



# Standard Practice for Quality Assurance in the Manufacture of Fixed Wing Light Sport Aircraft<sup>1</sup>

This standard is issued under the fixed designation F 2279; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

## 1. Scope

1.1 This practice establishes the minimum requirements for the development of a Quality Assurance and Production Acceptance Program, to be used for the manufacture of LSA's or LSA kits.

1.2 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.*

## 2. Referenced Documents

2.1 *ASTM Standards:*<sup>2</sup>

**F 2245** Specification for Design and Performance of a Light Sport Airplane

**F 2564** Specification for Design and Performance of a Light Sport Glider

## 3. Terminology

3.1 *Definitions:*

3.1.1 *design and performance specification*—used herein to refer to Specifications **F 2245** and **F 2564**.

3.1.2 *LSA (light sport aircraft)*—used herein to refer to both LSA airplanes and LSA gliders, including kits.

3.1.3 *LSA airplane (light sport aircraft airplane)*—powered aircraft designed in accordance with Specification **F 2245** that is manufactured and delivered ready to fly.

3.1.4 *LSA glider (light sport aircraft glider)*—aircraft designed in accordance with Specification **F 2564** that is manufactured and delivered ready to fly.

3.1.5 *LSA kit (light sport aircraft kit)*—aircraft designed in accordance with Specifications **F 2245** or **F 2564** that is manufactured and delivered as a kit.

3.1.6 *manufacturer*—any entity engaged in the production of a LSA.

3.1.7 *permanent record*—where specified herein, applicable quality assurance records shall be kept for each LSA produced for as long as the relative airworthiness certificate remains in effect.

3.1.8 *reserved holding area—for rejected parts, materials, and assemblies*, shall mean an area for the containment of rejected non-airworthy items awaiting proper disposition, where such rejected items shall not be distributed for use on a LSA.

3.1.9 *satellite manufacturing, assembly, and distribution facilities*—refers to facilities being operated by commercial or private entities that, though authorized by the original manufacturer, are not directly associated with or controlled by the original manufacturer.

3.1.10 *secure storage area—for accepted parts, materials, and assemblies*, shall mean an area of storage where the preservation of the contents to required design specifications is reasonably assured until distributed for use on a LSA.

3.2 *Acronyms:*

3.2.1 *AOI*—aircraft operating instructions

3.2.2 *QAM*—quality assurance manual; the documentation of the Quality Assurance Program

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<sup>2</sup> For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

3.2.3 *QAP*—quality assurance program; the method of inspections used by the manufacturer of a LSA to verify and ensure the proper production thereof

3.2.4 *QAR*—quality assurance record; the record of Quality Assurance associated with each LSA produced

#### 4. Significance and Use

4.1 The purpose of this practice is to provide the minimum requirements necessary for the establishment of a quality assurance and production acceptance program for a manufacturer of light sport aircraft.

#### 5. Quality Assurance Program (QAP)

5.1 Manufacturers of LSA shall develop a Quality Assurance Program (QAP) in accordance with the criteria established within this practice.

5.2 *Quality Assurance Manual (QAM)*—Each manufacturer shall document their QAP in the form of a Quality Assurance Manual (QAM).

5.3 *Quality Assurance Administration*—The manufacturer's administration that is charged with the implementation of the QAP may consist of one or more: company employees, company officials, or manufacturer's agents or assigns. The individual(s) that make up the quality assurance administration shall be identified within the QAM.

5.4 *Quality Assurance Record (QAR)*—A record shall be maintained of the date of acceptance, the origin, and the certifications of materials used in the production of airframe components considered by the manufacturer to be critical to the structural integrity of a LSA (see [Note 1](#)).

**NOTE 1**—The intent of this record is to provide a means for the manufacturer to identify and reduce the number of LSA within a fleet that may be affected by a materials anomaly that would require corrective action, thereby reducing the economic impact of such corrective action. This paragraph should not be construed as a requirement for specific parts traceability.

5.4.1 Manufacturer shall maintain a Quality Assurance Record (QAR) for each LSA produced. Each QAR shall consist of the following:

5.4.1.1 Applicable final inspection records, check, and test documentation from the production acceptance procedures (see [Section 8](#)),

5.4.1.2 A copy of the Manufacturers Statement of Compliance, and

5.4.1.3 The configuration of each aircraft at its point of delivery (for continued operational safety monitoring purposes), including associated parts lists and installed equipment lists.

**NOTE 2**—Each item listed in [5.4.1](#) shall include the LSA serial number and date of manufacture.

5.5 *Quality Assurance Revisions*—A system shall be implemented to ensure that only the latest revisions to the QAM are in use.

5.6 *Quality Assurance Audits*—Manufacturer shall conduct an annual audit of their QAP. Manufacturer shall maintain a record of all such audits. Any determination of non-compliance shall be resolved and a revision to the QAM shall be made if necessary to address any anomalies found.

#### 6. Engineering and Manufacture

6.1 *Record of Compliance*—The manufacturer shall keep a permanent record of the design documentation used to show compliance of a particular configuration to the version of the design and performance specification in effect at the time of manufacture.

6.2 *Configuration Control*—All LSA configurations in production shall have Records of Compliance to the latest released revision of the design and performance specification.

6.3 *Production Documentation*—The manufacturer shall maintain a record of all production documentation, including revisions. Production documentation may include, but is not limited to, the following:

6.3.1 Parts lists,

6.3.2 Process routings,

6.3.3 Component and assembly drawings,

6.3.4 Manufacturing instructions and specifications,

6.3.5 Tooling and gauge drawings,

6.3.6 The AOI,

6.3.7 The maintenance manual, and

6.3.8 The QAM.

6.4 *Special Processes*—A system shall be implemented to control all special processes and services related to the production of airframe components considered by the manufacturer to be critical to the structural integrity of the LSA, such as welding, brazing, heat treatment, plating, structural composites, adhesive bonding, and so forth, that ensures that each process and service is performed in accordance with approved specifications containing definitive standards of quality, and that periodic inspection of gauges, solutions, or any critical equipment is controlled and documented.

#### 7. Quality Assurance Inspections

7.1 Manufacturers shall implement and document a system of inspections to verify conformity of product to all applicable engineering requirements and production specifications.

7.1.1 Conforming, non-conforming, and items awaiting inspection must be separated or clearly distinguishable. Items found to be nonconforming shall either be evaluated by a Materials Review Board (MRB) in accordance with [7.4](#) or rejected in accordance with [7.5](#).

7.2 *Receiving Inspection*—Manufacturer shall implement a purchasing procedure that shall ensure all items ordered are clearly specified. Incoming items provided by outside vendors shall be inspected for conformity to applicable specifications.

7.3 *Acceptance of Conforming Items*—Conforming items shall be distributed as required or placed in a secure storage area for future use.

7.4 *Evaluation of Non-Conforming Items by a Materials Review Board*—A Materials Review Board (MRB) may be established to determine the disposition of non-conforming items, and shall consist of one or more manufacturer designated technical representatives. MRB representatives shall be identified within the QAM. If analysis, additional inspection, functional checks, repair, rework, and so forth assures that an item meets all of the relevant design requirements, the MRB may authorize its use in the production of a LSA. Otherwise,

the item must be rejected in accordance with 7.5. The manufacturer shall keep a permanent record showing the disposition of non-conforming items that have been evaluated and accepted by the MRB.

**7.5 Rejection of Non-Conforming Items**—A process for disposing of items found to be unusable due to damage, shelf life limits, or other variations must be defined and implemented. A rejected item must be mutilated, disposed of, or sufficiently marked as rejected to ensure that it is not used in the production of a LSA. A rejected component may be placed in a reserved holding area for future disposition or disposal.

## 8. Production Acceptance

**NOTE 3**—The following criteria should not be construed as requirements for specific features to be included on a LSA. When a requirement specifies a feature that does not exist on a LSA, the requirement does not apply.

**8.1 Final Inspections**—Manufacturer shall verify and record that a shop order for each LSA produced has been completed prior to conducting the following Production Acceptance procedures.

**8.1.1 LSA Kit**—Manufacturer shall verify and document the proper completion of the production process prior to the further distribution of any LSA kit or subsystem kit. Manufacturer shall provide the builder of a LSA kit with appropriate Production Acceptance Ground Check and Flight Test Procedures, as described below.

**8.1.2 LSA Airplane and LSA Glider**—Manufacturer shall verify the proper completion of the production process prior to the further distribution of any ready-to-fly LSA. The following ground check and flight test procedures shall be conducted and documented for each ready to fly LSA.

**8.1.2.1 Ground Check**—Prior to flight testing, the manufacturer shall conduct a thorough ground inspection of each LSA produced to verify at least the following:

(1) **Weight and Balance**—Empty weight and proper center of gravity location has been calculated and verified to be within limits, and that a weight and balance report has been completed for the airplane;

(2) **Systems Check**—The proper function of all switches and circuits, instrumentation, brakes, and any other appropriate systems shall be verified.

(3) **Flight Controls Check**—All flight controls shall be checked for smooth and proper function and proper maximum deflections. Control system connections and safeties shall be checked and verified intact.

(4) **Seats and Safety Belts**—Seats and pilot restraint system shall be checked for security and visual defects.

(5) **Engine Check**—Engine checks and procedures shall be performed to verify:

(a) Proper engine installation,  
 (b) Proper servicing of all engine fluids,  
 (c) No apparent fuel, oil, or coolant leaks, as appropriate,  
 (d) Propeller installation and pitch adjustment, as applicable,

(e) Performance of an engine “run-in” with adjustments, as required,

(f) Tachometer indicates engine idle RPM and maximum static RPM is within manufacturer’s published limits,

(g) Proper function of engine instrumentation,

(h) Proper function of ignition system(s),

(i) Proper function of induction heating system, and

(j) For powered LSA gliders, proper retraction and extension of the engine.

(6) **Placards Check**—The aircraft shall be checked to verify that all required placards, switch, and instrument markings are in place.

(7) **Preflight Inspection**—The following shall be verified:

(a) All required documentation is on board,

(b) All visible surfaces are free of deformation, distortion, or other evidence of failure or damage,

(c) Inspection of all visible fittings and connections for defective or insecure attachment, and

(d) Complete walk-around inspection in accordance with the AOI.

**8.1.2.2 Taxi Test**—After completion of the Ground Check, a Taxi Test shall be conducted to verify:

(1) Brake function,

(2) Landing gear tracking and steering, and

(3) Proper compass readings, to be verified by a reference, and corrected with compass card if needed.

**8.1.2.3 Flight Test for LSA Airplanes**—After completion of the Taxi Test, a flight test shall be conducted.

(1) Safe flight operation of each completed LSA airplane shall be verified to include acceptable handling and control characteristics, stall characteristics, engine operation, airspeed indications, and overall suitability for normal flight in accordance with the AOI. The flight test procedure, at a minimum, shall include recorded verification of the following:

(a) Takeoff runway wind, outside air temperature, and pressure altitude,

(b) Verification that takeoff distance meets manufacturer’s published specification,

(c) Verification that the climb rate meets or exceeds the manufacturer’s published specification,

(d) Appropriate response to flight controls in all configurations,

(e) Wings-level idle-power stall speed in all configurations, including verification of appropriate stall warning and stall recovery characteristics,

(f) Verification of no unusual performance or handling characteristics, and

(g) Proper engine operating temperatures.

**8.1.2.4 Flight Test for LSA Gliders**—After completion of the taxi test, a flight test shall be conducted. Safe flight operation of each completed LSA glider shall be verified to include acceptable handling and control characteristics, stall characteristics, airspeed indications, and overall suitability for normal flight in accordance with the AOI. The flight test procedure, at a minimum, shall include recorded verification of the following:

(1) Wings-level stall speed in all configurations, including verification of appropriate stall warning and stall recovery characteristics,

(2) Appropriate response to flight controls in all configurations, including, if allowed, which launching and aerotowing, and

(3) Verification of no unusual performance or handling characteristics.

8.1.2.5 In addition, for powered LSA gliders, the flight test procedure shall also include the recorded verification of the following:

(1) Takeoff runway wind, outside air temperature, and pressure altitude,

(2) Verification that takeoff distance meets manufacturer's published specification,

(3) Verification that the climb rate meets or exceeds the manufacturer's published specification,

(4) Wings-level idle-power stall speed in all configurations, including verification of appropriate stall warning and stall recovery characteristics,

(5) Proper retraction and extension of engine, and

(6) Proper engine operation and operating temperatures.

8.1.2.6 *Design Confirmation Flight Test*—For each completed LSA, or by random sampling at a frequency determined appropriate by the manufacturer, and for the first production unit off the production line, an in-depth test flight shall be conducted to verify production uniformity to the flight criteria of the design and performance specification.

8.2 *Instrument Calibration*—Any aircraft instrument requiring periodic calibrations shall have a current calibration.

8.3 *Resolution of Discrepancies*—Manufacturer shall develop and implement a system to correct any anomalies found during ground checks or flight testing.

8.3.1 *Non-Compliance*—Any aircraft which fails any production acceptance test required by this standard shall be physically tagged as non-compliant. Anomalies shall be reworked per manufactures instructions, and each reworked anomaly must be re-evaluated.

8.3.1.1 *Non-Compliance Tag*—A non-compliance notice must be attached to the aircraft in such a manner that it is in clear view of a potential operator of the LSA.

8.4 *Production Acceptance Documentation*—A written checklist may be used as an acceptable method of documenting Production Acceptance inspections, checks, and tests.

## 9. Assignment of QA Duties and Responsibilities

9.1 LSA Manufacturers may assign QA duties and responsibilities to outside parties for the purpose of establishing satellite manufacturing, assembly, or distribution facilities, or a combination thereof.

## 10. Keywords

10.1 LSA; manufacture; production acceptance; quality assurance

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