Aluminium alloy AL-P2024-T42 — Plate — $6 \text{ mm} < a \leq 140 \text{ mm}$

The European Standard EN 4247:2005 has the status of a British Standard

ICS 49.025.20



National foreword

This British Standard is the official English language version of EN 4247:2005.

The UK participation in its preparation was entrusted by Technical Committee ACE/61, Metallic materials for aerospace purposes, to Subcommittee ACE/61/-/24, Light alloys, which has the responsibility to:

- aid enquirers to understand the text;
- present to the responsible international/European committee any enquiries on the interpretation, or proposals for change, and keep UK interests informed;
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Summary of pages

This document comprises a front cover, an inside front cover, the EN title page, pages 2 to 6, an inside back cover and a back cover.

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EUROPÄISCHE NORM

EN 4247

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English version

Aerospace series - Aluminium alloy AL-P2024-T42 - Plate - 6 mm <a ≤140 mm

Série aérospatiale - Alliage d'aluminium AL-P2024-T42 - Tôles épaisses - 6 mm <a ≤140 mm Luft- und Raumfahrt - Aluminiumlegierung AL-P2024-T42 - Platten - 6 mm <a ≤140 mm

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EUROPEAN COMMITTEE FOR STANDARDIZATION COMITÉ EUROPÉEN DE NORMALISATION EUROPÄISCHES KOMITEE FÜR NORMUNG

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Foreword

This document (EN 4247:2005) has been prepared by the European Association of Aerospace Manufacturers - Standardization (AECMA-STAN).

After enquiries and votes carried out in accordance with the rules of this Association, this Standard has received the approval of the National Associations and the Official Services of the member countries of AECMA, prior to its presentation to CEN.

This European Standard shall be given the status of a national standard, either by publication of an identical text or by endorsement, at the latest by December 2005, and conflicting national standards shall be withdrawn at the latest by December 2005.

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Introduction

This standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This standard has been prepared in accordance with EN 4500-2.

1 Scope

This standard specifies the requirements relating to:

Aluminium alloy AL-P2024-T42 Plate $6 \text{ mm} < a \le 140 \text{ mm}$

for aerospace application.

2 Normative references

The following referenced documents are indispensable for the application of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

EN 4258, Aerospace series — Metallic materials — General organization of standardization — Links between types of EN standards and their use.

EN 4400-1, Aerospace series — Aluminium and aluminium alloy wrought products — Technical specification — Part 1: Plate. 1)

EN 4500-2, Aerospace series — Metallic materials — Rules for drafting and presentation of material standards — Part 2: Specific rules for aluminium, aluminium alloys and magnesium alloys. 1)

¹⁾ Published as AECMA Prestandard at the date of publication of this standard.

EN 4247:2005

1	Material designation			Aluminium alloy AL-P2024-										
2	Chemical Element			Si	Fe	Cu	Mn	Mg	Cr	Zn	Ti	Others		Al
	composition	Element	OI	16	Cu	IVIII	ivig	Ci	4 11	''	Each	Total	\ \text{\rm I}	
	%	min.		-	1	3,8	0,30	1,2	-	-	-	_	1	Base
	max.			0,50	0,50	4,9	0,9	1,8	0,10	0,25	0,15	0,05	0,15	Dase
3	Method of melting													
4.1	Form			Plate										
4.2	Method of production			Rolled										
4.3	Limit dimension(s) mm			6 < a ≤ 140										
5	Technical specification			EN 4400-1										

6.1	Delivery condition	F
	Heat treatment	-
6.2	Delivery condition code	F
7	Use condition	T42
	Heat treatment	Delivery condition $ +490 ^{\circ}\text{C} \leq \theta \leq 500 ^{\circ}\text{C} / \text{WQ} \theta \leq 40 ^{\circ}\text{C} \\ +\theta = \text{ambient} / \text{t} \geq 5 \text{d} $

Characteristics

8.1	Τe	est sample(s)			See EN 4400-1.							
8.2	2 Test piece(s)				See EN 4400-1.							
8.3	Не	eat treatment			Use condition.							
9	Di	mensions concerne	ed	mm	6 < a \le 12,5 12,5 < a \le 25 25 < a \le 50 50 < a \le 100 100 < a \le 100							
10	Th ea	ickness of cladding ich face	on	%	-	-	-	_	-			
11	Di	rection of test piece	;		LT LT LT LT				LT			
12		Temperature	θ	°C	Ambient	Ambient	Ambient	Ambient	Ambient			
13	Т	Proof stress	R _{p0,2}	MPa	≥ 260	≥ 260	≥ 255	≥ 255	≥ 245			
14] '	Strength	R _m	MPa	≥ 420	≥ 410	≥ 400	≥ 390	≥ 380			
15		Elongation	Α	%	$A_{50\;mm} \geq 12$	≥ 12	≥ 9	≥ 8	≥ 7			
16		Reduction of area	Z	%			_					
17	' Hardness				-							
18	Shear strength R _c MPa				-							
19	Bending k -			_	-							
20	Impact strength				-							
21		Temperature	θ	°C	-							
22		Time		h	-							
23	С	Stress	σ _a MPa		-							
24		Elongation	а		-							
25		Rupture stress	σ_{R}	MPa	-							
26		Elongation at rupture	Α	%			-					
27	No	otes (see line 98)					_					

			EN 4247:2005
44	External defects	_	See EN 4400-1.
95	Marking inspection	-	See EN 4400-1.
96	Dimensional inspection	_	See EN 4400-1.
98	Notes	_	_
	Typical use	_	_
99	i ypicai use		=

EN 4247:2005

100	-	roduct qualification –		See EN 4400-1.
				Qualification programme to be agreed between manufacturer and purchaser.



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