Aluminium alloy AL-P8090-T89 — Sheet — $0,6 \text{ mm} \leq a \leq 6 \text{ mm}$

The European Standard EN 4203:2005 has the status of a British Standard

ICS 49.025.20



National foreword

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The UK participation in its preparation was entrusted by Technical Committee ACE/61, Metallic materials for aerospace purposes, to Subcommittee ACE/61/-/24, Light alloys, which has the responsibility to:

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Summary of pages

This document comprises a front cover, an inside front cover, the EN title page, pages 2 to 6, an inside back cover and a back cover.

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EUROPEAN STANDARD

EN 4203

NORME EUROPÉENNE EUROPÄISCHE NORM

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English version

Aerospace series - Aluminium alloy AL-P8090-T89 - Sheet - 0,6 mm ≤a ≤6 mm

Série aérospatiale - Alliage d'aluminium AL-P8090-T89 - Tôles - 0,6 mm \leq a \leq 6 mm

Luft- und Raumfahrt - Aluminiumlegierung AL-P8090-T89 - Bleche - 0,6 mm \leq a \leq 6 mm

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EUROPEAN COMMITTEE FOR STANDARDIZATION COMITÉ EUROPÉEN DE NORMALISATION EUROPÄISCHES KOMITEE FÜR NORMUNG

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Foreword

This document (EN 4203:2005) has been prepared by the European Association of Aerospace Manufacturers - Standardization (AECMA-STAN).

After enquiries and votes carried out in accordance with the rules of this Association, this Standard has received the approval of the National Associations and the Official Services of the member countries of AECMA, prior to its presentation to CEN.

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Introduction

This standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This standard has been prepared in accordance with EN 4500-2.

1 Scope

This standard specifies the requirements relating to:

Aluminium alloy AL-P8090-T89 Sheet $0.6 \text{ mm} \le a \le 6 \text{ mm}$

for aerospace application.

2 Normative references

The following referenced documents are indispensable for the application of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

EN 4258, Aerospace series — Metallic materials — General organization of standardization — Links between types of EN standards and their use.

EN 4400-2, Aerospace series — Aluminium and aluminium alloy wrought products — Technical specification — Part 2: Sheet and strip. 1)

EN 4500-2, Aerospace series — Metallic materials — Rules for drafting and presentation of material standards — Part 2: Specific rules for aluminium, aluminium alloys and magnesium alloys. ¹⁾

¹⁾ Published as AECMA Prestandard at the date of publication of this standard.

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1	1 Material designation							Α	luminiun	n alloy A	L-P809	0-											
2	Chemical	Chemical Element			Element	*		Si	0:		C:	Fe	Cu	Mn	Mg	Cr	Zn	Li	Zr	Ti	Oth	Others	
	composition			51	D	Ö	IVIII	y y	5	211	ם י	۷۱	11	Each	Total	Al							
	%	min.		_	1	1,0	-	0,6	ı	-	2,2	0,04	_	_	ı	Base							
		max.		0,20	0,30	1,6	0,10	1,3	0,10	0,25	2,7	0,16	0,10	0,05 ^a	0,15	Dase							
3	Method of melting									-													
4.1	1 Form									Sheet													
4.2	2 Method of production									Rolled													
4.3	3 Limit dimension(s) mm								0	,6 ≤ <i>a</i> ≤	6												
5	5 Technical specification			•		•			Е	N 4400-	2		•		•	-							

6.1	Delivery condition	Т39	Т89
	Heat treatment	525 °C $\leq \theta \leq$ 535 °C / WQ $\theta \leq$ 40 °C + 4,0 % \leq controlled stretched \leq 4,5 % + θ = ambient / t \geq 5 d	$525 ^{\circ}\text{C} \le \theta \le 535 ^{\circ}\text{C} \ / \ \text{WQ} \ \theta \le 40 ^{\circ}\text{C} \ + 4,0 \% \le \text{controlled stretched} \le 4,5 \% \ + 162 ^{\circ}\text{C} \le \theta \le 168 ^{\circ}\text{C} \ / \ 44 \text{h} \le \text{t} \le 46 \text{h}$
6.2	Delivery condition code	К	U
7	Use condition	Т89	Т89
	Heat treatment	Delivery condition + 162 °C $\leq \theta \leq$ 168 °C / 44 h \leq t \leq 46 h	Delivery condition

Characteristics

8.1	Test sample(s)			See EN 4400-2.				
8.2	Test piece(s)			Test piece(s)				See EN 4400-2.
8.3 Heat treatment					Use condition.			
9 Dimensions concerned mm $0,6 \le a \le 6$				0,6 ≤ <i>a</i> ≤ 6				
10	Th ea	ickness of cladding ch face	on	%	-			
11	Di	rection of test piece)		LT			
12		Temperature	θ	°C	Ambient			
13		Proof stress	R _{p0,2}	MPa	≥ 340			
14	Т	Strength	R _m	MPa	≥ 440			
15		Elongation	Α	%	≥ 6			
16		Reduction of area	Z	%	-			
17	' Hardness			-				
18	Shear strength R _c MPa		Shear strength R _c MF		MPa	-		
19	9 Bending k -		Bending k		-	-		
20	lm	pact strength			-			
21		Temperature	θ	°C	-			
22		Time		h	-			
23	С	Stress	σ_{a}	MPa	-			
24		Elongation	а	%	-			
25		Rupture stress	σ_{R}	MPa	-			
26		Elongation at rupture	Α	%	-			
27					a			

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30	Microstructure	- See EN 4400-2.							
		3	a ≤ 4 mm			4 mm < <i>a</i>	4 mm < a ≤ 6 mm		
		7	Microstructure shall be	fully recryst	allized	to agreement between	/stallization shall be subject /een the manufacturer and ourchaser		
40	Fracture toughness (K _c)	_			See EN	· · · · · · · · · · · · · · · · · · ·			
		3			T-	L			
		7			≥ 40 MF	Pa √m			
44	External defects	-			See EN	4400-2.			
49	Exfoliation corrosion	-			See EN	4400-2.			
		6			t = 4	8 h			
		7	Exfoliation	n corrosion	shall not be	e greater than that of grad	de EB		
68	Density	_			See EN	4400-2.			
		7			ρ≤2,56 k	kg dm ⁻³			
82	Batch uniformity	_			See EN	4400-2.			
		5	-			T39	T89		
		7	Electrical conductivity	γ	MS/m	8,5 (Typical value)	11,4 (Typical value)		
				l .	10				
				-		95 (Typical value)	142 (Typical value)		
			Hardness	δ	НВ	≤ 20 per product	≤ 20 per product		
				Δ	1	≤ 30 per batch	≤ 30 per batch		
95	Marking inspection	_			See EN	4400-2.			
96	Dimensional inspection	_			See EN	4400-2.			
98	Notes	_	^a Na ≤ 10 ppm, Ca ≤ 13	20 ppm.					
99	Typical use	_			_				

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100	-	Product qualification	-	See EN 4400-2.
				Qualification programme to be agreed between manufacturer and purchaser.



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