

BS EN 4685:2011



BSI Standards Publication

**Aerospace series — Titanium  
Ti10V2Fe3Al — Bars —  
D < 110 mm — Rm  $\geq$  1 240 MPa**

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**National foreword**

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The UK participation in its preparation was entrusted to Technical Committee ACE/61/-/49, Titanium and its Alloys for Aerospace Purposes.

A list of organizations represented on this committee can be obtained on request to its secretary.

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EUROPEAN STANDARD

**EN 4685**

NORME EUROPÉENNE

EUROPÄISCHE NORM

June 2011

ICS 49.025.30

English Version

**Aerospace series - Titanium Ti10V2Fe3Al - Bars -  $D < 110$  mm -  
 $R_m \geq 1\ 240$  MPa**

Série aérospatiale - Titane Ti10V2Fe3Al - Barres -  $D < 110$   
mm -  $R_m \geq 1\ 240$  MPa

Luft- und Raumfahrt - Titan Ti10V2Fe3Al - Stangen -  $D <$   
110 mm -  $R_m \geq 1\ 240$  MPa

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**Management Centre: Avenue Marnix 17, B-1000 Brussels**

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## Foreword

This document (EN 4685:2011) has been prepared by the Aerospace and Defence Industries Association of Europe - Standardization (ASD-STAN).

After enquiries and votes carried out in accordance with the rules of this Association, this Standard has received the approval of the National Associations and the Official Services of the member countries of ASD, prior to its presentation to CEN.

This European Standard shall be given the status of a national standard, either by publication of an identical text or by endorsement, at the latest by December 2011, and conflicting national standards shall be withdrawn at the latest by December 2011.

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## Introduction

This European Standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This European Standard has been prepared in accordance with EN 4500-4.

## 1 Scope

This European Standard specifies the requirements relating to:

Titanium Ti-10V-2Fe-3Al  
Bars  
 $D < 110$  mm  
 $R_m \geq 1\,240$  MPa

for aerospace applications.

## 2 Normative references

The following referenced documents are indispensable for the application of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

EN 2032-2, *Aerospace series — Metallic materials — Part 2: Coding of metallurgical condition in delivery condition*

EN 2954 (all parts), *Aerospace series — Macrostructure of titanium and titanium alloy wrought products*

EN 3684, *Aerospace series — Test methods — Titanium alloy wrought products — Determination of  $\beta$  transus temperature — Metallographic method*

EN 4050-1, *Aerospace series — Test method for metallic materials — Part 1: General requirements*

EN 4258, *Aerospace series — Metallic materials — General organization of standardization — Links between types of EN standards and their use*

EN 4500-4, *Aerospace series — Metallic materials — Rules for drafting and presentation of material standards — Part 4: Specific rules for titanium and titanium alloys* <sup>1)</sup>

EN 4800-002, *Aerospace series — Titanium and titanium alloys — Technical specification — Part 002: Bar and section* <sup>1)</sup>

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1) Published as ASD-STAN Prestandard at the date of publication of this standard.

1	Material designation		Titanium Ti-10V-2Fe-3Al									
2	Chemical composition %	Element	Al	V	Fe	O	C	N	Y	Other		Ti
			Each	Total								
		min.	2,60	9,00	1,60	–	–	–	–	–	–	–
max.	3,40	11,00	2,20	0,13	0,05	0,05	0,005	0,10	0,30			
3	Method of melting		Vacuum melted. Last melting by consumable electrode									
4.1	Form		Bars									
4.2	Method of production		Hot rolling									
4.3	Limit dimension(s)	mm	< 110									
5	Technical specification		EN 4800-002									
6.1	Delivery condition		Not heat treated									
	Heat treatment		–									
6.2	Delivery condition code		F according to EN 2032-2									
7	Use condition		Solution treated and aged									
	Heat treatment		[33-56] °C below the beta transus / ≥ 1 h / WQ + ≥ 480 °C / ≥ 8 h / AC									

Characteristics

8.1	Test sample(s)		–									
8.2	Test piece(s)		–									
8.3	Heat treatment		Solution treated and aged									
9	Dimensions concerned	mm	≤ 75				75 < D < 110					
10	Thickness of cladding on each face	%	–									
11	Direction of test piece		L			L			T <sup>a</sup>			
12	Temperature	$\theta$	°C	Ambient								
13	Proof stress	R <sub>p0,2</sub>	MPa	≥ 1 110			≥ 1 110			≥ 1 110		
14	T Strength	R <sub>m</sub>	MPa	≥ 1 240			≥ 1 240			≥ 1 240		
15	Elongation	A	%	≥ 4			≥ 4			≥ 4		
16	Reduction of area	Z	%	≥ 10			≥ 10			≥ 8		
17	Hardness		–									
18	Shear strength	R <sub>c</sub>	MPa	–								
19	Bending	k	–	–								
20	Impact strength		–									
21	Temperature	$\theta$	°C	–								
22	Time		h	–								
23	C Stress	$\sigma_a$	MPa	–								
24	Elongation	a	%	–								
25	Rupture stress	$\sigma_R$	MPa	–								
26	Elongation at rupture	A	%	–								
27	Notes (see line 98)		a									

30	Microstructure	–	[33-56] °C below the beta transus / ≥ 1 h / WQ + ≥ 480 °C / ≥ 8 h / AC			
		7	Shall consist of primary alpha phase in a matrix of aged beta phase. An unbroken continuous alpha phase network along prior beta phase grain boundaries is not acceptable.			
44	External defects	–	See EN 4800-002.			
		6	Visual			
51	Macrostructure	1	See EN 4800-002.			
		7	Delivery condition		Beta transus – 25 °C / 1 h / WQ	
			<i>D</i> mm	Maximum acceptable macrostructure EN 2954		No beta fleck. (No area without primary alpha phase greater than [0,76 × 0,76] mm). Inspection from locations representing the top and the bottom end of each batch.
			≤ 50	2MA2		
50 < <i>D</i> ≤ 110	2MA3					
61	Internal defects	–	See EN 4800-002.			
		1	EN 4050-1			
		7	Class 5			
86	β-transus temperature	–	See EN 3684.			
95	Marking inspection	–	–			
96	Dimensional inspection	–	–			
98	Notes	–	<sup>a</sup> Test sample over 55 mm thickness.			
99	Typical use	–	–			



100	-	Product qualification	-	-
Qualification programme to be agreed between manufacturer and purchaser.				





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