

**BS EN 3312:2012**

*Incorporating corrigendum March 2012*



**BSI Standards Publication**

**Aerospace series — Titanium  
alloy Ti-6Al-4V — Annealed  
— Forgings  $D_e \leq 150$  mm**

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**National foreword**

This British Standard is the UK implementation of EN 3312:2012. It supersedes BS 3TA 13:2009, which is withdrawn.

The UK participation in its preparation was entrusted by Technical Committee ACE/61, Metallic materials for aerospace purposes, to Subcommittee ACE/61/-/49, Titanium and its Alloys for Aerospace Purposes.

A list of organizations represented on this subcommittee can be obtained on request to its secretary.

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Date	Text affected
31 March 2012	Implementation of CEN correction notice 2 February 2012: Table on page 6 amended

ICS 49.025.30

English Version

## Aerospace series - Titanium alloy Ti-6Al-4V - Annealed - Forgings $De \leq 150$ mm

Série aérospatiale - Alliage de titane Ti-6Al-4V - Recuit -  
Pièces forgées et matricées -  $De \leq 150$  mm

Luft- und Raumfahrt - Titanlegierung Ti-6Al-4V - Geglüht -  
Schmiedestücke -  $De \leq 150$  mm

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## **Foreword**

This document (EN 3312:2012) has been prepared by the Aerospace and Defence Industries Association of Europe - Standardization (ASD-STAN).

After enquiries and votes carried out in accordance with the rules of this Association, this Standard has received the approval of the National Associations and the Official Services of the member countries of ASD, prior to its presentation to CEN.

This European Standard shall be given the status of a national standard, either by publication of an identical text or by endorsement, at the latest by August 2012, and conflicting national standards shall be withdrawn at the latest by August 2012.

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## **Introduction**

This European Standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This European Standard has been prepared in accordance with EN 4500-4.

## 1 Scope

This European Standard specifies the requirements relating to: <sup>1)</sup>

Titanium alloy Ti-6Al-4V  
Annealed  
Forgings  
 $D_e \leq 150$  mm

for aerospace applications.

## 2 Normative references

The following referenced document, in whole or in part, are normatively referenced in this document and are indispensable for its application. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

EN 2003-009, *Aerospace series — Test methods — Titanium and titanium alloys — Part 009: Determination of surface contamination*

EN 2032-2, *Aerospace series — Metallic materials — Part 2: Coding of metallurgical condition in delivery condition*

EN 2954-002, *Aerospace series — Macrostructure of titanium and titanium alloy wrought products — Part 002: Macrostructure of bar, section, forging stock and forgings*

EN 3114-002, *Aerospace series — Test method — Microstructure of ( $\alpha + \beta$ ) titanium alloy wrought products — Part 002: Microstructure of bars, sections, forging stock and forgings*

EN 3310, *Aerospace series — Titanium alloy Ti-P64001 (Ti-6Al-4V) — Not heat treated — Forging stock, for annealed forgings —  $D_e \leq 360$  mm <sup>2)</sup>*

EN 4050-4, *Aerospace series — Test method for metallic materials — Part 4: Acceptance criteria*

EN 4800-005, *Aerospace series — Titanium and titanium alloys — Technical specification — Part 005: Forging stock*

EN 4800-006, *Aerospace series — Titanium and titanium alloys — Technical specification — Part 006: Pre-production and production forgings*

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1) Grade 2 according to EN 4800-005.

2) Published as ASD-STAN Prestandard at the date of publication of this standard ([www.asd-stan.org](http://www.asd-stan.org)).

**BS EN 3312:2012**  
**EN 3312:2012 (E)**

1	Material designation		Titanium alloy Ti-6Al-4V										
2	Chemical composition %	Element	Al	V	O+2N	N	H	Fe	C	Y	Others		Ti
											Each	Total	
		min.	5,50	3,50	–	–	–	–	–	–	–	–	–
max.	6,75	4,50	0,25	0,03	0,012 5	0,30	0,08	0,005 0	0,10	0,40			
3	Method of melting		Grade 2 – see EN 4800-005.										
4.1	Form		Forgings										
4.2	Method of production		Forged from forging stock EN 3310										
4.3	Limit dimension(s)	mm	$D_e \leq 150$										
5	Technical specification		EN 4800-006										

6.1	Delivery condition		Annealed									
	Heat treatment		$690\text{ °C} \leq \theta \leq 840\text{ °C} / t \geq 30\text{ min} / \text{AC}$ or cool in inert atmosphere									
6.2	Delivery condition code		U <sup>a</sup>									
7	Use condition		Delivery condition									
	Heat treatment		–									

Characteristics

8.1	Test sample(s)		See EN 4800-006.										
8.2	Test piece(s)		See EN 4800-006.										
8.3	Heat treatment		Use condition										
9	Dimensions concerned	mm	$D_e \leq 150$										
10	Thickness of cladding on each face	%	–										
11	Direction of test piece		L			LT			ST				
12	Temperature	$\theta$	°C	Ambient									
13	Proof stress	$R_{p0,2}$	MPa	$\geq 830$			$\geq 830$			$\geq 830$			
14	T Strength	$R_m$	MPa	$900 \leq R_m \leq 1\ 160$			$900 \leq R_m \leq 1\ 160$			$900 \leq R_m \leq 1\ 160$			
15	Elongation	A	%	$\geq 10$			$\geq 8$			$\geq 6$			
16	Reduction of area	Z	%	$\geq 25$			$\geq 20$			$\geq 15$			
17	Hardness		–										
18	Shear strength	$R_c$	MPa	–									
19	Bending	k	–	–									
20	Impact strength		–										
21	Temperature	$\theta$	°C	–									
22	Time		h	–									
23	C Stress	$\sigma_a$	MPa	–									
24	C Elongation	a	%	–									
25	C Rupture stress	$\sigma_R$	MPa	–									
26	C Elongation at rupture	A	%	–									
27	Notes (see line 98)		a										



30	Microstructure	–	See EN 4800-006.			
		1	See EN 3114-002.			
		7	$D_e$ mm	Acceptable microstructure	Not acceptable microstructure	
			$D_e \leq 25$	From 2 T 1 to 2 T 6 Occasionally 2 T 8	2 T 7 and 2 T 9 From 2 T 10 to 2 T 15	
				From 2 L 1 to 2 L 4	From 2 L 5 to 2 L 15	
				2 T 100, 2 T 102	2T 101 From 2 T 103 to 2 T 117	
				2 A 1, 2 A 2	–	
		–	From 2 T 200 to 2 T 201			
		$25 < D_e < 150$	From 2 T 1 to 2 T 7 Occasionally 2 T 8 to 2 T 12	From 2 T 13 to 2 T 15		
			From 2 L 1 to 2 L 6, 2 L 8, 2 L 10 Occasionally 2 L 7, 2 L 9, 2 L 12	2 L 11 From 2 L 13 to 2 L 15		
2 T 100, 2 T 102 Occasionally 2 T 103, 2 T 109 to 2 T 111	2 T 101 From 2 T 104 to 2 T 108 From 2 T 112 to 2 T 117					
From 2 T 108 to 2 T 110, if number of defects less than 5 per cm <sup>2</sup> of the sampling section	From 2 T 108 to 2 T 110, if number of defects less than 5 per cm <sup>2</sup> of the sampling section					
–	From 2 T 111 to 2 T 117					
2 A 1	–					
–	From 2 T 200 to 2 T 201					
44	External defects	–	See EN 4800-006.			
		1	Visual examination			
51	Macrostructure	–	See EN 4800-006.			
		1	EN 2954-002			
		7	$D_e$ mm	Maximum acceptable macrostructure	Not acceptable macrostructure	Macrostructure submitted for approval
$50 < D_e < 110$	2 MA 3		2 MA 80 to 2 MA 84 and 2 MA 100	2 MA 40 to 2 MA 42 and 2 MA 60 to 2 MA 62		
61	Internal defects	–	See EN 4800-006.			
		1	EN 4050-4			
		7	Class 3			
74	Surface contamination	–	See EN 4800-006.			
		1	EN 2003-009			
		2	Pre-production part unless otherwise specified			
		3	See inspection schedule			
82	Batch uniformity (Material verification)	–	See EN 4800-006.			
95	Marking inspection	–	–			
96	Dimensional inspection	–	–			
98	Notes	–	<sup>a</sup> According to EN 2032-2.			
99	Typical use	–	–			

100	-	Product qualification	-	See EN 4800-006.
				Qualification programme to be agreed between manufacturer and purchaser.

## **Bibliography**

EN 4258, *Aerospace series — Metallic materials — General organization of standardization — Links between types of EN standards and their use*

EN 4500-4, *Aerospace series — Metallic materials — Rules for drafting and presentation of material standards — Part 4: Specific rules for titanium and titanium alloys* <sup>2)</sup>





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