, and calculation of properties beyond this range

The full range of application covers temperatures of  $60 \text{ K} \le T \le 700 \text{ K}$  ( $-352 \text{ °F} \le T \le 800 \text{ °F}$ ) and pressures up to 70 MPa (10150 psia). The uncertainty of the equation in gas phase density at temperatures and pressures within this range is roughly (0,2 – 0,5) %, except for states near the critical region. For certain mixtures, the full range of application covers temperatures of T > 700 K (800 °F) and pressures of p > 70 MPa (10150 psia). For example, the equation accurately describes gas phase density data of air to within  $\pm (0,1-0,2)$  % at temperatures up to 900 K and pressures up to 90 MPa. For other thermodynamic properties, well-founded estimates of uncertainty cannot be given due to the limited data situation outside the normal range of application.

When larger uncertainties are acceptable, tests have shown that the equation can be reasonably used outside the full range of application. For example, density data (frequently of questionable and low accuracy) for certain binary mixtures are described to within  $\pm (0.5 - 1)$  % at pressures up to 100 MPa and more.

#### 7.5 Uncertainties in other properties

For pipeline quality natural gas, similar types of properties will have similar uncertainties to those listed in <u>Table 12</u>. For example, energies and entropies will have uncertainties similar to those in enthalpy, and uncertainties in Joule-Thomson values will be similar to those in heat capacities.

#### 7.6 Impact of uncertainties of input variables

The user should recognize that uncertainties in the input variables, usually pressure, temperature and composition in mole fractions, will have additional effects upon the uncertainty of any calculated result. The uncertainties given so far for calculated results assume that the input data are exact. In any particular application where the additional uncertainty may be of importance, the user should carry out sensitivity tests to determine its magnitude.

#### 8 Reporting of results

Results for the thermodynamic properties shall be given with the number of significant figures that are appropriate for the uncertainties listed in <u>Tables 7</u> through <u>10</u>. The report shall identify the temperature,

## BS ISO 20765-2:2015 **ISO 20765-2:2015(E)**

pressure (or density), and detailed composition to which the results refer. The method of calculation used shall be identified by reference to ISO 20765-2 Natural gas – Calculation of Thermodynamic Properties.

For example, for pure methane at 25 °C, 0,101 325 MPa, Z is calculated as 0,998 2511. Taking the expanded uncertainty from Table 7 to be that of the gas phase density, 0,03%, this is 0,000 299 48. This is rounded to two significant figures, 0,000 30, and Z is reported in accordance with this uncertainty, i.e., 0,998 25. Thus, one would report Z(methane, 298,15 K, 0,101 325 MPa) = 0,998 25 ±0,000 30 (95 % confidence).

As a second example, for pure liquid methane at its normal boiling point (111,66 K, 0,101 325 MPa), Z is calculated as 0,004 145 252. Taking the expanded uncertainty from Table 7 to be that of the liquid phase density, 0,05 %, this is 0,000 002 0726. This is rounded to two significant figures, 0,000 0021, and Z is reported in accordance with this uncertainty, i.e., 0,004 1453. Thus, one would report Z (methane, 111,66 K, 0,101 325 MPa) = 0,004 1453 ±0,000 0021 (95 % confidence).

For the validation of calculations and for subsequent calculations based on thermodynamic properties obtained using this standard, it may be appropriate to carry extra digits (see example calculations in  $\underline{\mathsf{Annex}\;\mathsf{G}}$ ).

## Annex A (normative)

### **Symbols and units**

Symbol	Meaning	First Use of Symbol	Units
а	Molar Helmholtz free energy	equation (1)	kJ/kmol
В	Second virial coefficient	equation (32)	m <sup>3</sup> /kmol
С	Density exponent	equation (9)	
$c_p$	Molar isobaric heat capacity	equation (26)	kJ/(kmol·K)
$c_v$	Molar isochoric heat capacity	equation (24)	kJ/(kmol·K)
С	Third virial coefficient	equation (33)	m <sup>6</sup> /kmol <sup>2</sup>
$C_{v}$	Specific isochoric heat capacity	Table 1	kJ/(kg·K)
$C_p$	Specific isobaric heat capacity	Table 1	kJ/(kg·K)
d	Density exponent	equation (9)	
D	Mass-based density	equation (19)	kg/m <sup>3</sup>
F	Mixture parameter	equation (11)	
g	Molar Gibbs free energy	equation (27)	kJ/kmol
h	Molar enthalpy	equation (25)	kJ/kmol
Н	Specific enthalpy	<u>Table 1</u>	kJ/kg
i	Serial number	equation (6)	
j	Serial number	equation (10)	
k	Coverage factor	Scope	
k	Serial number	equation (7)	
K	Number of terms	equation (9)	
m	Molar mass	equation (18)	kg/kmol
n	Coefficient	equation (7)	
N	Number of components in the mixture	equation (6)	
р	Pressure	equation (20)	MPa
R	Molar gas constant, R = 8,314 472,[4]	equation (7)	kJ/(kmol·K)
R*	Molar gas constant, $R^*$ = 8,314 51, used in [5]	equation (7)	kJ/(kmol·K)
S	Molar entropy	equation (23)	kJ/(kmol·K)
S	Specific entropy	Table 1	kJ/(kg·K)
t	Temperature exponent	equation (9)	
T	Temperature in K (ITS-90)	equation (1)	К
и	Molar internal energy	equation (22)	kJ/kmol
U	Specific internal energy	<u>Table 1</u>	kJ/kg
v	Molar volume	equation (20)	M <sup>3</sup> /kmol
W	Speed of sound	equation (28)	m/s
X	Mole fraction	equation (6)	
$\overline{X}$	Molar composition (vector of mole fractions)	equation (1)	

Symbol	Meaning	First Use of Symbol	Units
Z	Any property	equation (37)	
$\overline{Z}$	Compression factor	equation (21)	
Greek Syn	abols		1
α	Reduced molar Helmhotz free energy, $\alpha = a/(RT)$	equation (2)	
β	Parameter	equation (12)	
$\beta_T$	Parameter	equation (14)	
$\beta_V$	Parameter	equation (13)	
γ	Parameter	equation (12)	
γτ	Parameter	equation (14)	
γν	Parameter	equation (13)	
δ	Reduced density, $\delta = \rho/\rho_{\rm r}$	equation (2)	
φ	Isothermal throttling coefficient	equation (30)	m³/kmol
ε	Parameter	equation (12)	
д	Partial derivative	equation (15)	
Δα	Departure function for the reduced molar Helmholtz free energy	equation (5)	
η	Parameter	equation (12)	
θ	Parameter	equation (7)	
К	Isentropic exponent	equation (31)	
μ	Joule-Thomson coefficient	equation (29)	К/МРа
ρ	Molar density	equation (1)	kmol/m <sup>3</sup>
τ	Inverse reduced temperature, $\tau = T_r/T$	equation (2)	
Inferior in	dices		
С	At the critical point	equation (7)	
Exp	Exponential term	equation (9)	
i	Serial number	equation (6)	
j	Serial number	equation (10)	
k	Serial number	equation (7)	
0	Property of the pure substance	equation (5)	
Pol	Polynomial term	equation (9)	
r	Reducing property	equation (3)	
S	At saturation (phase equilibrium)	Table 8	
δ	Partial derivative with respect to $\delta$	equation (15)	
τ	Partial derivative with respect to $ au$	equation (15)	
0	Reference state, $T_0$ = 298,15 K, $p_0$ = 0,101325 MPa	equation (B.4)	
Superior i	ndices		
0	Ideal-gas state	equation (1)	
r	Residual part	equation (1)	
,	Saturated liquid state	<u>Table 8</u>	
u .	Saturated vapour state	Table 8	

## **Annex B**

(normative)

## The reduced Helmholtz free energy of the ideal gas

## B.1 Calculation of the reduced Helmholtz free energy of the ideal gas

For a pure component, the Helmholtz free energy of the ideal gas is given by

$$a^{\circ}(\rho,T) = h^{\circ}(T) - RT - Ts^{\circ}(\rho,T)$$
(B.1)

For the ideal gas, the enthalpy  $h^0$  is a function of temperature only, whereas the entropy  $s^0$  depends on temperature and density. Both properties can be expressed in terms of the ideal-gas isobaric heat capacity  $c_p^0$  as follows:

$$h^{0}(T) = \int_{T_{0}}^{T} c_{p}^{0} dT + h_{0}^{0}$$
(B.2)

$$s^{o}(\rho,T) = \int_{T_{0}}^{T} \frac{c_{p}^{o} - R}{T} dT - R \ln \left(\frac{\rho}{\rho_{0}^{o}}\right) + s_{0}^{o}$$
(B.3)

When the above expressions for  $h^0(T)$  and  $s^0(\rho,T)$  are inserted into equation (B.1), one obtains

$$a^{o}(\rho,T) = \left[ \int_{T_{0}}^{T} c_{p}^{o} dT + h_{0}^{o} \right] - RT - T \left[ \int_{T_{0}}^{T} \frac{c_{p}^{o} - R}{T} dT - R \ln \left( \frac{\rho}{\rho_{0}^{o}} \right) + s_{0}^{o} \right]$$
(B.4)

where

all variables with the subscript "0" refer to an arbitrary reference state. The reference state of zero enthalpy and zero entropy is here adopted at  $T_0$  = 298,15 K and  $p_0$  = 0,101 325 MPa for the ideal gas. The integration constants  $h_0^0$  and  $s_0^0$  are then determined so as to conform to this definition. The reference density  $\rho_0^0$  is given by  $\rho_0^0 = p_0/(RT_0)$ .

In order to obtain an explicit equation for  $a^{o}(\rho,T)$ , an equation for the ideal-gas isobaric heat capacity  $c_{p}^{o}$  is needed. The ideal-gas isobaric heat capacity may be written as follows [1], [2],

$$\frac{c_p^o}{R} = 1 + n_3^o + \sum_{k=4,6} n_k^o \left( \frac{\vartheta_k^o \frac{T_c}{T}}{\sinh\left(\vartheta_k^o \frac{T_c}{T}\right)} \right)^2 + \sum_{k=5,7} n_k^o \left( \frac{\vartheta_k^o \frac{T_c}{T}}{\cosh\left(\vartheta_k^o \frac{T_c}{T}\right)} \right)^2$$
(B.5)

The values of the coefficients  $n_k^0$  and parameters  $\theta_k^0$  of equation (B.5) are given in <u>Table B.1</u>.

NOTE 1 The equation for the ideal-gas isobaric heat capacity is taken from [5] and given as a function of the temperature T. The values resulting from the parameter product  $\mathcal{G}_k^0$   $T_c$  (for k=4 to 7) are in agreement with the original parameters  $D^0$ ,  $F^0$ ,  $H^0$ , and  $J^0$ , respectively, and the coefficients  $n_k^0$  (for k=4 to 7) have identical values as the coefficients  $C^0$ ,  $E^0$ ,  $G^0$ , and  $I^0$ , respectively. The coefficient  $n_3^0 = B^0 - 1$ .

For a mixture at a given mixture density  $\rho$ , temperature T, and molar composition  $\overline{x}$ , the reduced Helmholtz free energy of the ideal gas is given as follows:

$$\alpha^{0}(\rho,T,\overline{x}) = \sum_{i=1}^{N} x_{i} \left[ \alpha_{0i}^{0}(\rho,T) + \ln x_{i} \right]$$
(B.6)

In this equation,  $\alpha_{0i}^{0}(\rho,T)$  is the dimensionless form of the Helmholtz free energy in the ideal-gas state of component i as given by

$$\alpha_{oi}^{o}(\rho,T) = \ln\left(\frac{\rho}{\rho_{c,i}}\right) + \frac{R^*}{R} \left[ n_{oi,1}^{o} + n_{oi,2}^{o} \frac{T_{c,i}}{T} + n_{oi,3}^{o} \ln\left(\frac{T_{c,i}}{T}\right) + \sum_{k=4,6} n_{oi,k}^{o} \ln\left|\sinh\left(\theta_{oi,k}^{o} \frac{T_{c,i}}{T}\right)\right| - \sum_{k=5,7} n_{oi,k}^{o} \ln\left|\cosh\left(\theta_{oi,k}^{o} \frac{T_{c,i}}{T}\right)\right| \right]$$
(B.7)

where

the term  $\sum x_i \ln x_i$  accounts for the entropy of mixing of the ideal-gas mixture.

NOTE 2 To indicate a pure substance as a component in a mixture, the inferior indices "o" (referring to pure substance) and "i" (referring to the considered component) are introduced.

It is important to observe that in equation (B.6)  $\alpha^0$  is a function of the molar density  $\rho$ , the temperature T, and the molar composition  $\overline{x}$ , where  $\rho$  is the molar density of the real mixture (i.e., not the molar density of the ideal gas). If the input variables are the absolute pressure p, the temperature T and the molar composition  $\overline{x}$ ,  $\rho$  may be calculated from the input variables as described in 5.

## B.2 Derivatives of the reduced Helmholtz free energy of the ideal gas

$$\alpha_{\delta}^{0} = \left(\frac{\partial \alpha^{0}}{\partial \delta}\right)_{\tau, \bar{X}} = \sum_{i=1}^{N} x_{i} \frac{\rho_{r}}{\rho_{c, i}} \left(\frac{\partial \alpha_{oi}^{0}}{\partial (\rho / \rho_{c, i})}\right)_{T} = \frac{\rho_{r}}{\rho}$$
(B.8)

$$\alpha_{\delta\delta}^{o} = \left(\frac{\partial^{2} \alpha^{o}}{\partial \delta^{2}}\right)_{\tau,\bar{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\rho_{r}}{\rho_{c,i}}\right)^{2} \left(\frac{\partial^{2} \alpha_{oi}^{o}}{\partial (\rho / \rho_{c,i})^{2}}\right)_{T} = -\frac{\rho_{r}^{2}}{\rho^{2}}$$
(B.9)

$$\alpha_{\delta\tau}^{o} = \left(\frac{\partial^{2}\alpha^{o}}{\partial\delta\partial\tau}\right)_{\overline{v}} = \sum_{i=1}^{N} x_{i} \frac{\rho_{r}}{\rho_{c,i}} \frac{T_{c,i}}{T_{r}} \left(\frac{\partial^{2}\alpha_{oi}^{o}}{\partial(\rho/\rho_{c,i})\partial(T_{c,i}/T)}\right) = 0$$
(B.10)

$$\alpha_{\tau}^{o} = \left(\frac{\partial \alpha^{o}}{\partial \tau}\right)_{\delta, \overline{\chi}} = \sum_{i=1}^{N} x_{i} \frac{T_{c,i}}{T_{r}} \left(\frac{\partial \alpha_{oi}^{o}}{\partial (T_{c,i}/T)}\right)$$
(B.11)

$$\left(\frac{\partial \alpha_{oi}^{o}}{\partial \left(T_{c,i}/T\right)}\right)_{\rho} = \frac{R^{*}}{R} \left[ n_{oi,2}^{o} + n_{oi,3}^{o} \frac{T}{T_{c,i}} + \sum_{k=4,6} n_{oi,k}^{o} \frac{\vartheta_{oi,k}^{o}}{\tanh \left(\vartheta_{oi,k}^{o} \frac{T_{c,i}}{T}\right)} - \sum_{k=5,7} n_{oi,k}^{o} \vartheta_{oi,k}^{o} \tanh \left(\vartheta_{oi,k}^{o} \frac{T_{c,i}}{T}\right) \right]$$
(B.12)

$$\alpha_{\tau\tau}^{o} = \left(\frac{\partial^{2} \alpha^{o}}{\partial \tau^{2}}\right)_{\delta, \overline{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{T_{c,i}}{T_{r}}\right)^{2} \left(\frac{\partial^{2} \alpha_{oi}^{o}}{\partial \left(T_{c,i} / T\right)^{2}}\right)_{\rho}$$
(B.13)

$$\left(\frac{\partial^{2}\alpha_{oi}^{o}}{\partial\left(T_{c,i}/T\right)^{2}}\right)_{\rho} = \frac{R^{*}}{R} \left[-n_{oi,3}^{o}\left(\frac{T}{T_{c,i}}\right)^{2} - \sum_{k=4,6} n_{oi,k}^{o} \frac{\left(9_{oi,k}^{o}\right)^{2}}{\left(\sinh\left(9_{oi,k}^{o}\frac{T_{c,i}}{T}\right)\right)^{2}} - \sum_{k=5,7} n_{oi,k}^{o} \frac{\left(9_{oi,k}^{o}\right)^{2}}{\left(\cosh\left(9_{oi,k}^{o}\frac{T_{c,i}}{T}\right)\right)^{2}}\right]$$
(B.14)

Table B.1 — Values of the coefficients and parameters in equation (B.7) for the 21 componentsa

k	$n_{{ m o}i,k}^{ m o}$	$\mathfrak{G}_{\text{o}i,k}^{\text{o}}$	k	$n_{\text{o}i,k}^{\text{o}}$	$\vartheta_{{\rm o}i,k}^{{ m o}}$
		Mei	hane		
1	19,597 508 817	_	5	0,004 60	0,936 220 902
2	-83,959 667 892		6	8,744 32	5,577 233 895
3	3,000 88	_	7	-4,469 21	5,722 644 361
4	0,763 15	4,306 474 465		·	
		Nit	rogen		
1	11,083 407 489	_	5	-0,146 60	5,393 067 706
2	-22,202 102 428	-	6	0,900 66	13,788 988 208
3	2,500 31	-	7	-	-
4	0,137 32	5,251 822 620			
		Carboi	ı dioxide		
1	11,925 152 758	_	5	-1,060 44	2,844 425 476
2	-16,118 762 264	_	6	2,033 66	1,589 964 364
3	2,500 02	_	7	0,013 93	1,121 596 090
4	2,044 52	3,022 758 166			
		Eal	nane		
1	24,675 437 527	EU	5	1,237 22	0,731 306 621
2					3,378 007 481
3	-77,425 313 760 3,002 63		7	13,197 4 -6,019 89	3,508 721 939
4	4,339 39	1,831 882 406	/	-0,019 09	3,300 /21 939
	,	,			
		Pro	pane		
1	31,602 908 195	-	5	3,197 00	0,543 210 978
2	-84,463 284 382	-	6	19,192 1	2,583 146 083
3	3,029 39	-	7	-8,372 67	2,777 773 271
4	6,605 69	1,297 521 801			
		n_D	utane		
1	20,884 143 364	- n-b	5	6,894 06	0,431 957 660
2	-91,638 478 026	_	6	24,461 8	4,502 440 459

 Table B.1 (continued)

k	$n_{{ m o}i,k}^{ m o}$	$\theta_{oi,k}^{\mathrm{o}}$	k	$n_{{\rm o}i,k}^{\rm o}$	$\mathfrak{G}_{\mathrm{o}i,k}^{\mathrm{o}}$
3	3,339 44	-	7	14,782 4	2,124 516 319
4	9,448 93	1,101 487 798			
		Isob	utane		
1	20,413 726 078	-	5	5,251 56	0,485 556 021
2	-94,467 620 036	-	6	25,142 3	4,671 261 865
3	3,067 14	-	7	16,138 8	2,191 583 480
4	8,975 75	1,074 673 199			
		n-Pe	entane		
1	28,587 336 516	_	5	21,836 0	1,789 520 971
2	-96,265 336 649	-	6	33,403 2	3,777 411 113
3	3,0	-	7	-	-
4	8,950 43	0,380 391 739			
		Isop	entane		
1	29,158 561 921	-	5	20,110 1	1,977 271 641
2	-111,216 048 893	_	6	33,168 8	4,169 371 131
3	3	_	7	-	_
4	11,761 8	0,635 392 636			
		n-H	exane		1
1	32,499 459 095	_	5	26,814 2	1,691 951 873
2	-103,869 150 117	-	6	38,616 4	3,596 924 107
3	3,0	_	7	-	-
4	11,697 7	0,359 036 667			
_		n-He	eptane		
1	37,237 679 271	-	5	30,470 7	1,548 136 560
2	-105,724 194 520	_	6	43,556 1	3,259 326 458
3	3,0	-	7	-	_
4	13,726 6	0,314 348 398			
4	42 442 402 464		ctane	22.0020	1 424 644 760
1	42,143 183 464	_	5	33,8029	1,431 644 769
2	-106,349 263 157	-	6	48,1731	2,973 845 992
3	3,0	0.270142540	7		_
4	15,686 5	0,279143540			
		%1	onar -		
1	46 722 62E 202		onane	20 122 50	1 270 507 150
1	46,723 625 203	_	5	38,123 50	1,370 586 158
2	-112,017 705 837	-	6	53,341 50	2,848 860 483

Table B.1 (continued)

k	$n_{oi,k}^{o}$	$\vartheta_{oi,k}^{\mathrm{o}}$	k	$n_{{ m o}i,k}^{ m o}$	$\mathfrak{G}^{\mathrm{o}}_{\mathrm{o}i,k}$
3	3,0	-	7	_	-
4	18,024 10	0,263 819 696			
	1				
		n-De	ecane		
1	50,353 023 354	-	5	43,493 10	1,353 835 195
2	-120,012 066 480	-	6	58,365 70	2,833 479 035
3	3,0	-	7	-	-
4	21,006 90	0,267 034 159			
		Hyd	rogen		
1	13,796 443 393	-	5	0,454 44	9,847 634 830
2	-175,864 487 294	-	6	1,560 39	49,765 290 750
3	1,479 06	-	7	-1,375 60	50,367 279 301
4	0,958 06	6,891 654 113			
		0x	ygen		
1	10,001 843 586	-	5	1,013 34	7,223 325 463
2	-14,996 095 135	-	6	-	-
3	2,501 46	-	7	-	-
4	1,075 58	14,461 722 565			
		Carbon	monoxide		
1	10,813 340 744	-	5	0,004 93	5,302 762 306
2	-19,834 733 959	-	6	-	-
3	2,500 55	-	7	-	-
4	1,028 65	11,669 802 800			
		W	ater		
1	8,216 535 516	-	5	0,987 63	1,763 895 929
2	-12,002 441 239	-	6	3,069 04	3,874 803 739
3	3,003 92	-	7	-	-
4	0,010 59	0,415 386 589			
		Hydrog	en sulfide		
1	9,336 197 742	-	5	1,002 43	2,270 653 980
2	-16,266 508 995	-	6	_	-
3	3,0	-	7	_	-
4	3,119 42	4,914 580 541			
		He	lium		
1	13,628 409 737	-	3	1,5	-
2	-143,470 759 602	-			

Table B.1 (continued)

k	$n_{{ m oi},k}^{ m o}$	$9^{\mathrm{o}}_{\mathrm{oi},k}$	k	$n_{{ m oi},k}^{ m o}$	$9^{\mathrm{o}}_{\mathrm{oi},k}$				
	Argon								
1	8,316 631 500	-	3	1,5	-				
2	2 -4,946 502 600 -								
a The v	The values of the coefficients and parameters are also valid for equation (7).								

# Annex C (normative)

# Values of critical parameters and molar masses of the pure components

Table C.1 — Critical parameters and molar masses of the 21 components

Component	Formula	$\rho_{c,i}/(\text{mol}\cdot\text{dm}^{-3})$	$T_{c,i}/K$	$M_i/(g\cdot mol^{-1})a$
Methane	CH <sub>4</sub>	10,139 342 719	190,564	16,042 46
Nitrogen	N <sub>2</sub>	11,183 9	126,192	28,013 4
Carbon dioxide	CO <sub>2</sub>	10,624 978 698	304,1282	44,009 5
Ethane	C <sub>2</sub> H <sub>6</sub>	6,870 854 540	305,322	30,069 04
Propane	C <sub>3</sub> H <sub>8</sub>	5,000 043 088	369,825	44,095 62
n-Butane	C <sub>4</sub> H <sub>10</sub>	3,920 016 792	425,125	58,122 2
Isobutane	C <sub>4</sub> H <sub>10</sub>	3,860 142 940	407,817	58,122 2
n-Pentane	C <sub>5</sub> H <sub>12</sub>	3,215 577 588	469,7	72,148 78
Isopentane	C <sub>5</sub> H <sub>12</sub>	3,271	460,35	72,148 78
n-Hexane	C <sub>6</sub> H <sub>14</sub>	2,705 877 875	507,82	86,175 36
n-Heptane	C <sub>7</sub> H <sub>16</sub>	2,315 324 434	540,13	100,201 94
n-Octane	C <sub>8</sub> H <sub>18</sub>	2,056 404 127	569,32	114,228 52
n-Nonane	C9H20	1,81	594,55	128,255 1
n-Decane	C <sub>10</sub> H <sub>22</sub>	1,64	617,7	142,281 68
Hydrogen	H <sub>2</sub>	14,94	33,19	2,015 88
Oxygen	02	13,63	154,595	31,998 8
Carbon monoxide	СО	10,85	132,86	28,010 1
Water	H <sub>2</sub> O	17,873 716 090	647,096	18,015 28
Hydrogen sulfide	H <sub>2</sub> S	10,19	373,1	34,080 88
Helium	Не	17,399	5,1953	4,002 602
Argon	Ar	13,407 429 659	150,687	39,948
a According to IUPAC	Гесhnical Report (	2006), <sup>[6]</sup> calculated using th	e exact atomic masses in th	ne IUPAC report.

## Annex D

(normative)

## The residual part of the reduced Helmholtz free energy

## D.1 Calculation of the residual part of the reduced Helmholtz free energy

The residual part  $\alpha$  r of the reduced Helmholtz free energy  $\alpha$  of the mixture is given by

$$\alpha^{r}(\delta,\tau,\overline{x}) = \alpha_{0}^{r}(\delta,\tau,\overline{x}) + \Delta\alpha^{r}(\delta,\tau,\overline{x})$$
(D.1)

where

$$\alpha_0^{\mathrm{r}}(\delta,\tau,\overline{x}) = \sum_{i=1}^N x_i \,\alpha_{0i}^{\mathrm{r}}(\delta,\tau) \tag{D.2}$$

and

$$\Delta \alpha^{\mathrm{r}}(\delta, \tau, \overline{x}) = \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_i x_j F_{ij} \alpha_{ij}^{\mathrm{r}}(\delta, \tau)$$
(D.3)

In equation (D.1), the first term on the right-hand side,  $\alpha_0^r(\delta,\tau,\overline{x})$ , describes the contribution of the residual parts of the reduced Helmholtz free energy of the pure substance equations of state, which are multiplied by the mole fraction of the corresponding substance and linearly combined using the reduced variables  $\delta$  and  $\tau$  of the mixture. The second term is the departure function,  $\Delta\alpha^r(\delta,\tau,\overline{x})$ , which is the summation over all binary specific and generalized departure functions. The variables  $\delta$  and  $\tau$  are the reduced mixture density and inverse reduced mixture temperature, respectively, as given by

$$\delta = \frac{\rho}{\rho_{\rm r}(\bar{x})} \tag{D.4}$$

and

$$\tau = \frac{T_{\rm r}(\overline{x})}{T} \tag{D.5}$$

where

 $\rho_r$  and  $T_r$  are reducing functions for the mixture density and mixture temperature depending on the molar composition of the mixture only (see 4.2.7 and Annex E.1).

## D.1.1 Derivatives of the residual part of the reduced Helmholtz free energy

The derivatives of the residual part  $\alpha^r(\delta, \tau, \overline{x})$  of the reduced Helmholtz free energy  $\alpha$  of the mixture with respect to the reduced mixture variables  $\delta$  and  $\tau$  are as follows:

$$\alpha_{\delta}^{r} = \left(\frac{\partial \alpha^{r}}{\partial \delta}\right)_{\tau, \overline{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\partial \alpha_{oi}^{r}}{\partial \delta}\right)_{\tau} + \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left(\frac{\partial \alpha_{ij}^{r}}{\partial \delta}\right)_{\tau}$$
(D.6)

$$\alpha_{\delta\delta}^{r} = \left(\frac{\partial^{2} \alpha^{r}}{\partial \delta^{2}}\right)_{\tau, \overline{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\partial^{2} \alpha_{oi}^{r}}{\partial \delta^{2}}\right)_{\tau} + \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left(\frac{\partial^{2} \alpha_{ij}^{r}}{\partial \delta^{2}}\right)_{\tau}$$
(D.7)

$$\alpha_{\delta\tau}^{r} = \left(\frac{\partial^{2}\alpha^{r}}{\partial\delta\partial\tau}\right)_{\overline{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\partial^{2}\alpha_{oi}^{r}}{\partial\delta\partial\tau}\right) + \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left(\frac{\partial^{2}\alpha_{ij}^{r}}{\partial\delta\partial\tau}\right)$$
(D.8)

$$\alpha_{\tau}^{r} = \left(\frac{\partial \alpha^{r}}{\partial \tau}\right)_{\delta_{r}\overline{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\partial \alpha_{oi}^{r}}{\partial \tau}\right)_{\delta} + \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left(\frac{\partial \alpha_{ij}^{r}}{\partial \tau}\right)_{\delta}$$
(D.9)

$$\alpha_{\tau\tau}^{r} = \left(\frac{\partial^{2} \alpha^{r}}{\partial \tau^{2}}\right)_{\delta, \overline{x}} = \sum_{i=1}^{N} x_{i} \left(\frac{\partial^{2} \alpha_{oi}^{r}}{\partial \tau^{2}}\right)_{\delta} + \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} x_{i} x_{j} F_{ij} \left(\frac{\partial^{2} \alpha_{ij}^{r}}{\partial \tau^{2}}\right)_{\delta}$$
(D.10)

## D.2 Calculation of the pure substance contribution to the residual part of the reduced Helmholtz free energy

The contribution of the residual parts of the reduced Helmholtz free energy of the pure substance equations of state  $\alpha_0^r$  to the residual part of the reduced Helmholtz free energy of the mixture is given by equation (D.2), where the residual part of the reduced Helmholtz free energy of component i,  $\alpha_{0i}^r(\delta,\tau)$ , (i.e., the residual part of the respective pure substance equation of state) is given by

$$\alpha_{oi}^{r}(\delta,\tau) = \sum_{k=1}^{K_{Pol,i}} n_{oi,k} \, \delta^{d_{oi,k}} \, \tau^{t_{oi,k}} + \sum_{k=K_{Pol,i}+1}^{K_{Pol,i}+K_{Exp,i}} n_{oi,k} \, \delta^{d_{oi,k}} \, \tau^{t_{oi,k}} \, e^{-\delta^{c_{oi,k}}}$$
(D.11)

For each of the pure substances, the equations for  $\alpha_{oi}^{\rm r}$  use the same basic structure, however, the number of terms differs. For the main components, the equations consist of 24 terms for methane, nitrogen, and ethane, and of 22 terms for carbon dioxide. For the secondary alkanes and the other secondary components (oxygen, carbon monoxide, hydrogen sulfide, helium, and argon), 12 terms are used, while the equations for hydrogen and water consist of 14 and 16 terms, respectively. The values of the coefficients  $n_{0i,k}$  and exponents  $d_{0i,k}$ ,  $t_{0i,k}$ , and  $c_{0i,k}$  for the 21 components are given in Annex D.2.2.

## D.2.1 Derivatives of $\alpha^{\, r}_{oi}(\delta, \tau)$ with respect to the reduced mixture variables $\pmb{\delta}$ and $\pmb{\tau}$

The derivatives of the residual part of the reduced Helmholtz free energy of component i,  $\alpha_{oi}^{r}(\delta,\tau)$ , (equation (D.11)) with respect to the reduced mixture variables  $\delta$  and  $\tau$  are as follows:

$$\begin{split} \delta \left( \frac{\partial \alpha_{\text{o}i}^{\text{r}}}{\partial \delta} \right)_{\tau} &= \sum_{k=1}^{K_{Pol,i}} n_{\text{o}i,k} d_{\text{o}i,k} \delta^{d_{\text{o}i,k}} \tau^{t_{\text{o}i,k}} \\ &+ \sum_{k=K_{Pol,i}+1}^{K_{Pol,i}+K_{\text{Exp},i}} n_{\text{o}i,k} \left( d_{\text{o}i,k} - c_{\text{o}i,k} \delta^{c_{\text{o}i,k}} \right) \delta^{d_{\text{o}i,k}} \tau^{t_{\text{o}i,k}} \exp \left( -\delta^{c_{\text{o}i,k}} \right) \\ \delta^{2} \left( \frac{\partial^{2} \alpha_{\text{o}i}^{\text{r}}}{\partial \delta^{2}} \right)_{\tau} &= \sum_{k=1}^{K_{Pol,i}} n_{\text{o}i,k} d_{\text{o}i,k} \left( d_{\text{o}i,k} - 1 \right) \delta^{d_{\text{o}i,k}} \tau^{t_{\text{o}i,k}} \\ &+ \sum_{k=K_{Pol,i}+1}^{K_{Pol,i}+K_{\text{Exp},i}} n_{\text{o}i,k} \left[ \left( d_{\text{o}i,k} - c_{\text{o}i,k} \delta^{c_{\text{o}i,k}} \right) \left( d_{\text{o}i,k} - 1 - c_{\text{o}i,k} \delta^{c_{\text{o}i,k}} \right) - c_{\text{o}i,k}^{2} \delta^{c_{\text{o}i,k}} \right] \delta^{d_{\text{o}i,k}} \tau^{t_{\text{o}i,k}} \exp \left( -\delta^{c_{\text{o}i,k}} \right) \end{split}$$
(D.13)

$$\delta \tau \left( \frac{\partial^{2} \alpha_{\text{o}i}^{\text{r}}}{\partial \delta \partial \tau} \right) = \sum_{k=1}^{K_{Pol,i}} n_{\text{o}i,k} d_{\text{o}i,k} t_{\text{o}i,k} \delta^{d_{\text{o}i,k}} \tau^{t_{\text{o}i,k}}$$

$$+ \sum_{\substack{k=K_{Pol,i}+K_{\text{Exp},i}\\k=K_{\text{Pol},i}+1\\\text{so 2015 - All rights reserved}} n_{\text{o}i,k} t_{\text{o}i,k} \left( d_{\text{o}i,k} - c_{\text{o}i,k} \delta^{c_{\text{o}i,k}} \right) \delta^{d_{\text{o}i,k}} \tau^{t_{\text{o}i,k}} \exp \left( -\delta^{c_{\text{o}i,k}} \right)$$

$$37$$

$$\tau \left( \frac{\partial \alpha_{0i}^{r}}{\partial \tau} \right)_{\delta} = \sum_{k=1}^{K_{Pol,i}} n_{0i,k} t_{0i,k} \delta^{d_{0i,k}} \tau^{t_{0i,k}} \\
+ \sum_{k=K_{Pol,i}+1}^{K_{Pol,i}+K_{Exp,i}} n_{0i,k} t_{0i,k} \delta^{d_{0i,k}} \tau^{t_{0i,k}} \exp\left(-\delta^{c_{0i,k}}\right) \\
\tau^{2} \left( \frac{\partial^{2} \alpha_{0i}^{r}}{\partial \tau^{2}} \right)_{\delta} = \sum_{k=1}^{K_{Pol,i}} n_{0i,k} t_{0i,k} \left( t_{0i,k} - 1 \right) \delta^{d_{0i,k}} \tau^{t_{0i,k}} \\
+ \sum_{k=K_{Pol,i}+K_{Exp,i}} n_{0i,k} t_{0i,k} \left( t_{0i,k} - 1 \right) \delta^{d_{0i,k}} \tau^{t_{0i,k}} \exp\left(-\delta^{c_{0i,k}}\right) \\
+ \sum_{k=K_{Pol,i}+1} n_{0i,k} t_{0i,k} \left( t_{0i,k} - 1 \right) \delta^{d_{0i,k}} \tau^{t_{0i,k}} \exp\left(-\delta^{c_{0i,k}}\right) \\$$
(D.16)

## **D.2.2** Coefficients and exponents of $\alpha_{oi}^{r}(\delta,\tau)$

Table D.1 — Values of the coefficients and exponents in equation (D.11) for methane, nitrogen, and ethane a,b

k	$n_{\text{OI}}$	i,k		$n_{0i,k}$			n <sub>oi,k</sub>	
	Meth	ane	Nitrogen			Ethane		
1	0,573 357 042 39	1 62	0,598 897 1	18 012 01		0,635 967 804 50714		
2	-0,167 606 875 2	37 30×10 <sup>1</sup>	-0,169 415	574 807 31×1	01	-0,173 77	9 817 854 59	)×10 <sup>1</sup>
3	0,234 052 918 34	9 16	0,245 797 3	61 917 18		0,289 140	609 262 72	
4	-0,219 473 763 4	34 41	-0,237 224	567 551 75		-0,337 14	2 768 456 94	
5	0,163 692 014 04	1 28×10 <sup>-1</sup>	0,179 549 1	87 151 41×10	<b>)</b> -1	0,224 059	646 995 61>	:10-1
6	0,150 044 063 89	92 80×10 <sup>-1</sup>	0,145 928 7	57 202 15×10	)-1	0,157 154	248 869 13>	10-1
7	0,989 904 894 92	29 18×10 <sup>−1</sup>	0,100 080 6	59 362 06		0,114 506	342 537 45	
8	0,583 827 709 29	00 55	0,731 571 1	53 855 32		0,106 120	493 797 45>	:101
9	-0,747 868 675 6	03 90	-0,883 722	723 363 66		-0,128 55	2 244 394 23	3×10 <sup>1</sup>
10	0,300 333 028 57	9 74	0,318 876 6	02 467 08		0,394 146 307 776 52		
11	0,209 855 438 06	55 68	0,207 664 9	17 287 99		0,313 909 246 820 41		
12	-0,185 901 511 3	30 61×10 <sup>-1</sup>	-0,193 793	154 541 58×	10-1	-0,215 92	2 771 172 47	′×10 <sup>-1</sup>
13	-0,157 825 583 3	90 49	-0,169 366	415 549 83		-0,217 236 665 649 05		
14	0,127 167 352 20	7 91	0,135 468 4	60 417 01		-0,289 995 744 394 89		
15	-0,320 197 438 9	43 46×10 <sup>-1</sup>	-0,330 667	120 953 07×	10-1	0,423 211 730 257 32		
16	-0,680 497 293 6	645 36×10 <sup>-1</sup>	-0,606 908	170 185 57×	10-1	0,464 341 002 592 60×10 <sup>-1</sup>		
17	0,242 914 128 53	7 36×10 <sup>-1</sup>	0,127 975 4	82 928 71×10	)-1	-0,131 38	3 983 297 41	
18	0,514 404 516 39	4 44×10 <sup>-2</sup>	0,587 436 6	41 072 99×1	0-2	0,114 928	503 643 68>	10-1
19	-0,190 849 497 3	35 32×10 <sup>-1</sup>	-0,184 519	519 719 69×1	0-1	-0,333 87	6 884 299 09	9×10 <sup>-1</sup>
20	0,552 296 772 41	2 91×10 <sup>-2</sup>	0,472 266 2	20 424 72×10	)-2	0,151 831	715 836 44>	10-1
21	-0,441 973 929 7	760 85×10−2	-0,520 240	-0,520 240 796 805 99×10 <sup>-2</sup>		-0,476 10	8 056 476 57	×10-2
22	0,400 614 167 08	4 29×10 <sup>-1</sup>	0,435 635 059 566 35×10 <sup>-1</sup>		0,469 171	662 778 85>	10-1	
23	-0,337 520 859 0	75×10 <sup>-1</sup>	-0,362 516	907 509 39×	10-1	-0,394 01	7 558 046 49	9×10 <sup>-1</sup>
24	-0,251 276 582 1	33 57×10 <sup>-2</sup>	-0,289 740	268 665 43×	10-2	-0,325 69	9 562 476 11	×10 <sup>-2</sup>
k	C <sub>Oi,k</sub>	$d_{0i,k}$	$t_{0i,k}$		k	$C_{0i,k}$	$d_{0i,k}$	$t_{0i,k}$
1	_	1	0,125		13	2	2	4,5

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rabie	ד.ע	(continued)

k	n <sub>oi</sub>	,k		$n_{0i,k}$			n <sub>oi,k</sub>	
2	-	1	1,125		14	2	3	4,75
3	_	2	0,375		15	2	3	5
4	_	2	1,125		16	2	4	4
5	_	4	0,625		17	2	4	4,5
6	_	4	1,5		18	3	2	7,5
7	1	1	0,625		19	3	3	14
8	1	1	2,625		20	3	4	11,5
9	1	1	2,75		21	6	5	26
10	1	2	2,125		22	6	6	28
11	1	3	2		23	6	6	30
12	1	6	1,75		24	6	7	16

<sup>&</sup>lt;sup>a</sup> The values of the coefficients and exponents are also valid for equation (9).

Table D.2 — Values of the coefficients and exponents in equation (D.11) for propane, n-butane, isobutane, n-pentane, isopentane, n-hexane, n-heptane, n-octane, n-nonane, n-decane, oxygen, carbon monoxide, hydrogen sulfide, and argon<sup>a-c</sup>.

k	$n_{0i,k}$	$n_{0i,k}$	$n_{0i,k}$
	Propane	n-Butane	Isobutane
1	0,104 039 731 073 58×10 <sup>1</sup>	0,106 262 774 114 55×10 <sup>1</sup>	0,104 293 315 891 00×10 <sup>1</sup>
2	-0,283 184 040 814 03×10 <sup>1</sup>	-0,286 209 518 283 50×10 <sup>1</sup>	-0,281 842 725 488 92×10 <sup>1</sup>
3	0,843 938 096 062 94	0,887 382 334 037 77	0,861 762 323 978 50
4	−0,765 595 918 500 23×10 <sup>-1</sup>	-0,125 705 811 553 45	-0,106 136 194 524 87
5	0,946 973 730 572 80×10 <sup>-1</sup>	0,102 863 087 081 06	0,986 157 493 021 34×10 <sup>-1</sup>
6	0,247 964 754 970 06×10 <sup>-3</sup>	0,253 580 406 026 54×10 <sup>-3</sup>	0,239 482 086 823 22×10 <sup>-3</sup>
7	0,277 437 604 228 70	0,323 252 002 339 82	0,303 300 048 569 50
8	-0,438 460 006 483 77×10 <sup>-1</sup>	-0,379 507 610 574 32×10 <sup>-1</sup>	-0,415 981 561 350 99×10 <sup>-1</sup>
9	-0,269 910 647 843 50	-0,325 348 020 144 52	-0,299 919 374 700 58
10	-0,693 134 130 898 60×10 <sup>-1</sup>	-0,790 509 690 510 11×10 <sup>-1</sup>	-0,803 693 427 641 09×10 <sup>-1</sup>
11	−0,296 321 459 816 53×10 <sup>-1</sup>	-0,206 367 205 477 75×10 <sup>-1</sup>	−0,297 613 732 511 51×10 <sup>-1</sup>
12	0,140 401 267 513 80×10 <sup>-1</sup>	0,570 538 093 347 50×10 <sup>-2</sup>	0,130 596 303 031 40×10 <sup>-1</sup>
	n-Pentane	Isopentane	n-Hexane
1	0,109 686 430 980 01×10 <sup>1</sup>	1,096 3	0,105 532 380 136 61×10 <sup>1</sup>
2	-0,299 888 882 980 61×10 <sup>1</sup>	-3,040 2	-0,261 206 158 906 29×10 <sup>1</sup>
3	0,995 168 867 992 12	1,031 7	0,766 138 829 672 60
4	-0,161 707 085 585 39	-0,154 10	-0,297 703 206 224 59
5	0,113 344 600 727 75	0,115 35	0,118 799 077 333 58
6	0,267 605 951 507 48×10 <sup>-3</sup>	0,000 298 09	0,279 228 610 626 17×10 <sup>-3</sup>
7	0,409 798 819 869 31	0,395 71	0,463 475 898 441 05
8	-0,408 764 230 830 75×10 <sup>-1</sup>	-0,045 881	0,114 331 969 802 97×10 <sup>-1</sup>
9	-0,381 694 824 694 47	-0,358 04	-0,482 569 687 381 31

b  $K_{\text{Pol},i} = 6$ ,  $K_{\text{Exp},i} = 18$ .

Table D.2 (continued)

k	$n_{0i,k}$	n <sub>oi,k</sub>	$n_{0i,k}$
10	-0,109 319 568 439 93	-0,101 07	-0,937 505 589 246 59×10 <sup>-1</sup>
11	-0,320 732 233 279 90×10 <sup>-1</sup>	-0,035 484	-0,672 732 471 559 94×10 <sup>-2</sup>
12	0,168 770 162 169 75×10 <sup>-1</sup>	0,018 156	-0,511 415 835 854 28×10 <sup>-2</sup>
	n-Heptane	n-Octane	n-Nonane
1	0,105 437 476 452 62×10 <sup>1</sup>	0,107 225 448 756 33×10 <sup>1</sup>	0,111 51×10 <sup>1</sup>
2	-0,265 006 815 061 44×10 <sup>1</sup>	-0,246 329 511 720 03×10 <sup>1</sup>	-0,270 20×10 <sup>1</sup>
3	0,817 300 478 275 43	0,653 866 740 549 28	0,834 16
4	-0,304 513 912 534 28	-0,363 249 740 856 28	-0,388 28
5	0,122 538 687 108 00	0,127 132 696 267 64	0,137 60
6	0,272 664 727 439 28×10 <sup>-3</sup>	0,307 135 727 779 30×10 <sup>-3</sup>	0,281 85×10 <sup>-3</sup>
7	0,498 658 256 816 70	0,526 568 569 875 40	0,620 37
8	-0,714 328 150 841 76×10 <sup>-3</sup>	0,193 628 628 576 53×10 <sup>-1</sup>	0,158 47×10 <sup>-1</sup>
9	-0,542 368 955 254 50	-0,589 394 268 491 55	-0,617 26
10	-0,138 018 216 107 56	-0,140 699 639 919 34	-0,150 43
11	-0,615 952 873 800 11×10 <sup>-2</sup>	-0,789 663 305 000 36×10 <sup>-2</sup>	-0,129 82×10 <sup>-1</sup>
12	0,486 025 103 930 22×10 <sup>-3</sup>	0,330 365 979 681 09×10 <sup>-2</sup>	0,443 25×10 <sup>-2</sup>
	n-Decane	Oxygen	Carbon monoxide
1	0,104 61×10 <sup>1</sup>	0,888 782 863 697 01	0,905 54
2	-0,248 07×10 <sup>1</sup>	-0,248 794 333 121 48×10 <sup>1</sup>	-2,451 5
3	0,743 72	0,597 501 907 758 86	0,531 49
4	-0,525 79	0,965 018 170 618 81×10 <sup>-2</sup>	0,024 173
5	0,153 15	0,719 704 287 127 70×10 <sup>-1</sup>	0,072 156
6	0,328 65×10 <sup>−3</sup>	0,223 374 430 001 95×10 <sup>-3</sup>	0,000 188 18
7	0,841 78	0,185 586 863 914 74	0,194 05
8	0,554 24×10 <sup>−1</sup>	-0,381 293 680 357 60×10 <sup>-1</sup>	-0,043 268
9	-0,735 55	-0,153 522 453 830 06	-0,127 78
10	-0,185 07	-0,267 268 149 109 19×10 <sup>-1</sup>	-0,027 896
11	−0,207 75×10 <sup>-1</sup>	-0,256 752 986 771 27×10 <sup>-1</sup>	-0,034 154
12	0,123 35×10 <sup>-1</sup>	0,957 143 021 236 68×10 <sup>-2</sup>	0,016 329
	Hydrogen sulfide	Argon	
1	0,876 41	0,850 957 148 039 69	
2	-0,203 67×10 <sup>1</sup>	-0,240 032 229 434 80×10 <sup>1</sup>	
3	0,216 34	0,541 278 414 764 66	
4	−0,501 99×10 <sup>-1</sup>	0,169 197 706 925 38×10 <sup>-1</sup>	
5	0,669 94×10 <sup>-1</sup>	0,688 259 650 190 35×10 <sup>-1</sup>	
6	0,190 76×10 <sup>-3</sup>	0,214 280 328 153 38×10 <sup>-3</sup>	
7	0,202 27	0,174 298 953 219 92	
8	-0,453 48×10 <sup>-2</sup>	-0,336 544 956 041 94×10 <sup>-1</sup>	
9	-0,222 30	-0,135 267 998 576 91	
10			
10	−0,347 14×10 <sup>−1</sup>	-0,163 873 507 915 52×10 <sup>-1</sup>	

**Table D.2** (continued)

k	$n_{0i,l}$	΄.		$n_{0i,k}$			$n_{0i,k}$	
12	0,741 54	×10 <sup>-2</sup>	0,887 69	92 048 157	09×10 <sup>-2</sup>			
k	$c_{0i,k}$	$d_{0i,k}$	$t_{0i,k}$		k	$C_{0i,k}$	$d_{0i,k}$	$t_{0i,k}$
1	-	1	0,25		7	1	2	0,625
2	-	1	1,125		8	1	5	1,75
3	-	1	1,5		9	2	1	3,625
4	-	2	1,375		10	2	4	3,625
5	-	3	0,25		11	3	3	14,5
6	-	7	0,875		12	3	4	12

<sup>&</sup>lt;sup>a</sup> The values of the coefficients and exponents are also valid for equation (9).

Table D.3 — Values of the coefficients and exponents in equation (D.11) for carbon dioxide, hydrogen, water, and heliuma

k	$C_{0i,k}$	$d_{0i,k}$	$t_{0i,k}$	$n_{0i,k}$
			Carbon dioxideb	
1	-	1	0	0,526 465 648 046 53
2	-	1	1,25	-0,149 957 250 425 92×10 <sup>1</sup>
3	-	2	1,625	0,273 297 867 337 82
4	_	3	0,375	0,129 495 000 227 86
5	1	3	0,375	0,154 040 883 418 41
6	1	3	1,375	-0,581 869 509 468 14
7	1	4	1,125	-0,180 224 948 382 96
8	1	5	1,375	-0,953 899 040 728 12×10 <sup>-1</sup>
9	1	6	0,125	-0,804 868 193 176 79×10 <sup>-2</sup>
10	1	6	1,625	-0,355 477 512 730 90×10 <sup>-1</sup>
11	2	1	3,75	-0,280 790 148 824 05
12	2	4	3,5	−0,824 358 900 816 77×10 <sup>-1</sup>
13	3	1	7,5	0,108 324 279 790 06×10 <sup>-1</sup>
14	3	1	8	-0,670 739 931 610 97×10 <sup>-2</sup>
15	3	3	6	-0,468 279 076 005 24×10 <sup>-2</sup>
16	3	3	16	-0,283 599 118 321 77×10 <sup>-1</sup>
17	3	4	11	0,195 001 747 440 98×10 <sup>-1</sup>
18	5	5	24	-0,216 091 375 071 66
19	5	5	26	0,437 727 949 269 72
20	5	5	28	-0,221 307 901 135 93
21	6	5	24	0,151 901 899 573 31×10 <sup>-1</sup>
22	6	5	26	-0,153 809 489 533 00×10 <sup>-1</sup>

b  $K_{\text{Pol},i} = 6$ ,  $K_{\text{Exp},i} = 6$ .

<sup>&</sup>lt;sup>c</sup> For the simultaneously optimized equations of state of Span and Wagner (2003), [10] the old molar gas constant  $R^*$  was substituted with the recent one R without conversion. This has nearly no effect on the quality of the equations of state (page 104 of [2]).

 Table D.3 (continued)

1.								
k	C <sub>Oi,k</sub>	$d_{0i,k}$	t <sub>oi,k</sub>	$n_{0i,k}$				
	Hydrogen <sup>c</sup>							
1	_	1	0,5	0,535 799 284 512 52×10¹				
2	_	1	0,625	-0,620 502 525 305 95×10 <sup>1</sup>				
3	_	2	0,375	0,138 302 413 270 86				
4	_	2	0,625	-0,713 979 548 961 29×10 <sup>-1</sup>				
5	_	4	1,125	0,154 740 539 597 33×10 <sup>-1</sup>				
6	1	1	2,625	-0,149 768 064 057 71				
7	1	5	0	-0,263 687 239 884 51×10 <sup>-1</sup>				
8	1	5	0,25	0,566 813 031 560 66×10 <sup>-1</sup>				
9	1	5	1,375	-0,600 639 580 304 36×10 <sup>-1</sup>				
10	2	1	4	-0,450 439 420 271 32				
11	2	1	4,25	0,424 788 402 445 00				
12	3	2	5	-0,219 976 408 271 39×10 <sup>-1</sup>				
13	3	5	8	-0,104 995 213 745 30×10 <sup>-1</sup>				
14	5	1	8	-0,289 559 028 668 16×10 <sup>-2</sup>				
			Waterd					
1	_	1	0,5	0,827 284 087 495 86				
2	_	1	1,25	-0,186 022 204 165 84×10 <sup>1</sup>				
3	_	1	1,875	-0,111 990 096 137 44×10 <sup>1</sup>				
4	_	2	0,125	0,156 357 539 760 56				
5	_	2	1,5	0,873 758 448 590 25				
6	_	3	1	-0,366 744 037 157 31				
7	_	4	0,75	0,539 878 934 324 36×10 <sup>-1</sup>				
8	1	1	1,5	0,109 576 902 144 99×10 <sup>1</sup>				
9	1	5	0,625	0,532 130 378 285 63×10 <sup>-1</sup>				
10	1	5	2,625	0,130 505 339 308 25×10 <sup>-1</sup>				
11	2	1	5	-0,410 795 204 344 76				
12	2	2	4	0,146 374 433 441 20				
13	2	4	4,5	-0,557 268 386 237 19×10 <sup>-1</sup>				
14	3	4	3	-0,112 017 741 438 00×10 <sup>-1</sup>				
15	5	1	4	-0,660 627 580 680 99×10 <sup>-2</sup>				
16	5	1	6	0,469 185 220 045 38×10 <sup>-2</sup>				
	1 5	<u> </u>		0,107 100 220 0 10 30 10				
			Helium <sup>e</sup>					
1	_	1	0	-0,455 790 240 067 37				
2	_	1	0,125	0,125 163 907 549 25×10 <sup>1</sup>				
3	_	1	0,75	-0,154 382 316 506 21×10 <sup>1</sup>				
4	_	4	1	0,204 674 897 072 21×10 <sup>-1</sup>				
5	1	1	0,75	-0,344 762 123 807 81				
6	1	3	2,625	-0,344 702 123 807 81 -0,208 584 595 127 87×10 <sup>-1</sup>				
	1	٥	2,023	-U,2U0 304 393 12/ 8/×1U <sup>-1</sup>				

k	c <sub>oi,k</sub>	$d_{0i,k}$	$t_{{ m o}i,k}$	$n_{0i,k}$
7	1	5	0,125	0,162 274 147 117 78×10 <sup>-1</sup>
8	1	5	1,25	-0,574 718 182 008 92×10 <sup>-1</sup>
9	1	5	2	0,194 624 164 307 15×10 <sup>-1</sup>
10	2	2	1	-0,332 956 801 230 20×10 <sup>-1</sup>
11	3	1	4,5	-0,108 635 773 723 67×10 <sup>-1</sup>
12	3	2	5	-0,221 733 652 459 54×10 <sup>-1</sup>

Table D.3 (continued)

- The values of the coefficients and exponents are also valid for equation (9).
- b  $K_{\text{Pol},i} = 4$ ,  $K_{\text{Exp},i} = 18$ .
- c  $K_{\text{Pol},i} = 5, K_{\text{Exp},i} = 9.$
- d  $K_{Pol,i} = 7, K_{Exp,i} = 9.$
- $K_{\text{Pol},i} = 4, K_{\text{Exp},i} = 8.$

## D.3 Calculation of the departure function contribution to the residual part of the reduced Helmholtz free energy

The purpose of the departure function is to further improve the accuracy of the mixture model in the description of thermodynamic properties in cases where fitting the parameters of the reducing functions for density and temperature (see Annex E) to accurate experimental data is not sufficient. The departure function  $\Delta \alpha^r$  of the multicomponent mixture is the double summation over all binary specific and generalized departure functions developed for the binary subsystems and is given by

$$\Delta \alpha^{\mathrm{r}}(\delta, \tau, \overline{x}) = \sum_{i=1}^{N-1} \sum_{j=i+1}^{N} \Delta \alpha_{ij}^{\mathrm{r}}(\delta, \tau, \overline{x})$$
(D.17)

with

$$\Delta \alpha_{ij}^{r}(\delta, \tau, \bar{x}) = x_i x_j F_{ij} \alpha_{ij}^{r}(\delta, \tau)$$
(D.18)

In this equation, the function  $\alpha_{ij}^{\,\rm r}(\delta,\tau)$  is the part of the departure function  $\Delta\alpha_{ij}^{\,\rm r}(\delta,\tau,\overline{x})$  that depends only on the reduced mixture variables  $\delta$  and  $\tau$ , as given by

$$\alpha_{ij}^{\mathrm{r}}(\delta,\tau) = \sum_{k=1}^{K_{\mathrm{Pol},ij}} n_{ij,k} \, \delta^{d_{ij,k}} \, \tau^{t_{ij,k}} + \sum_{k=K_{\mathrm{Pol},ij}+1}^{K_{\mathrm{Pol},ij}+K_{\mathrm{Exp},ij}} n_{ij,k} \, \delta^{d_{ij,k}} \, \tau^{t_{ij,k}} \exp\left(-\eta_{ij,k} \left(\delta - \varepsilon_{ij,k}\right)^2 - \beta_{ij,k} \left(\delta - \gamma_{ij,k}\right)\right)$$
(D.19)

where

 $\alpha_{ij}^{\,\rm r}(\delta,\tau)$  was developed either for a specific binary mixture (a binary specific departure function with binary specific coefficients, exponents, and parameters) or for a group of binary mixtures (generalized departure function with a uniform structure for the group of binary mixtures).

The values of the coefficients  $n_{ij,k}$ , the exponents  $d_{ij,k}$  and  $t_{ij,k}$ , and the parameters  $\eta_{ij,k}$ ,  $\varepsilon_{ij,k}$ ,  $\beta_{ij,k}$ , and  $\gamma_{ij,k}$  in equation (D.19) for all binary specific and generalized departure functions considered in the mixture model are given in Table D.4. The values of the non-zero  $F_{ij}$  parameters in equation (D.18) are listed in Table D.5.

## D.3.1 Binary specific departure functions

Binary specific departure functions were developed for the binary mixtures of methane with nitrogen, carbon dioxide, ethane, propane, and hydrogen, and of nitrogen with carbon dioxide and ethane. For a binary specific departure function, the adjustable factor  $F_{ij}$  in equation (D.18) equals unity.

## D.3.2 Generalized departure functions

A generalized departure function was developed for the binary mixtures of methane with either n-butane and isobutane, of ethane with propane, n-butane, and isobutane, of propane with n-butane and isobutane, and of n-butane with isobutane. For each mixture in the group of generalized binary mixtures, the parameter  $F_{ij}$  is fitted to the corresponding binary specific data (except for the binary system methane–n-butane, where  $F_{ij}$  equals unity).

### D.3.3 No departure functions

For all of the remaining binary mixtures, no departure function was developed, and  $F_{ij}$  equals zero, i.e.,  $\Delta \alpha_{ij}^{\mathbf{r}}(\delta, \tau, \overline{x})$  equals zero. For most of these mixtures, however, the parameters of the reducing functions for density and temperature are fitted to selected experimental data (see Annex E).

## D.3.4 Derivatives of $\alpha_{ii}^{\rm r}(\delta,\tau)$ with respect to the reduced mixture variables $\delta$ and $\tau$

The derivatives of the function  $\alpha_{ij}^{\rm r}(\delta,\tau)$  (equation D.19) with respect to the reduced mixture variables  $\delta$  and  $\tau$  are as follows:

$$\delta \left(\frac{\partial \alpha_{ij}^{r}}{\partial \delta}\right)_{\tau} = \sum_{k=1}^{K_{Pol,ij}} n_{ij,k} d_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} + \sum_{k=K_{Pol,ij}+1}^{K_{Pol,ij}+K_{Exp,ij}} n_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} \\
\exp \left(-\eta_{ij,k} \left(\delta - \varepsilon_{ij,k}\right)^{2} - \beta_{ij,k} \left(\delta - \gamma_{ij,k}\right)\right) \left[d_{ij,k} - 2\eta_{ij,k} \delta\left(\delta - \varepsilon_{ij,k}\right) - \delta\beta_{ij,k}\right] \\
\delta^{2} \left(\frac{\partial^{2} \alpha_{ij}^{r}}{\partial \delta^{2}}\right)_{\tau} = \sum_{k=1}^{K_{Pol,ij}} n_{ij,k} d_{ij,k} \left(d_{ij,k} - 1\right) \delta^{d_{ij,k}} \tau^{t_{ij,k}} + \sum_{k=K_{Pol,ij}+1}^{K_{Pol,ij}+K_{Exp,ij}} n_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} \\
\exp \left(-\eta_{ij,k} \left(\delta - \varepsilon_{ij,k}\right)^{2} - \beta_{ij,k} \left(\delta - \gamma_{ij,k}\right)\right) \left[\left(d_{ij,k} - 2\eta_{ij,k} \delta\left(\delta - \varepsilon_{ij,k}\right) - \delta\beta_{ij,k}\right)^{2} - d_{ij,k} - 2\eta_{ij,k} \delta^{2}\right]$$
(D.21)

$$\delta\tau \left(\frac{\partial^{2}\alpha_{ij}^{\mathbf{r}}}{\partial\delta\partial\tau}\right) = \sum_{k=1}^{K_{Pol,ij}} n_{ij,k} d_{ij,k} t_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} + \sum_{k=K_{Pol,ij}+1}^{K_{Pol,ij}+K_{Exp,ij}} n_{ij,k} t_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}}$$

$$\exp\left(-\eta_{ij,k} \left(\delta - \varepsilon_{ij,k}\right)^{2} - \beta_{ij,k} \left(\delta - \gamma_{ij,k}\right)\right) \left[d_{ij,k} - 2\eta_{ij,k} \delta\left(\delta - \varepsilon_{ij,k}\right) - \delta\beta_{ij,k}\right]$$
(D.22)

$$\tau \left( \frac{\partial \alpha_{ij}^{r}}{\partial \tau} \right)_{\delta} = \sum_{k=1}^{K_{Pol,ij}} n_{ij,k} t_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} \\
+ \sum_{k=K_{Pol,ij}+1}^{K_{Pol,ij}+K_{Exp,ij}} n_{ij,k} t_{ij,k} \delta^{d_{ij,k}} \tau^{t_{ij,k}} \exp \left( -\eta_{ij,k} \left( \delta - \varepsilon_{ij,k} \right)^{2} - \beta_{ij,k} \left( \delta - \gamma_{ij,k} \right) \right)$$
(D.23)

$$\tau^{2} \left( \frac{\partial^{2} \alpha_{ij}^{r}}{\partial \tau^{2}} \right)_{\delta} = \sum_{k=1}^{K_{Pol,ij}} n_{ij,k} t_{ij,k} \left( t_{ij,k} - 1 \right) \delta^{d_{ij,k}} \tau^{t_{ij,k}}$$

$$+ \sum_{k=K_{Pol,ij}+1}^{K_{Pol,ij}+K_{Exp,ij}} n_{ij,k} t_{ij,k} \left( t_{ij,k} - 1 \right) \delta^{d_{ij,k}} \tau^{t_{ij,k}} \exp \left( -\eta_{ij,k} \left( \delta - \varepsilon_{ij,k} \right)^{2} - \beta_{ij,k} \left( \delta - \gamma_{ij,k} \right) \right)$$
(D.24)

## D.3.5 Coefficients, exponents, and parameters for the departure functions

Table D.4 — Values of the coefficients, exponents, and parameters in equation (D.19) for the binary specific and generalized departure functionsa

k	$d_{ij,k}$	$t_{ij,k}$	$n_{ij,k}$	$\eta_{ij,k}$	$\varepsilon_{ij,k}$	$\beta_{ij,k}$	γij,k
			Methane-Nitrogen <sup>b</sup>				
1	1	0	-0,980 389 855 173 35×10 <sup>-2</sup>	_	_	_	-
2	4	1,85	0,424 872 701 430 05×10 <sup>-3</sup>	_	-	-	-
3	1	7,85	-0,348 002 145 761 42×10 <sup>-1</sup>	1	0,5	1	0,5
4	2	5,4	-0,133 338 130 138 96	1	0,5	1	0,5
5	2	0	-0,119 936 949 746 27×10 <sup>-1</sup>	0,25	0,5	2,5	0,5
6	2	0,75	0,692 433 797 751 68×10 <sup>-1</sup>	0	0,5	3	0,5
7	2	2,8	-0,310 225 081 482 49	0	0,5	3	0,5
8	2	4,45	0,244 954 917 532 26	0	0,5	3	0,5
9	3	4,25	0,223 698 167 169 81	0	0,5	3	0,5
			Methane-Carbon dioxide <sup>c</sup>				
1	1	2,6	-0,108 593 873 549 42	_	-	-	-
2	2	1,95	0,802 285 767 273 89×10 <sup>-1</sup>	_	-	-	-
3	3	0	-0,933 039 851 157 17×10 <sup>-2</sup>	_	-	-	-
4	1	3,95	0,409 892 740 058 48×10 <sup>-1</sup>	1	0,5	1	0,5
5	2	7,95	-0,243 380 197 724 94	0,5	0,5	2	0,5
6	3	8	0,238 553 472 811 24	0	0,5	3	0,5
			Methane-Ethane <sup>d</sup>				
1	3	0,65	-0,809 260 502 987 46×10 <sup>-3</sup>	_	-	-	-
2	4	1,55	-0,753 819 250 800 59×10 <sup>-3</sup>	_	-	-	-
3	1	3,1	-0,416 187 688 912 19×10 <sup>-1</sup>	1	0,5	1	0,5
4	2	5,9	-0,234 521 736 815 69	1	0,5	1	0,5
5	2	7,05	0,140 038 405 845 86	1	0,5	1	0,5
6	2	3,35	0,632 817 448 077 38×10 <sup>-1</sup>	0,875	0,5	1,25	0,5
7	2	1,2	-0,346 604 258 488 09×10 <sup>-1</sup>	0,75	0,5	1,5	0,5
8	2	5,8	-0,239 187 473 342 51	0,5	0,5	2	0,5
9	2	2,7	0,198 552 550 668 91×10 <sup>-2</sup>	0	0,5	3	0,5
10	3	0,45	0,617 777 461 715 55×10 <sup>1</sup>	0	0,5	3	0,5
11	3	0,55	-0,695 753 582 711 05×10 <sup>1</sup>	0	0,5	3	0,5

Table D.4 (continued)

1-				T	_	0	
k	$d_{ij,k}$	$t_{ij,k}$	n <sub>ij,k</sub>	$\eta_{ij,k}$	ε <sub>ij,k</sub>	$\beta_{ij,k}$	γij,k
12	3	1,95	0,106 301 853 063 88×10 <sup>1</sup>	0	0,5	3	0,5
		4.05	Methane-Propane <sup>e</sup>				
1	3	1,85	0,137 464 299 585 76×10 <sup>-1</sup>	-	-	-	_
2	3	3,95	-0,744 250 121 295 52×10 <sup>-2</sup>	-	_	-	_
3	4	0	-0,455 166 002 136 85×10 <sup>-2</sup>	-	_	-	-
4	4	1,85	-0,545 466 033 502 37×10 <sup>-2</sup>	-	_	-	-
5	4	3,85	0,236 820 168 244 71×10 <sup>-2</sup>	-	-	-	_
6	1	5,25	0,180 077 637 214 38	0,25	0,5	0,75	0,5
7	1	3,85	-0,447 739 429 324 86	0,25	0,5	1	0,5
8	1	0,2	0,193 273 748 882 00×10 <sup>-1</sup>	0	0,5	2	0,5
9	2	6,5	-0,306 321 978 046 24	0	0,5	3	0,5
		T	Nitrogen-Carbon dioxidef		Г	1	ı
1	2	1,85	0,286 616 250 283 99	_	-	-	-
2	3	1,4	-0,109 198 338 612 47	-	-	-	-
3	1	3,2	-0,113 740 320 822 70×10 <sup>1</sup>	0,25	0,5	0,75	0,5
4	1	2,5	0,765 805 442 373 58	0,25	0,5	1	0,5
5	1	8	0,426 380 009 268 19×10 <sup>-2</sup>	0	0,5	2	0,5
6	2	3,75	0,176 735 382 045 34	0	0,5	3	0,5
			Nitrogen-Ethane <sup>g</sup>				
1	2	0	-0,473 765 181 266 08	-	-	-	-
2	2	0,05	0,489 611 934 610 01	-	-	-	-
3	3	0	−0,570 110 620 905 35×10 <sup>-2</sup>	_	-	_	_
4	1	3,65	-0,199 668 200 413 20	1	0,5	1	0,5
5	2	4,9	-0,694 111 031 017 23	1	0,5	1	0,5
6	2	4,45	0,692 261 927 390 21	0,875	0,5	1,25	0,5
			$\textbf{Methane-Hydrogen}^h$				
1	1	2	-0,251 571 349 719 34	-	-	_	-
2	3	-1	-0,622 038 411 119 83×10 <sup>-2</sup>	-	-	-	-
3	3	1,75	0,888 503 151 843 96×10 <sup>-1</sup>	-	-	-	-
4	4	1,4	−0,355 922 125 732 39×10 <sup>-1</sup>	-	-	_	-
			ne, Methane-Isobutane, Ethane-Pr				
	Ethan	1	ropane-n-Butane, Propane-Isobuta	ine, and n-Bi	utane-Isob	outane <sup>i</sup>	I
1	1	1	0,255 747 768 441 18×10 <sup>1</sup>	_	_	-	_
2	1	1,55	-0,798 463 571 363 53×10 <sup>1</sup>	-	-	-	_
3	1	1,7	0,478 591 314 658 06×10 <sup>1</sup>	_	_	-	_
4	2	0,25	-0,732 653 923 695 87	_	_	-	_
5	2	1,35	0,138 054 713 453 12×10 <sup>1</sup>	_	_	-	_
6	3	0	0,283 496 034 763 65	-	_	-	_
7	3	1,25	-0,490 873 859 404 25	-	-	-	-

Table D.4 (continued)

k	$d_{ij,k}$	$t_{ij,k}$	$n_{ij,k}$	$\eta_{ij,k}$	$\varepsilon_{ij,k}$	$\beta_{ij,k}$	γij,k
8	4	0	-0,102 918 889 214 47	-	-	-	_
9	4	0,7	0,118 363 146 819 68	-	_	_	_
10	4	5,4	0,555 273 857 219 43×10 <sup>-4</sup>	-	-	-	-

- a The values of the coefficients, exponents, and parameters are also valid for equation (12).
- b  $K_{\text{Pol},ij} = 2, K_{\text{Exp},ij} = 7.$
- c  $K_{\text{Pol},ij} = 3, K_{\text{Exp},ij} = 3.$
- d  $K_{\text{Pol},ij} = 2$ ,  $K_{\text{Exp},ij} = 10$ .
- e  $K_{\text{Pol},ij} = 5, K_{\text{Exp},ij} = 4.$
- f  $K_{\text{Pol},ij} = 2, K_{\text{Exp},ij} = 4.$
- g  $K_{\text{Pol},ij} = 3, K_{\text{Exp},ij} = 3.$
- h  $K_{\text{Pol},ij} = 4$ ,  $K_{\text{Exp},ij} = 0$ .
- $K_{\text{Pol},ij} = 10, K_{\text{Exp},ij} = 0.$

Table D.5 — Values of the non-zero  $F_{ij}$  parameters in equation (D.18) for the binary specific and generalized departure functionsa,b

Mixture <i>i–j</i>	$F_{ij}$
Methane-Nitrogen	1
Methane-Carbon dioxide	1
Methane-Ethane	1
Methane-Propane	1
Methane-n-Butane	1
Methane-Isobutane	0,771 035 405 688
Methane-Hydrogen	1
Nitrogen–Carbon dioxide	1
Nitrogen-Ethane	1
Ethane–Propane	0,130 424 765 150
Ethane-n-Butane	0,281 570 073 085
Ethane-Isobutane	0,260 632 376 098
Propane-n-Butane	0,312 572 600 489×10 <sup>-1</sup>
Propane-Isobutane	-0,551 609 771 024×10 <sup>-1</sup>
n-Butane–Isobutane	-0,551 240 293 009×10 <sup>-1</sup>
The values of the $F_{ij}$ parameters are also	valid for equation (11).

The  $F_{ij}$  parameters equal zero for all other binary combinations.

## **Annex E**

(normative)

## The reducing functions for density and temperature

## E.1 Calculation of the reducing functions for density and temperature

The composition-dependent reducing functions are used to calculate the reduced mixture variables  $\delta$  and  $\tau$  (i.e., dimensionless mixture density and temperature, respectively) according to

$$\delta = \frac{\rho}{\rho_{\rm r}(\bar{x})} \tag{E.1}$$

and

$$\tau = \frac{T_{\rm r}(\overline{x})}{T} \tag{E.2}$$

The reducing functions for density and temperature can be written as

$$\frac{1}{\rho_{\rm r}(\bar{x})} = \sum_{i=1}^{N} \sum_{j=1}^{N} x_i x_j \, \beta_{\nu,ij} \, \gamma_{\nu,ij} \, \frac{x_i + x_j}{\beta_{\nu,ij}^2 \, x_i + x_j} \, \frac{1}{8} \left( \frac{1}{\rho_{\rm c,i}^{1/3}} + \frac{1}{\rho_{\rm c,j}^{1/3}} \right)^3 \tag{E.3}$$

and

$$T_{\rm r}(\overline{x}) = \sum_{i=1}^{N} \sum_{j=1}^{N} x_i x_j \, \beta_{T,ij} \, \gamma_{T,ij} \, \frac{x_i + x_j}{\beta_{T,ij}^2 \, x_i + x_j} \left( T_{\rm c,i} \cdot T_{\rm c,j} \right)^{0.5} \tag{E.4}$$

with

$$\beta_{v,ij} = 1/\beta_{v,ji}$$
,  $\gamma_{v,ij} = \gamma_{v,ji}$ ,  $\beta_{T,ij} = 1/\beta_{T,ji}$ , and  $\gamma_{T,ij} = \gamma_{T,ji}$ 

These functions are based on quadratic mixing rules and are thus reasonably connected to physically well-founded mixing rules. The binary parameters of equations (E.3) and (E.4) consider the deviation between the behaviour of the real mixture and the one resulting from the ideal combining rules (with  $\beta$  and  $\gamma$  set to 1) for the critical parameters of the pure components.

The values of the binary parameters  $\beta_{v,ij}$ ,  $\gamma_{v,ij}$ ,  $\beta_{T,ij}$ , and  $\gamma_{T,ij}$  in equations (E.3) and (E.4) for all binary mixtures are listed in <u>Table E.1</u>. The critical parameters  $\rho_{c,i}$  and  $T_{c,i}$  of the pure components are given in <u>Annex C</u>.

### E.1.1 Binary parameters for mixtures with no or very poor experimental data

The binary parameters  $\beta_{\nu,ij}$  and  $\gamma_{\nu,ij}$  in equation (E.3) and  $\beta_{T,ij}$  and  $\gamma_{T,ij}$  in equation (E.4) are fitted to experimental data for binary mixtures. In those cases where sufficient experimental data are not available, the parameters of equations (E.3) and (E.4) are either set to unity or modified (calculated) in such a manner that the critical parameters of the pure components are combined in a different way that proved to be more suitable for certain binary subsystems. For example, for binary hydrocarbon

mixtures for which no data or only few or very poor data are available,  $\beta_{v,ij}$  and  $\beta_{T,ij}$  are set to one, and the binary parameters  $\gamma_{v,ij}$  and  $\gamma_{T,ij}$  are calculated from the following conversions:

$$\gamma_{v,ij} = 4 \frac{\left(\frac{1}{\rho_{c,i}} + \frac{1}{\rho_{c,j}}\right)}{\left(\frac{1}{\rho_{c,i}^{1/3}} + \frac{1}{\rho_{c,j}^{1/3}}\right)^3} \text{ and } \gamma_{T,ij} = \frac{1}{2} \frac{\left(T_{c,i} + T_{c,j}\right)}{\left(T_{c,i} \cdot T_{c,j}\right)^{0.5}}$$
(E.5)

In this way, the original Lorentz and Berthelot combining rules for the critical parameters of the pure components in equations (E.3) and (E.4) are substituted by the arithmetic mean of the critical parameters. The use of different combining rules is superfluous when data are used to adjust the binary parameters.

Table E.1 — Values of the binary parameters in equations (E.3) and (E.4) for the reducing functions for density and temperature

Mixture <i>i–j</i>	$eta_{ u,ij}$	γ <sub>ν,ij</sub>	$eta_{T,ij}$	ΥT,ij
CH <sub>4</sub> -N <sub>2</sub>	0,998 721 377	1,013 950 311	0,998 098 830	0,979 273 013
CH <sub>4</sub> -CO <sub>2</sub>	0,999 518 072	1,002 806 594	1,022 624 490	0,975 665 369
CH <sub>4</sub> -C <sub>2</sub> H <sub>6</sub>	0,997 547 866	1,006 617 867	0,996 336 508	1,049 707 697
CH <sub>4</sub> -C <sub>3</sub> H <sub>8</sub>	1,004 827 070	1,038 470 657	0,989 680 305	1,098 655 531
CH <sub>4</sub> -n-C <sub>4</sub> H <sub>10</sub>	0,979 105 972	1,045 375 122	0,994 174 910	1,171 607 691
CH <sub>4</sub> -i-C <sub>4</sub> H <sub>10</sub>	1,011 240 388	1,054 319 053	0,980 315 756	1,161 117 729
CH <sub>4</sub> -n-C <sub>5</sub> H <sub>12</sub>	0,948 330 120	1,124 508 039	0,992 127 525	1,249 173 968
CH <sub>4</sub> -i-C <sub>5</sub> H <sub>12</sub>	1,0	1,343 685 343	1,0	1,188 899 743
CH <sub>4</sub> -n-C <sub>6</sub> H <sub>14</sub>	0,958 015 294	1,052 643 846	0,981 844 797	1,330 570 181
CH <sub>4</sub> -n-C <sub>7</sub> H <sub>16</sub>	0,962 050 831	1,156 655 935	0,977 431 529	1,379 850 328
CH <sub>4</sub> -n-C <sub>8</sub> H <sub>18</sub>	0,994 740 603	1,116 549 372	0,957 473 785	1,449 245 409
CH <sub>4</sub> -n-C <sub>9</sub> H <sub>20</sub>	1,002 852 287	1,141 895 355	0,947 716 769	1,528 532 478
CH <sub>4</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,033 086 292	1,146 089 637	0,937 777 823	1,568 231 489
CH <sub>4</sub> -H <sub>2</sub>	1,0	1,018 702 573	1,0	1,352 643 115
CH <sub>4</sub> -O <sub>2</sub>	1,0	1,0	1,0	0,950 000 000
CH <sub>4</sub> -CO	0,997 340 772	1,006 102 927	0,987 411 732	0,987 473 033
CH <sub>4</sub> -H <sub>2</sub> O	1,012 783 169	1,585 018 334	1,063 333 913	0,775 810 513
CH <sub>4</sub> -H <sub>2</sub> S	1,012 599 087	1,040 161 207	1,011 090 031	0,961 155 729
CH <sub>4</sub> -He	1,0	0,881 405 683	1,0	3,159 776 855
CH <sub>4</sub> -Ar	1,034 630 259	1,014 678 542	0,990 954 281	0,989 843 388
N <sub>2</sub> -CO <sub>2</sub>	0,977 794 634	1,047 578 256	1,005 894 529	1,107 654 104
N <sub>2</sub> -C <sub>2</sub> H <sub>6</sub>	0,978 880 168	1,042 352 891	1,007 671 428	1,098 650 964
N <sub>2</sub> -C <sub>3</sub> H <sub>8</sub>	0,974 424 681	1,081 025 408	1,002 677 329	1,201 264 026
N <sub>2</sub> -n-C <sub>4</sub> H <sub>10</sub>	0,996 082 610	1,146 949 309	0,994 515 234	1,304 886 838
N <sub>2</sub> -i-C <sub>4</sub> H <sub>10</sub>	0,986 415 830	1,100 576 129	0,992 868 130	1,284 462 634
N <sub>2</sub> -n-C <sub>5</sub> H <sub>12</sub>	1,0	1,078 877 166	1,0	1,419 029 041
N <sub>2</sub> -i-C <sub>5</sub> H <sub>12</sub>	1,0	1,154 135 439	1,0	1,381 770 770
N <sub>2</sub> -n-C <sub>6</sub> H <sub>14</sub>	1,0	1,195 952 177	1,0	1,472 607 971
N <sub>2</sub> -n-C <sub>7</sub> H <sub>16</sub>	1,0	1,404 554 090	1,0	1,520 975 334
N <sub>2</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,0	1,186 067 025	1,0	1,733 280 051

 Table E.1 (continued)

Mixture <i>i–j</i>	$eta_{v,ij}$	$\gamma_{v,ij}$	$\beta_{T,ij}$	ΥΤ,ij
N <sub>2</sub> -n-C <sub>9</sub> H <sub>20</sub>	1,0	1,100 405 929	0,956 379 450	1,749 119 996
N <sub>2</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,0	1,0	0,957 934 447	1,822 157 123
N <sub>2</sub> -H <sub>2</sub>	0,972 532 065	0,970 115 357	0,946 134 337	1,175 696 583
N <sub>2</sub> -O <sub>2</sub>	0,999 521 770	0,997 082 328	0,997 190 589	0,995 157 044
N <sub>2</sub> -CO	1,0	1,008 690 943	1,0	0,993 425 388
N <sub>2</sub> -H <sub>2</sub> O	1,0	1,094 749 685	1,0	0,968 808 467
N <sub>2</sub> -H <sub>2</sub> S	0,910 394 249	1,256 844 157	1,004 692 366	0,960 174 200
N <sub>2</sub> -He	0,969 501 055	0,932 629 867	0,692 868 765	1,471 831 580
N <sub>2</sub> -Ar	1,004 166 412	1,002 212 182	0,999 069 843	0,990 034 831
CO <sub>2</sub> -C <sub>2</sub> H <sub>6</sub>	1,002 525 718	1,032 876 701	1,013 871 147	0,900 949 530
CO <sub>2</sub> -C <sub>3</sub> H <sub>8</sub>	0,996 898 004	1,047 596 298	1,033 620 538	0,908 772 477
CO <sub>2</sub> -n-C <sub>4</sub> H <sub>10</sub>	1,174 760 923	1,222 437 324	1,018 171 004	0,911 498 231
CO <sub>2</sub> -i-C <sub>4</sub> H <sub>10</sub>	1,076 551 882	1,081 909 003	1,023 339 824	0,929 982 936
CO <sub>2</sub> -n-C <sub>5</sub> H <sub>12</sub>	1,024 311 498	1,068 406 078	1,027 000 795	0,979 217 302
$CO_2$ -i- $C_5H_{12}$	1,060 793 104	1,116 793 198	1,019 180 957	0,961 218 039
CO <sub>2</sub> -n-C <sub>6</sub> H <sub>14</sub>	1,0	0,851 343 711	1,0	1,038 675 574
CO <sub>2</sub> -n-C <sub>7</sub> H <sub>16</sub>	1,205 469 976	1,164 585 914	1,011 806 317	1,046 169 823
CO <sub>2</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,026 169 373	1,104 043 935	1,029 690 780	1,074 455 386
CO <sub>2</sub> -n-C <sub>9</sub> H <sub>20</sub>	1,0	0,973 386 152	1,007 688 620	1,140 671 202
CO <sub>2</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,000 151 132	1,183 394 668	1,020 028 790	1,145 512 213
CO <sub>2</sub> -H <sub>2</sub>	0,904 142 159	1,152 792 550	0,942 320 195	1,782 924 792
CO <sub>2</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
CO <sub>2</sub> -CO	1,0	1,0	1,0	1,0
CO <sub>2</sub> -H <sub>2</sub> O	0,949 055 959	1,542 328 793	0,997 372 205	0,775 453 996
CO <sub>2</sub> -H <sub>2</sub> S	0,906 630 564	1,024 085 837	1,016 034 583	0,926 018 880
CO <sub>2</sub> -He	0,846 647 561	0,864 141 549	0,768 377 630	3,207 456 948
CO <sub>2</sub> -Ar	1,008 392 428	1,029 205 465	0,996 512 863	1,050 971 635
C <sub>2</sub> H <sub>6</sub> -C <sub>3</sub> H <sub>8</sub>	0,997 607 277	1,003 034 720	0,996 199 694	1,014 730 190
C <sub>2</sub> H <sub>6</sub> -n-C <sub>4</sub> H <sub>10</sub>	0,999 157 205	1,006 179 146	0,999 130 554	1,034 832 749
C <sub>2</sub> H <sub>6</sub> -i-C <sub>4</sub> H <sub>10</sub>	1,0	1,006 616 886	1,0	1,033 283 811
C <sub>2</sub> H <sub>6</sub> -n-C <sub>5</sub> H <sub>12</sub>	0,993 851 009	1,026 085 655	0,998 688 946	1,066 665 676
$C_2H_6-i-C_5H_{12}a$	1,0	1,045 439 935	1,0	1,021 150 247
C <sub>2</sub> H <sub>6</sub> -n-C <sub>6</sub> H <sub>14</sub>	1,0	1,169 701 102	1,0	1,092 177 796
$C_2H_6-n-C_7H_{16}$	1,0	1,057 666 085	1,0	1,134 532 014
C <sub>2</sub> H <sub>6</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,007 469 726	1,071 917 985	0,984 068 272	1,168 636 194
C <sub>2</sub> H <sub>6</sub> -n-C <sub>9</sub> H <sub>20</sub> <sup>a</sup>	1,0	1,143 534 730	1,0	1,056 033 030
C <sub>2</sub> H <sub>6</sub> -n-C <sub>10</sub> H <sub>22</sub>	0,995 676 258	1,098 361 281	0,970 918 061	1,237 191 558
C <sub>2</sub> H <sub>6</sub> -H <sub>2</sub>	0,925 367 171	1,106 072 040	0,932 969 831	1,902 008 495
$C_2H_6-O_2$	1,0	1,0	1,0	1,0
C <sub>2</sub> H <sub>6</sub> -CO	1,0	1,201 417 898	1,0	1,069 224 728
C <sub>2</sub> H <sub>6</sub> -H <sub>2</sub> O	1,0	1,0	1,0	1,0

 Table E.1 (continued)

C <sub>2</sub> H <sub>6</sub> -H <sub>2</sub> S				ΥT,ij
-202-	1,010 817 909	1,030 988 277	0,990 197 354	0,902 736 660
C <sub>2</sub> H <sub>6</sub> -He	1,0	1,0	1,0	1,0
C <sub>2</sub> H <sub>6</sub> -Ar	1,0	1,0	1,0	1,0
C <sub>3</sub> H <sub>8</sub> -n-C <sub>4</sub> H <sub>10</sub>	0,999 795 868	1,003 264 179	1,000 310 289	1,007 392 782
C <sub>3</sub> H <sub>8</sub> -i-C <sub>4</sub> H <sub>10</sub>	0,999 243 146	1,001 156 119	0,998 012 298	1,005 250 774
C <sub>3</sub> H <sub>8</sub> -n-C <sub>5</sub> H <sub>12</sub>	1,044 919 431	1,019 921 513	0,996 484 021	1,008 344 412
C <sub>3</sub> H <sub>8</sub> -i-C <sub>5</sub> H <sub>12</sub>	1,040 459 289	0,999 432 118	0,994 364 425	1,003 269 500
C <sub>3</sub> H <sub>8</sub> -n-C <sub>6</sub> H <sub>14</sub>	1,0	1,057 872 566	1,0	1,025 657 518
C <sub>3</sub> H <sub>8</sub> -n-C <sub>7</sub> H <sub>16</sub>	1,0	1,079 648 053	1,0	1,050 044 169
C <sub>3</sub> H <sub>8</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,0	1,102 764 612	1,0	1,063 694 129
C <sub>3</sub> H <sub>8</sub> -n-C <sub>9</sub> H <sub>20</sub>	1,0	1,199 769 134	1,0	1,109 973 833
C <sub>3</sub> H <sub>8</sub> -n-C <sub>10</sub> H <sub>22</sub>	0,984 104 227	1,053 040 574	0,985 331 233	1,140 905 252
C <sub>3</sub> H <sub>8</sub> -H <sub>2</sub>	1,0	1,074 006 110	1,0	2,308 215 191
C <sub>3</sub> H <sub>8</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
C <sub>3</sub> H <sub>8</sub> -CO	1,0	1,108 143 673	1,0	1,197 564 208
C <sub>3</sub> H <sub>8</sub> -H <sub>2</sub> O	1,0	1,011 759 763	1,0	0,600 340 961
C <sub>3</sub> H <sub>8</sub> -H <sub>2</sub> S	0,936 811 219	1,010 593 999	0,992 573 556	0,905 829 247
C <sub>3</sub> H <sub>8</sub> -He	1,0	1,0	1,0	1,0
C <sub>3</sub> H <sub>8</sub> -Ar	1,0	1,0	1,0	1,0
n-C <sub>4</sub> H <sub>10</sub> -i-C <sub>4</sub> H <sub>10</sub>	1,000 880 464	1,000 414 440	1,000 077 547	1,001 432 824
n-C <sub>4</sub> H <sub>10</sub> -n-C <sub>5</sub> H <sub>12</sub>	1,0	1,018 159 650	1,0	1,002 143 640
n-C <sub>4</sub> H <sub>10</sub> -i-C <sub>5</sub> H <sub>12</sub> <sup>a</sup>	1,0	1,002 728 434	1,0	1,000 792 201
n-C <sub>4</sub> H <sub>10</sub> -n-C <sub>6</sub> H <sub>14</sub>	1,0	1,034 995 284	1,0	1,009 157 060
n-C <sub>4</sub> H <sub>10</sub> -n-C <sub>7</sub> H <sub>16</sub>	1,0	1,019 174 227	1,0	1,021 283 378
n-C <sub>4</sub> H <sub>10</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,0	1,046 905 515	1,0	1,033 180 106
n-C <sub>4</sub> H <sub>10</sub> -n-C <sub>9</sub> H <sub>20</sub> <sup>a</sup>	1,0	1,049 219 137	1,0	1,014 096 448
n-C <sub>4</sub> H <sub>10</sub> -n-C <sub>10</sub> H <sub>22</sub>	0,976 951 968	1,027 845 529	0,993 688 386	1,076 466 918
n-C <sub>4</sub> H <sub>10</sub> -H <sub>2</sub>	1,0	1,232 939 523	1,0	2,509 259 945
n-C <sub>4</sub> H <sub>10</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
n-C <sub>4</sub> H <sub>10</sub> -CO <sup>a</sup>	1,0	1,084 740 904	1,0	1,173 916 162
n-C <sub>4</sub> H <sub>10</sub> -H <sub>2</sub> O	1,0	1,223 638 763	1,0	0,615 512 682
$n-C_4H_{10}-H_2S$	0,908 113 163	1,033 366 041	0,985 962 886	0,926 156 602
n-C <sub>4</sub> H <sub>10</sub> -He	1,0	1,0	1,0	1,0
n-C <sub>4</sub> H <sub>10</sub> -Ar	1,0	1,214 638 734	1,0	1,245 039 498
i-C <sub>4</sub> H <sub>10</sub> -n-C <sub>5</sub> H <sub>12</sub> <sup>a</sup>	1,0	1,002 779 804	1,0	1,002 495 889
$i-C_4H_{10}-i-C_5H_{12}^a$	1,0	1,002 284 353	1,0	1,001 835 788
i-C <sub>4</sub> H <sub>10</sub> -n-C <sub>6</sub> H <sub>14</sub> <sup>a</sup>	1,0	1,010 493 989	1,0	1,006 018 054
$i-C_4H_{10}-n-C_7H_{16}^a$	1,0	1,021 668 316	1,0	1,009 885 760
$i-C_4H_{10}-n-C_8H_{18}^a$	1,0	1,032 807 063	1,0	1,013 945 424
i-C <sub>4</sub> H <sub>10</sub> -n-C <sub>9</sub> H <sub>20</sub> <sup>a</sup>	1,0	1,047 298 475	1,0	1,017 817 492

 Table E.1 (continued)

Mixture <i>i–j</i>	$eta_{v,ij}$	$\gamma_{v,ij}$	$\beta_{T,ij}$	ΥΤ,ij
i-C <sub>4</sub> H <sub>10</sub> -n-C <sub>10</sub> H <sub>22</sub> <sup>a</sup>	1,0	1,060 243 344	1,0	1,021 624 748
i-C <sub>4</sub> H <sub>10</sub> -H <sub>2</sub> <sup>a</sup>	1,0	1,147 595 688	1,0	1,895 305 393
i-C <sub>4</sub> H <sub>10</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
i-C <sub>4</sub> H <sub>10</sub> -CO <sup>a</sup>	1,0	1,087 272 232	1,0	1,161 390 082
i-C <sub>4</sub> H <sub>10</sub> -H <sub>2</sub> O	1,0	1,0	1,0	1,0
i-C <sub>4</sub> H <sub>10</sub> -H <sub>2</sub> S	1,012 994 431	0,988 591 117	0,974 550 548	0,937 130 844
i-C <sub>4</sub> H <sub>10</sub> –He	1,0	1,0	1,0	1,0
i-C <sub>4</sub> H <sub>10</sub> -Ar	1,0	1,0	1,0	1,0
n-C <sub>5</sub> H <sub>12</sub> -i-C <sub>5</sub> H <sub>12</sub> <sup>a</sup>	1,0	1,000 024 335	1,0	1,000 050 537
n-C <sub>5</sub> H <sub>12</sub> -n-C <sub>6</sub> H <sub>14</sub> a	1,0	1,002 480 637	1,0	1,000 761 237
n-C <sub>5</sub> H <sub>12</sub> -n-C <sub>7</sub> H <sub>16</sub> a	1,0	1,008 972 412	1,0	1,002 441 051
n-C <sub>5</sub> H <sub>12</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,0	1,069 223 964	1,0	1,016 422 347
n-C <sub>5</sub> H <sub>12</sub> -n-C <sub>9</sub> H <sub>20</sub>	1,0	1,034 910 633	1,0	1,103 421 755
n-C <sub>5</sub> H <sub>12</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,0	1,016 370 338	1,0	1,049 035 838
n-C <sub>5</sub> H <sub>12</sub> -H <sub>2</sub> a	1,0	1,188 334 783	1,0	2,013 859 174
n-C <sub>5</sub> H <sub>12</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
n-C <sub>5</sub> H <sub>12</sub> -CO <sup>a</sup>	1,0	1,119 954 454	1,0	1,206 043 295
n-C <sub>5</sub> H <sub>12</sub> -H <sub>2</sub> O	1,0	0,956 677 310	1,0	0,447 666 011
n-C <sub>5</sub> H <sub>12</sub> -H <sub>2</sub> S	0,984 613 203	1,076 539 234	0,962 006 651	0,959 065 662
n-C <sub>5</sub> H <sub>12</sub> –He	1,0	1,0	1,0	1,0
n-C <sub>5</sub> H <sub>12</sub> -Ar	1,0	1,0	1,0	1,0
U 12		,	·	
i-C <sub>5</sub> H <sub>12</sub> -n-C <sub>6</sub> H <sub>14</sub> <sup>a</sup>	1,0	1,002 995 876	1,0	1,001 204 174
i-C <sub>5</sub> H <sub>12</sub> -n-C <sub>7</sub> H <sub>16</sub> <sup>a</sup>	1,0	1,009 928 206	1,0	1,003 194 615
i-C <sub>5</sub> H <sub>12</sub> -n-C <sub>8</sub> H <sub>18</sub> <sup>a</sup>	1,0	1,017 880 545	1,0	1,005 647 480
i-C <sub>5</sub> H <sub>12</sub> -n-C <sub>9</sub> H <sub>20</sub> <sup>a</sup>	1,0	1,028 994 325	1,0	1,008 191 499
i-C <sub>5</sub> H <sub>12</sub> -n-C <sub>10</sub> H <sub>22</sub> <sup>a</sup>	1,0	1,039 372 957	1,0	1,010 825 138
i-C <sub>5</sub> H <sub>12</sub> -H <sub>2</sub> a	1,0	1,184 340 443	1,0	1,996 386 669
i-C <sub>5</sub> H <sub>12</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
i-C <sub>5</sub> H <sub>12</sub> -CO <sup>a</sup>	1,0	1,116 694 577	1,0	1,199 326 059
i-C <sub>5</sub> H <sub>12</sub> -H <sub>2</sub> O	1,0	1,0	1,0	1,0
i-C <sub>5</sub> H <sub>12</sub> -H <sub>2</sub> S	1,0	0,835 763 343	1,0	0,982 651 529
i-C <sub>5</sub> H <sub>12</sub> –He	1,0	1,0	1,0	1,0
i-C <sub>5</sub> H <sub>12</sub> -Ar	1,0	1,0	1,0	1,0
				·
n-C <sub>6</sub> H <sub>14</sub> -n-C <sub>7</sub> H <sub>16</sub>	1,0	1,001 508 227	1,0	0,999 762 786
n-C <sub>6</sub> H <sub>14</sub> -n-C <sub>8</sub> H <sub>18</sub> <sup>a</sup>	1,0	1,006 268 954	1,0	1,001 633 952
n-C <sub>6</sub> H <sub>14</sub> -n-C <sub>9</sub> H <sub>20</sub>	1,0	1,020 761 680	1,0	1,055 369 591
n-C <sub>6</sub> H <sub>14</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,001 516 371	1,013 511 439	0,997 641 010	1,028 939 539
n-C <sub>6</sub> H <sub>14</sub> -H <sub>2</sub>	1,0	1,243 461 678	1,0	3,021 197 546
n-C <sub>6</sub> H <sub>14</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
n-C <sub>6</sub> H <sub>14</sub> -CO <sup>a</sup>	1,0	1,155 145 836	1,0	1,233 272 781

 Table E.1 (continued)

Mixture <i>i–j</i>	$eta_{ u,ij}$	γ <sub>v,ij</sub>	$\beta_{T,ij}$	γ <sub>T,ij</sub>
n-C <sub>6</sub> H <sub>14</sub> -H <sub>2</sub> O	1,0	1,170 217 596	1,0	0,569 681 333
n-C <sub>6</sub> H <sub>14</sub> -H <sub>2</sub> S	0,754 473 958	1,339 283 552	0,985 891 113	0,956 075 596
n-C <sub>6</sub> H <sub>14</sub> -He	1,0	1,0	1,0	1,0
n-C <sub>6</sub> H <sub>14</sub> -Ar	1,0	1,0	1,0	1,0
n-C <sub>7</sub> H <sub>16</sub> -n-C <sub>8</sub> H <sub>18</sub>	1,0	1,006 767 176	1,0	0,998 793 111
n-C7H <sub>16</sub> -n-C9H <sub>20</sub>	1,0	1,001 370 076	1,0	1,001 150 096
n-C <sub>7</sub> H <sub>16</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,0	1,002 972 346	1,0	1,002 229 938
n-C <sub>7</sub> H <sub>16</sub> -H <sub>2</sub>	1,0	1,159 131 722	1,0	3,169 143 057
n-C <sub>7</sub> H <sub>16</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
n-C <sub>7</sub> H <sub>16</sub> –CO <sup>a</sup>	1,0	1,190 354 273	1,0	1,256 123 503
n-C <sub>7</sub> H <sub>16</sub> -H <sub>2</sub> O	1,0	1,0	1,0	1,0
n-C <sub>7</sub> H <sub>16</sub> -H <sub>2</sub> S	0,828 967 164	1,087 956 749	0,988 937 417	1,013 453 092
n-C <sub>7</sub> H <sub>16</sub> –He	1,0	1,0	1,0	1,0
n-C <sub>7</sub> H <sub>16</sub> –Ar	1,0	1,0	1,0	1,0
n-C <sub>8</sub> H <sub>18</sub> -n-C <sub>9</sub> H <sub>20</sub> <sup>a</sup>	1,0	1,001 357 085	1,0	1,000 235 044
n-C <sub>8</sub> H <sub>18</sub> -n-C <sub>10</sub> H <sub>22</sub>	1,0	1,002 553 544	1,0	1,007 186 267
n-C <sub>8</sub> H <sub>18</sub> -H <sub>2</sub> <sup>a</sup>	1,0	1,305 249 405	1,0	2,191 555 216
n-C <sub>8</sub> H <sub>18</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
n-C <sub>8</sub> H <sub>18</sub> -CO <sup>a</sup>	1,0	1,219 206 702	1,0	1,276 565 536
n-C <sub>8</sub> H <sub>18</sub> -H <sub>2</sub> O	1,0	0,599 484 191	1,0	0,662 072 469
n-C <sub>8</sub> H <sub>18</sub> -H <sub>2</sub> S	1,0	1,0	1,0	1,0
n-C <sub>8</sub> H <sub>18</sub> –He	1,0	1,0	1,0	1,0
n-C <sub>8</sub> H <sub>18</sub> -Ar	1,0	1,0	1,0	1,0
n-C <sub>9</sub> H <sub>20</sub> -n-C <sub>10</sub> H <sub>22</sub> a	1,0	1,000 810 520	1,0	1,000 182 392
n-C <sub>9</sub> H <sub>20</sub> -H <sub>2</sub> a	1,0	1,342 647 661	1,0	2,234 354 040
n-C <sub>9</sub> H <sub>20</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
n-C <sub>9</sub> H <sub>20</sub> -CO <sup>a</sup>	1,0	1,252 151 449	1,0	1,294 070 556
n-C <sub>9</sub> H <sub>20</sub> -H <sub>2</sub> O	1,0	1,0	1,0	1,0
n-C <sub>9</sub> H <sub>20</sub> -H <sub>2</sub> S	1,0	1,082 905 109	1,0	1,086 557 826
n-C <sub>9</sub> H <sub>20</sub> –He	1,0	1,0	1,0	1,0
n-C <sub>9</sub> H <sub>20</sub> –Ar	1,0	1,0	1,0	1,0
n-C <sub>10</sub> H <sub>22</sub> -H <sub>2</sub>	1,695 358 382	1,120 233 729	1,064 818 089	3,786 003 724
n-C <sub>10</sub> H <sub>22</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
	1,0	0,870 184 960	1,049 594 632	1,803 567 587
n-C <sub>10</sub> H <sub>22</sub> -CO		†	0.007.100.00	0.740.416.400
n-C <sub>10</sub> H <sub>22</sub> -CO n-C <sub>10</sub> H <sub>22</sub> -H <sub>2</sub> O	1,0	0,551 405 318	0,897 162 268	0,740 416 402
	1,0 0,975 187 766	0,551 405 318 1,171 714 677	0,897 162 268	1,103 693 489
n-C <sub>10</sub> H <sub>22</sub> -H <sub>2</sub> O				

Table E.1 (continued)

Mixture <i>i–j</i>	$eta_{ u,ij}$	$\gamma_{v,ij}$	$\beta_{T,ij}$	ΥΤ,ij
H <sub>2</sub> -O <sub>2</sub>	1,0	1,0	1,0	1,0
H <sub>2</sub> -CO	1,0	1,121 416 201	1,0	1,377 504 607
H <sub>2</sub> -H <sub>2</sub> O	1,0	1,0	1,0	1,0
H <sub>2</sub> -H <sub>2</sub> S	1,0	1,0	1,0	1,0
H <sub>2</sub> -He	1,0	1,0	1,0	1,0
H <sub>2</sub> -Ar	1,0	1,0	1,0	1,0
O <sub>2</sub> -CO	1,0	1,0	1,0	1,0
O <sub>2</sub> -H <sub>2</sub> O	1,0	1,143 174 289	1,0	0,964 767 932
O <sub>2</sub> -H <sub>2</sub> S	1,0	1,0	1,0	1,0
О2-Не	1,0	1,0	1,0	1,0
O <sub>2</sub> -Ar	0,999 746 847	0,993 907 223	1,000 023 103	0,990 430 423
CO-H <sub>2</sub> O	1,0	1,0	1,0	1,0
CO-H <sub>2</sub> S	0,795 660 392	1,101 731 308	1,025 536 736	1,022 749 748
СО-Не	1,0	1,0	1,0	1,0
CO-Ar	1,0	1,159 720 623	1,0	0,954 215 746
H <sub>2</sub> O-H <sub>2</sub> S	1,0	1,014 832 832	1,0	0,940 587 083
H <sub>2</sub> O-He	1,0	1,0	1,0	1,0
H <sub>2</sub> O-Ar	1,0	1,038 993 495	1,0	1,070 941 866
H <sub>2</sub> S-He	1,0	1,0	1,0	1,0
H <sub>2</sub> S-Ar	1,0	1,0	1,0	1,0
He-Ar	1,0	1,0	1,0	1,0
<sup>a</sup> The values of the bin	ary parameters $\gamma_{ u,ij}$ and $\gamma_{T,ij}$ we	re calculated from equa	tion (E.5).	

## **Annex F** (informative)

## **Assignment of trace components**

In order to calculate, with the use of the method described in this part of ISO 20765, the thermodynamic properties of a natural gas or similar mixture that contains trace amounts of one or more components that do not appear in <u>Table 6</u>, it is necessary to assign each trace component to one of the 21 major and minor components for which the GERG-2008 equation of state was developed. Recommendations for appropriate assignments are given in <u>Table F.1</u>.

Each recommendation is based on an assessment of which substance is likely to give the best overall compromise of accuracy for the complete set of thermodynamic properties. The factors taken into account in this assessment include molar mass, critical temperature, and critical volume. Because, however, no single assignment is likely to be equally satisfactory for all properties, it is reasonable that the user may prefer an alternative assignment for a particular application in which, for example, only a single property is needed. For this reason the recommendations are not normative. Implementations of the method that include assignments for trace components need to be carefully documented in this respect.

NOTE The additional components given in <u>Table F.1</u> are the same as those included in ISO 6976.[Z]

 ${\bf Table~F.1-Assignment~of~trace~components}$ 

Trace component	Formula	Recommended assignment
2,2-dimethylpropane (neo-pentane)	C <sub>5</sub> H <sub>12</sub>	n-pentane
2-methylpentane	C <sub>6</sub> H <sub>14</sub>	n-hexane
3-methylpentane	C <sub>6</sub> H <sub>14</sub>	n-hexane
2,2-dimethylbutane	C <sub>6</sub> H <sub>14</sub>	n-hexane
2,3-dimethylbutane	C <sub>6</sub> H <sub>14</sub>	n-hexane
ethylene (ethene)	C <sub>2</sub> H <sub>4</sub>	ethane
propylene (propene)	C <sub>3</sub> H <sub>6</sub>	propane
1-butene	C <sub>4</sub> H <sub>8</sub>	n-butane
cis-2-butene	C <sub>4</sub> H <sub>8</sub>	n-butane
trans-2-butene	C <sub>4</sub> H <sub>8</sub>	n-butane
2-methylpropene	C <sub>4</sub> H <sub>8</sub>	n-butane
1-pentene	C <sub>5</sub> H <sub>10</sub>	n-pentane
propadiene	C <sub>3</sub> H <sub>4</sub>	propane
1,2-butadiene	C <sub>4</sub> H <sub>6</sub>	n-butane
1,3-butadiene	C <sub>4</sub> H <sub>6</sub>	n-butane
acetylene (ethyne)	C <sub>2</sub> H <sub>2</sub>	ethane
cyclopentane	C <sub>5</sub> H <sub>10</sub>	n-pentane
methylcyclopentane	C <sub>6</sub> H <sub>12</sub>	n-hexane
ethylcyclopentane	C <sub>7</sub> H <sub>14</sub>	n-heptane
cyclohexane	C <sub>6</sub> H <sub>12</sub>	n-hexane
methylcyclohexane	C <sub>7</sub> H <sub>14</sub>	n-heptane
ethylcyclohexane	C <sub>8</sub> H <sub>16</sub>	n-octane

 Table F.1 (continued)

Trace component	Formula	Recommended assignment
benzene	C <sub>6</sub> H <sub>6</sub>	n-pentane
toluene (methylbenzene)	C <sub>7</sub> H <sub>8</sub>	n-hexane
ethylbenzene	C <sub>8</sub> H <sub>10</sub>	n-heptane
o-xylene	C <sub>8</sub> H <sub>10</sub>	n-heptane
all other C6 hydrocarbons		n-hexane
all other C7 hydrocarbons		n-heptane
all other C8 hydrocarbons		n-octane
all other C9 hydrocarbons		n-nonane
all other C10 hydrocarbons		n-decane
all higher hydrocarbons		n-decane
methanol (methyl alcohol)	CH <sub>4</sub> O	ethane
methanethiol (methyl mercaptan)	CH <sub>4</sub> S	propane
ammonia	H <sub>3</sub> N	methane
hydrogen cyanide	CHN	ethane
carbonyl sulphide (carbon oxysulfide)	COS	n-butane
carbon disulfide	CS <sub>2</sub>	n-pentane
sulfur dioxide	O <sub>2</sub> S	n-butane
nitrous oxide	N <sub>2</sub> O	carbon dioxide
neon	Ne	argon
krypton	Kr	argon
xenon	Xe	argon

# **Annex G** (informative)

## **Examples**

The following examples are provided for the purpose of software validation. The number of digits is not indicative of the uncertainty in the calculation, but is only given for testing purposes.

Table G.1 — Gas analysis by mole fraction

	Component	Gas 1	Gas 2	Gas 3	Gas 4	Gas 5	Gas 6
1	Methane	0,796	0,650	0,720	0,550		
2	Nitrogen	0,100	0,065	0,010		0,25	
3	Carbon dioxide	0,010	0,190	0,010		0,15	
4	Ethane	0,057	0,010	0,150			
5	Propane	0,020	0,010	0,010	0,100		0,10
6	n-Butane	0,005	0,010	0,010			0,10
7	iso-Butane	0,005	0,010	0,010			0,10
8	n-Pentane	0,002	0,010	0,010			0,10
9	iso-Pentane	0,002	0,010	0,010			0,10
10	n-Hexane	0,001	0,010	0,020			0,10
11	n-Heptane	0,001	0,020	0,010			0,10
12	n-Octane	0,001		0,010			0,10
13	n-Nonane						0,10
14	n-Decane						0,05
15	Hydrogen				0,200	0,10	
16	Oxygen			0,010	0,049 99	0,10	
17	Carbon monoxide			0,010	0,100	0,10	
18	Water				0,000 01		0,05
19	Hydrogen sulfide					0,10	
20	Helium		0,005			0,10	
21	Argon					0,10	
	Total	1,000	1,000	1,000	1,000	1,000	1,000

Table G.2 — Calculations

Gas 1		M = 19,778	8 kg/kmol							
T	p	ρ	Z	U	Н	S	$C_{v}$	$C_p$	w	μ
						kJ/	kJ/	kJ/		
К	MPa	kg/m <sup>3</sup>		kJ/kg	kJ/kg	(kg·K)	(kg·K)	(kg·K)	m/s	K/MPa
180,0	10,0	389,18	0,33956	-599,26	-573,56	-3,8855	1,6233	3,3622	796,80	0,0080269
220,0	10,0	267,65	0,40397	-450,10	-412,74	-3,0842	1,6813	5,0560	427,36	1,8447
200,0	20,0	380,27	0,62552	-560,41	-507,82	-3,6760	1,6149	3,1519	826,44	-0,052376
250,0	20,0	283,04	0,67232	-411,32	-340,66	-2,9315	1,6215	3,4875	568,65	0,77140

Table G.2 (continued)

305,0	3,0	24,835	0,94211	-135,65	-14,850	-1,1125	1,5553	2,1284	394,93	4,5112
350,0	10,0	74,667	0,91021	-100,62	33,305	-1,4321	1,7043	2,5065	430,98	2,6236
Gas 2 $M = 26,843 \text{ kg/kmol}$										
T	р	ρ	Z	U	Н	S	$C_{v}$	$C_p$	w	μ
К	МРа	kg/m³		kJ/kg	kJ/kg	kJ/ (kg·K)	kJ/ (kg·K)	kJ/ (kg·K)	m/s	K/MPa
180,0	13,0	548,61	0,42502	-539,66	-515,96	-3,0347	1,3262	2,4027	945,53	-0,28327
220,0	11,0	452,06	0,35708	-436,41	-412,08		1,3851	2,8949	615,88	0,26965
250,0	20,0	425,35	0,60721	-381,58	-334,56	-2,2525	1,4023	2,7341	615,09	0,26645
350,0	20,0	226,57	0,81426	-141,22	-52,944	-1,3037	1,5242	2,6216	414,60	1,6170
355,0	3,0	28,743	0,94921	-32,462	71,912	-0,43567	1,4383	1,8513	357,41	4,0492
400,0	10,0	88,041	0,91676	4,0059	117,59	-0,66139	1,5813	2,1613	383,33	2,4644
	-,-	/ -		,	,	.,	,	,		, -
Gas 3		M = 24.29	5 kg/kmol							
T	p	ρ	Z	U	Н	S	$C_{v}$	$C_p$	w	μ
К	МРа	kg/m <sup>3</sup>		kJ/kg	kJ/kg	kJ/ (kg·K)	kJ/ (kg·K)	kJ/ (kg·K)	m/s	K/MPa
150,0	10,0	504,68	0,38600	-711,53	-691,71	-3,9440	1,6072	2,6157	1267,9	-0,46108
200,0	10,0	431,60	0,33852	-577,62	-554,45	-3,1562	1,6072	2,9153	889,61	-0,17932
250,0	20,0	376,46	0,62095	-456,79	-403,67	-2,5938	1,6569	3,0291	731,32	0,072674
300,0	20,0	289,77	0,67228	-312,74	-243,72	-2,0114	1,7557	3,3290	523,24	0,83762
380,0	5,0	41,741	0,92111	4,6359	124,42	-0,54631	1,8869	2,4301	378,29	3,5788
400,0	10,0	82,020	0,89066	13,893	135,81	-0,73163	1,9839	2,6995	392,34	2,6674
100,0	10,0	02,020	0,0000	10,070	100,01	0). 0100		_,,	0,2,01	
Gas 4		M = 18,03	7 kg/kmol							
T	р	ρ	Z	U	Н	S	$C_{v}$	$C_p$	w	μ
К	МРа	kg/m³		kJ/kg	kJ/kg	kJ/ (kg·K)	kJ/ (kg·K)	kJ/ (kg·K)	m/s	K/MPa
180,0	25,0	378,94	0,79511	-587,59	-521,61	-3,9519	1,6680	3,1029	849,33	-0,15641
220,0	25,0	307,84	0,80080	-474,56	-393,34	-3,3091	1,6573	3,2832	660,81	0,27097
240,0	1,7	16,192	0,94898	-240,29	-135,30	-1,2079	1,5070	2,0930	371,93	5,5555
260,0	25,0	242,33	0,86077	-365,93	-262,77	-2,7635	1,6780	3,1985	564,70	0,69308
300,0	10,0	80,581	0,89738	-193,05	-68,953	-1,7099	1,6729	2,5985	425,19	2,6008
400,0	10,0	54,906	0,98774	1,9608	184,09	-0,98172	1,8988	2,5469	495,53	1,3590
100,0	20,0	2 2,700	0,20,71		101,07	0,701,1		_,=,=,=	1,5,65	
Gas 5		M = 27,610	) kg/kmol							
T	p	ρ	Z	U	Н	S	$C_{v}$	$C_p$	w	μ
К	МРа	kg/m³		kJ/kg	kJ/kg	kJ/ (kg·K)	kJ/ (kg·K)	kJ/ (kg·K)	m/s	K/MPa
150,0	30,0	704,83	0,94229	-353,11	-310,55	-2,4550	0,89982	1,7205	713,43	-0,26388
		1		1	1					
200,0	25,0	502,27	0,82643	-269,63	-219,85	-1,8878	0,84707	1,8818	485,49	0,41958

## Table G.2 (continued)

300,0	15,0	172,80	0,96087	-123,76	-36,952	-1,0017	0,79210	1,3398	387,22	1,5610
350,0	10,0	95,443	0,99409	-69,180	35,595	-0,65789	0,78914	1,1995	401,08	1,4409
400,0	5,0	41,380	1,0031	-19,179	101,65	-0,27251	0,79493	1,1359	416,75	1,2625
Gas 6		M = 81,36	5 kg/kmol							
T	p	ρ	Z	U	Н	S	$C_{\nu}$	$C_p$	W	μ
						kJ/	kJ/	kJ/		
K	MPa	kg/m³		kJ/kg	kJ/kg	(kg·K)	(kg·K)	(kg·K)	m/s	K/MPa
180,0	10,0	758,08	0,71716	-612,96	-599,77	-1,9233	1,4744	1,9300	1603,4	-0,54380
250,0	10,0	699,02	0,55998	-474,26	-459,95	-1,2685	1,6214	2,0839	1283,4	-0,48029
300,0	25,0	671,21	1,2150	-374,55	-337,30	-0,90152	1,7870	2,2353	1184,5	-0,43573
400,0	5,0	549,38	0,22266	-114,62	-105,52	-0,14589	2,1625	2,7652	645,44	-0,091974
450,0	0,5	11,811	0,92057	247,23	289,56	0,86891	2,2505	2,4030	203,37	11,662
500,0	2,0	52,175	0,75024	337,90	376,23	0,93137	2,5111	2,8942	176,26	11,031

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