Incorporating Amendment Nos. 1 and 2

Specification for

Coupling dimensions for aero-engine refrigerant pressure replenishment connections

Confirmed January 2012

 $UDC\ 629.7.03{:}621.43 - 713{:}629.7.063.4{:}621.643.415$



Foreword

This British Standard relates only to features affecting the coupling to an aircraft of the hose unit or adaptor for the delivery under pressure of supplies of aero-engine refrigerant: it does not specify all the requirements for the connections. The standard implements the requirements of ISO 485 "Aircraft water-methanol pressure connections" of the International Organization for Standardization, prepared at the request of the International Air Transport Association as one of a series of standards for aircraft ground servicing connections suitable for adoption on an international basis.

I

This revision enlarges the scope of the standard from methanol-water to aero-engine refrigerant applications and incorporates editorial amendments consequent on the publication of ISO 485. It includes a change of position tolerances on the mating features of the ¾ inch connection to ensure engagement in the extreme adverse tolerance condition and interchangeability with that of ISO 485.

This standard makes reference to the following British Standard:

BS 2856 "Precise conversion of inch and metric sizes on engineering drawings".

NOTE Where metric equivalents are stated the figures in inch units are to be regarded as the standard. The metric conversions are approximate. A table is given in Appendix A to provide a ready means of calculating the approximate metric equivalents of the inch values. More accurate conversions should be based on the table in BS 350, "Conversion factors and tables" or in BS 2856, "Precise conversion of inch and metric sizes on engineering drawings".

A British Standard does not purport to include all the necessary provisions of a contract. Users of British Standards are responsible for their correct application.

Compliance with a British Standard does not of itself confer immunity from legal obligations.

Summary of pages

This document comprises a front cover, an inside front cover, pages i and ii, pages 1 to 4, an inside back cover and a back cover.

This standard has been updated (see copyright date) and may have had amendments incorporated. This will be indicated in the amendment table on the inside front cover.

This British Standard, having been approved by the Aerospace Industry Standards Committee, was published under the authority of the Executive Board on 31 October 1973

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The following BSI references relate to the work on this standard:
Committee reference ACE/14
Draft for approval 72/34767

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Amendments issued since publication

| Amd. No. | Date | Comments |
|----------|-------------|---------------------------------------|
| 2643 | June 1978 | |
| 3326 | August 1982 | Indicated by a sideline in the margin |
| | | |
| | | |

Contents

| | Page |
|---|--------------------|
| Foreword | Inside front cover |
| 1 Scope | 1 |
| 2 Dimensions | 1 |
| Appendix A Table for conversion of inches to approximate millimetre equivalents | Inside back cover |
| Figure 1 — Coupling dimensions and connection detail for ¾ inch aero-engine refrigerant pressure replenishment connection | 1 |
| Figure 2 — Space envelope and connection detail for ¾ inch aero-engine refrigerant pressure | |
| replenishment connection | 2 |
| Figure 3 — Typical ground half view | 2 |
| Figure 4 — Coupling dimensions and connection detail for 1½ inch aero-engine refrigerant pressure | 0 |
| replenishment connection | 3 |
| Figure 5 — Space envelope and connection detail for 1½ inch aero-engine refrigerant pressure | |
| replenishment connection | 4 |
| Figure 6 — Typical ground half view | 4 |

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ii blank

1 Scope

This British Standard specifies the basic dimensional requirements for connections on aircraft to ensure coupling to ground equipment supplying engine refrigerant under pressure.

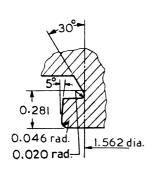
Provision is made for two sizes of connection:

³/₄ inch — suitable for rates of flow up to 25 gallons (114 litres) per minute.

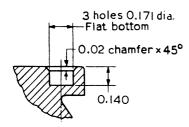
1½ inch — suitable for rates of flow up to 120 gallons (546 litres) per minute.

2 Dimensions

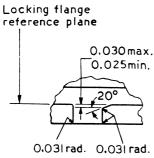
- **2.1** The dimensions of the connection shall conform to the basic requirements shown in Figure 1 or Figure 4.
- **2.2** The clearance allowed in the aircraft structure surrounding the connection for the introduction and handling of the hose unit shall be as indicated in Figure 2 or Figure 5.



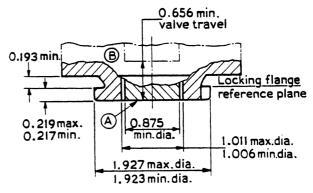
Detail of locking flange



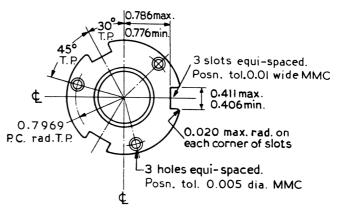
Section of selective pin holes



NOTE 3 slots are identical. Enlarged scrap view in direction of slot



NOTE In this revised drawing, arrow A has been amended to terminate at the face of the valve.



NOTE A The valve face is to be flat within this diameter and no part of the valve may extend beyond this face, but it may be recessed up to 0.030 in to allow for compressibility of valve seal. Configuration of the valve behind this face is optional.

NOTE B $\,$ The valve is to be spring-loaded. Loading at 0.656 in travel is not to exceed 10 lbf (44.5 N) and loading in the valve shut position is to be 3 % lbf to 5 lbf (15.6 N to 22.25 N).

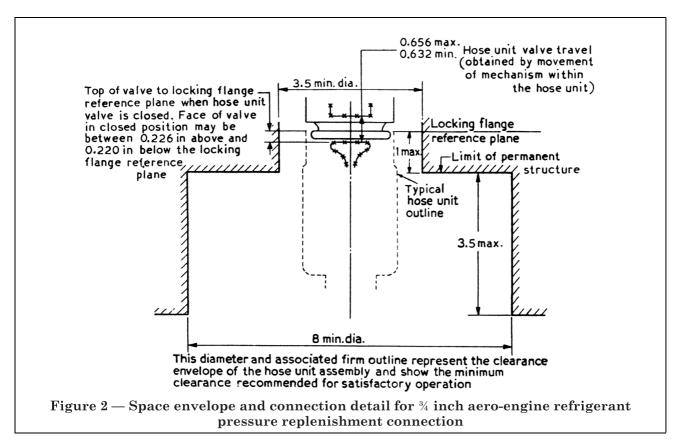
Dimensions are in inches

Dimensional tolerance ± 0.005 in

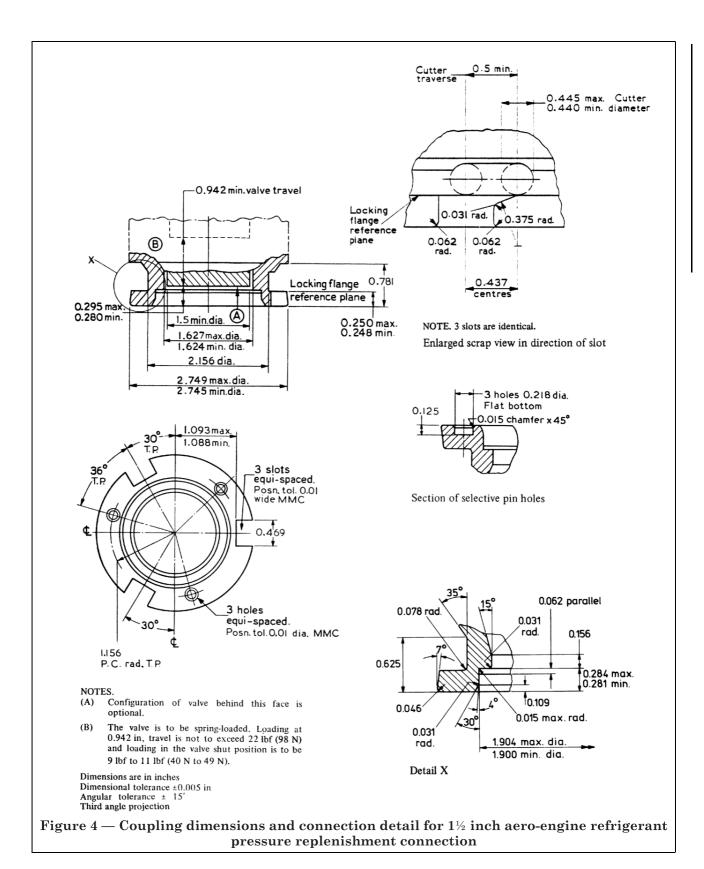
Angular tolerance \pm 15

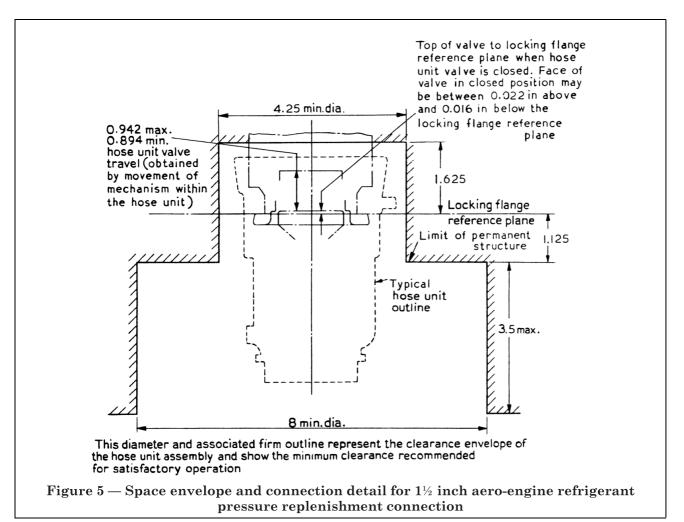
Third angle projection

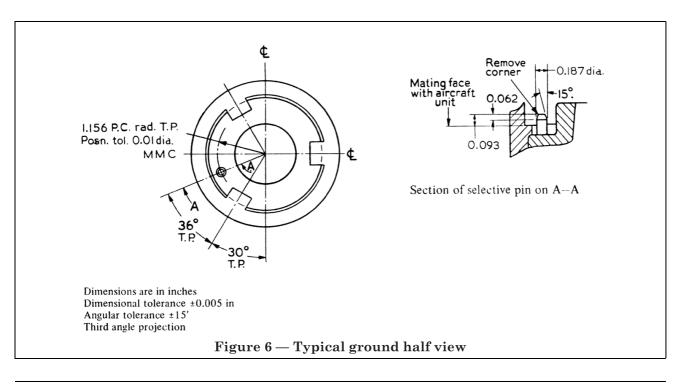
Figure~1-Coupling~dimensions~and~connection~detail~for~%~inch~aero-engine~refrigerant~pressure~replenishment~connection



Remove 0.7969 P.C. rad T.P. corner posn. tol. O.160 max.dia 0.005 dia MMC O.156 min.dia. 0.125 Mating face with aircraft unit 0.047 Section of selective pin on A-A Dimensions are in inches Dimensional tolerance ±0.005 in Angular tolerance ± 15' Third angle projection Figure 3 — Typical ground half view







Appendix A Table for conversion of inches to approximate millimetre equivalents

| in | mm |
|----|-------|
| 1 | 25.4 |
| 2 | 50.8 |
| 3 | 76.2 |
| | |
| 4 | 101.6 |
| 5 | 127.0 |
| 6 | 152.4 |
| | |
| 7 | 177.8 |
| 8 | 203.2 |
| 9 | 228.6 |
| | |
| 10 | 254.0 |

Example of use of table:

$$1^{9}/_{16}$$
 in = 1.56 in = 25.4 + 12.7 + 1.524 = 39.6 mm

$$0.875 \text{ in} = 20.32 + 1.778 + 0.127 = 22.2 \text{ mm}$$

These examples give arithmetic conversions. For conversion to the required accuracy to ensure dimensional interchangeability, use BS 2856.

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