Aluminium alloy AL-P7010-T7651 — Plate — 6 mm < a \leq 140 mm

The European Standard EN 2684:2004 has the status of a British Standard

ICS 49.025.20



National foreword

This British Standard is the official English language version of EN 2684:2004.

The UK participation in its preparation was entrusted to Technical Committee ACE/61/-/24, Light alloys, which has the responsibility to:

- aid enquirers to understand the text;
- present to the responsible international/European committee any enquiries on the interpretation, or proposals for change, and keep the UK interests informed;
- monitor related international and European developments and promulgate them in the UK.

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EUROPEAN STANDARD NORME EUROPÉENNE

EN 2684

EUROPÄISCHE NORM

December 2004

ICS 49.025.20

English version

Aerospace series - Aluminium alloy AL-P7010- - T7651 - Plate - $6 \text{ mm} < a \le 140 \text{ mm}$

Série aérospatiale - Alliage d'aluminium AL-P7010- - T7651 - Tôle épaisse - 6 mm < a \leq 140 mm

Luft- und Raumfahrt - Aluminiumlegierung AL-P7010- - T7651 - Platten - 6 mm < a \leq 140 mm

This European Standard was approved by CEN on 15 July 2004.

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Foreword

This document (EN 2684:2004) has been prepared by the European Association of Aerospace Manufacturers - Standardization (AECMA-STAN).

After enquiries and votes carried out in accordance with the rules of this Association, this Standard has received the approval of the National Associations and the Official Services of the member countries of AECMA, prior to its presentation to CEN.

This European Standard shall be given the status of a national standard, either by publication of an identical text or by endorsement, at the latest by June 2005, and conflicting national standards shall be withdrawn at the latest by June 2005.

According to the CEN/CENELEC Internal Regulations, the national standards organizations of the following countries are bound to implement this European Standard: Austria, Belgium, Cyprus, Czech Republic, Denmark, Estonia, Finland, France, Germany, Greece, Hungary, Iceland, Ireland, Italy, Latvia, Lithuania, Luxembourg, Malta, Netherlands, Norway, Poland, Portugal, Slovakia, Slovenia, Spain, Sweden, Switzerland and the United Kingdom.

Introduction

This standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This standard has been prepared in accordance with EN 4500-2.

1 Scope

This standard specifies the requirements relating to:

Aluminium alloy AL-P7010-T7651 Plate $6 \text{ mm} < a \le 140 \text{ mm}$

for aerospace applications.

2 Normative references

The following referenced documents are indispensable for the application of this document. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

- EN 4258, Aerospace series Metallic materials General organization of standardization Links between types of EN standards and their use
- EN 4400-1, Aerospace series Aluminium and aluminium alloy wrought products Technical Specification Part 1: Plate 1)
- EN 4500-2, Aerospace series Metallic materials Rules for drafting and presentation of material standards Part 2: Specific rules for aluminium, aluminium alloys and magnesium alloys ¹⁾

¹⁾ Published as AECMA Prestandard at the date of publication of this standard

4	1.1.	-4		<u> </u>					^	li iina lin li		AL D	7040						
1	IVI	aterial designatio		Aluminium alloy AL-P7010- Others															
2		Chemical composition	Elemen	t	Si	Fe	Cu		Mn	Mg	Cr	١	li Z	n Z	'r	Ti	Each	Total	Al
		%	min.		-	-	1,5		-	2,1	_	-	- 5		10	_	_	_	Base
			max.		0,12	0,15	2,0	(0,10	2,6	0,05	0,	05 6	,7 0,	16	0,06	0,05	0,15	
3	Me	ethod of melting											_						
4.1		orm											ate						
4.2		ethod of producti	on	1	Rolled														
4.3		mit dimension(s)		mm	6 < a ≤ 140														
5	Те	echnical specifica	ition		EN 4400-1														
6.1	De	elivery condition			W51 T7651														
					470 °C ≤ θ≤ 485 °C ° / WQ θ≤ 40 °C														
	He	eat treatment			$470 ^{\circ}\text{C} \le \theta \le 485 ^{\circ}\text{C} / \text{WQ} \theta \le 40 ^{\circ}\text{C} $ $+ 1,5 ^{\circ}\text{S} \le \text{controlled stretched} \le 3 ^{\circ}\text{C} $ $+ 115 ^{\circ}\text{C} \le \theta$									≤ θ ≤	ontrolled stretched ≤ 3 % l≤ 125 °C / 4 h ≤ t ≤ 24 h ^a l≤ 175 °C / 6 h ≤ t ≤ 15 h ^a				
6.2	De	elivery condition	code			W										U	l		
7	Us	se condition						T76								T76	51		
	He	eat treatment			+	115 °C	Deliv	ery (condition °C / 4 h	n < t < :	24 h ^a				Del	livery o	condition		
									°C / 6 h		15 h ^a				D01		, or randor r		
0.4	_												teristics						
8.1		est sample(s)				See EN 4400-1.													
8.2		est piece(s)				See EN 4400-1.													
8.3		eat treatment	I	T	Use condition										^				
9		mensions conce	mm %	6 < a ≤ 12,5 12,5 < a ≤ 25						25 < a ≤ 40 40 < a ≤ 60									
10	ea	ich face		 															
11	ווט	Temporature	θ	°C	L	Ambio	LT		L	A mala	LT		L	LT		51	L	LT Ambient	ST
12		Temperature Proof stress	R _{p0,2}		≥ 45	Ambie	÷iii. ≥ 450	١	≥ 45	Amb	≥ 45	:0	≥ 450	Ambier ≥ 450		415	≥ 445	≥ 440	≥ 400
14	т	Strength	R _m	MPa	≥ 52		≥ 525		≥ 52		≥ 52		≥ 515	≥ 515		490	≥ 515	≥ 515	≥ 490
15		Elongation	A	%	A _{50mm} 2		≥ 320 A _{50mm} ≥		≥ 32	-	≥ 52		≥ 7	≥ 5		430 ≥ 3 ^b	≥ 7	≥ 5	≥ 430 ≥ 3 ^b
16		Reduction of are		%	7 (50mm =	- 0	/ \50mm =	. •					'			- 0	= /	_ 0	
	tinu		2 2	,,,															
9		mensions conce		mm	60 < a ≤ 80 80 < a ≤ 100							100 < a ≤ 120 120 < a ≤ 140					40		
10	Th ea	nickness of cladd och face	ing on	%							-								
11	Dii	rection of test pie	ece		L	LT	S	Т	L	L٦	- (ST	L	LT		ST	L	LT	ST
12		Temperature	θ	°C		Ambie	ent			Amb	ent			Ambier	nt			Ambient	
13		Proof stress	R _{p0,2}	MPa	≥ 440	≥ 43	5 ≥ 3	90	≥ 435	≥ 43	30 ≥	390	≥ 430	≥ 430	≥	380	≥ 430	≥ 425	≥ 370
14	Т	Strength	R _m	MPa	≥ 505	≥ 51			≥ 500	≥ 50		480	≥ 495	≥ 500		470	≥ 490	≥ 495	≥ 460
15		Elongation	Α	%	≥ 6	≥ 5	≥ 3	3 ^b	≥ 6	≥ ;	5 ≥ 3	2,5 ^c	≥ 6	≥ 5	≥	2,5 ^c	≥ 5	≥ 4	\geq 2,5 c
16		Reduction of are	ea Z	%								_	-						
17	На	ardness	-																
18	<u> </u>					-													
19	Ů Ů					-													
20	<u> </u>				+														
21	Temperature θ °C			_															
22		Time		h								-	_						
23	С	Stress	σ _a	MPa									-						
24		Elongation	а	%									-						
25		Rupture stress	σ_{R}	MPa								-	-						
26		Elongation at rupture	Α	%									-						
27	7 Notes (see line 98)																		

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32	Electrical conductivity			See El	N 4400-1.						
		7	$\gamma \ge 22,5$ MS/m Acceptable								
			21,5 MS/m ≤ γ < 22,5 MS	/m Accep	otable if $R_{p0,2}$ LT $\leq R_{p0,2}$ m if stress corrosion test is	in. LT + 70 MPa acceptable.					
			γ < 21,5 MS/m		Not acceptable						
39	Stress corrosion	-		See EN 4400-1.							
		6		σ = 1	75 MPa						
		7		t≥	20 d						
40	Fracture toughness (K _{IC})	-	See EN 4400-1.								
		2	The "capability clause" applies.								
		7	Dimensions mm	L-T MPa √m	T-L MPa √m	S-L MPa √m					
			25 < a ≤ 50	≥ 28	≥ 25	-					
			50 < a ≤ 75	≥ 27	≥ 25	≥ 23					
			75 < a ≤ 100	≥ 26	≥ 24	≥ 23					
			100 < a ≤ 140	≥ 25	≥ 23	≥ 22					
44	External defects	_		See El	N 4400-1.	1					
47	Notch/yield ratio R _e / R _{p0,2}	-	See EN 4400-1.								
49	Exfoliation corrosion	-	See EN 4400-1.								
		7	Exfoliation shall not be greater than that of grade EB.								
61	Internal defects	-	See EN 4400-1.								
82	Batch uniformity	-	See EN 4400-1.								
95	Marking inspection	_	See EN 4400-1.								
96	Dimensional inspection	_		See El	N 4400-1.						
98	Notes	-	Artificial ageing may be carried out using the following alternative single stage method : heating to a temperature of 165 °C $\leq \theta \leq$ 175 °C at a rate not exceeding 20 °C / h and soaking at this temperature for 6 h \leq t \leq 15 h. Or $A_{4D} \geq 3,5$ if required by the purchaser Or $A_{4D} \geq 3$ if required by the purchaser								
99	Typical use	-			-						

100	_	Product qualification	-	See EN 4400-1.
				Qualification programme to be agreed between manufacturer and purchaser.

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