



Standard Test Method for Thick-Adherend Metal Lap-Shear Joints for Determination of the Stress-Strain Behavior of Adhesives in Shear by Tension Loading¹

This standard is issued under the fixed designation D5656; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

1. Scope

1.1 This test method covers the preparation and testing of thick-adherend lap-shear samples for the determination of the stress-strain behavior of adhesives.

1.2 This test method covers data reduction and analysis of stress-strain curves obtained using thick-adherend lap-shear samples.

1.3 The values stated in SI units are to be regarded as the standard. The inch-pound units in parentheses are for information only.

1.4 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use. Specific precautionary statements are given in 7.3.*

2. Referenced Documents

2.1 *ASTM Standards:*²

[D907 Terminology of Adhesives](#)

[D2651 Guide for Preparation of Metal Surfaces for Adhesive Bonding](#)

[D4896 Guide for Use of Adhesive-Bonded Single Lap-Joint Specimen Test Results](#)

[E4 Practices for Force Verification of Testing Machines](#)

3. Terminology

3.1 *Definitions:*

3.1.1 Many terms in this test method are defined in Terminology [D907](#).

¹ This test method is under the jurisdiction of ASTM Committee D14 on Adhesives and is the direct responsibility of Subcommittee D14.80 on Metal Bonding Adhesives.

Current edition approved Oct. 1, 2010. Published October 2010. Originally approved in 1995. Last previous edition approved in 2004 as D5656 – 04^{e1}. DOI: 10.1520/D5656-10.

² For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

3.2 *Definitions of Terms Specific to This Standard:*

3.2.1 *knee, n*—inflection point on a load-deflection curve; the point at which plastic yielding of the adhesive begins to dominate the deformation response of the load-deflection curve.

3.2.2 *panel, n*—two plates of a rigid material having the same dimensions that are adhesively bonded together on one face such that the thickness of the panel is approximately twice the thickness of each individual plate.

4. Summary of Test Method

4.1 This test method consists of testing thick-adherend lap-shear samples whereby a tensile load is applied to the sample but the adhesive is placed in shear. Unlike Guide [D4896](#), the adherends used in this standard test method are thick and rigid, peel forces are minimized and the shear stress-strain properties of the adhesive are obtained. Referenced data presents a finite element analysis of the thick-adherent metal lap shear specimen and calculates the effect of varying the adherent and adhesive stiffness.³ User of this standard is cautioned that since the stress is not totally uniform across the over-lap of the specimen, factors such as adhesive stiffness, adherent stiffness and location of shear strain measurement along the length of the bond can affect the experimentally obtained load-displacement curve.^{3,4}

NOTE 1—The peel forces generated during testing are a function of the bending deformation in the adherends. Greater adherend stiffness results in less bending and smaller peel forces.

5. Significance and Use

5.1 This test method can be used to determine the stress-strain properties of an adhesive in shear and to establish the proportional-limit of the stress-strain relationship. This data may be useful for the design and analysis of adhesively bonded joints.

³ Kassapoglou, C. and Adelman, John C., "KGR-1 Thick Adherend Specimen Evaluation for the Determination of Adhesive Mechanical Properties," 23rd International SAMPE Conference, Oct. 21–24, 1991.

⁴ Krieger, R. B., Jr., "Stiffness Characteristics of Structural Adhesives for Stress Analysis in Hostile Environment," American Cyanamid Co., Havre de Grace, MD, 1975.

5.2 This test method is not intended to determine adhesion characteristics of an adhesive to a particular substrate; rather this test method is intended to characterize the adhesive shear stress-strain properties that may be relevant for design considerations.

5.3 This test method has been developed and applied using bonded aluminum adherends. At this time no assumptions regarding the validity of this test method with non-aluminum adherends can be made.

6. Apparatus

6.1 Testing machine conforming to the requirements of Practices E4. Select the testing machine so that the breaking load of the specimens falls between 15 and 85 % of the full-scale capacity. Ensure that the machine is capable of maintaining a rate of loading of 2455 N(550 lbf)/min, or, if the rate is dependent on cross-head motion, set the machine to approach this rate of loading. Use a suitable pair of self-aligning grips to hold the specimen. Ensure that the grips and attachments are so constructed that they will move into alignment with the test specimen as soon as the load is applied, so that the long axis of the test specimen will coincide with the direction of the applied pull through the center line of the grip assembly.

NOTE 2—The cross-head speed setting required to approach the specified loading rate is dependent on the modulus of the adherends and the adhesive being evaluated.

6.2 Use two three-point displacement sensors, linear variable differential transformers (extensometers), with the capability of measuring displacements in the adhesive bond overlap area and a sensitivity sufficient to measure displacements with an accuracy of 1 part/1000.

7. Sample Preparation

7.1 Suitably prepare the bonding surfaces of two 229 by 229-mm (9 by 9-in.) by 9.5 ± 0.05 -mm (0.375 ± 0.002 -in.) rigid adherends, such as 2024-T3 aluminum, for the adhesive being used. For aluminum, solvent degreasing and acid etching, such as Method G of Guide D2651, are recommended for removing the oxide coating from the aluminum surfaces to be bonded.

7.2 Apply the adhesive in a uniform layer to one face of one of the adherend plates. In the case of paste adhesives and unsupported film adhesives, metal wires or shims may be used to control the bondline thickness. In this case, take care to ensure that the shims are outside of the lap area or are located in the trim region around the panel perimeter so that they do not affect the test results. Place the other adherend plate on top of the adhesive-coated plate in accordance with Fig. 1. To achieve the desired adhesive thickness, apply sufficient pressure to cause the adhesive to flow between the plates such that the plates are in contact with the shims if they are being used. Cure the adhesive in accordance with the manufacturer's recommended practices for the adhesive being tested.

NOTE 3—The use of glass beads in the adhesive to control the bondline thickness is not recommended, since their use will have an effect on the calculated values of modulus, strain, and strength.

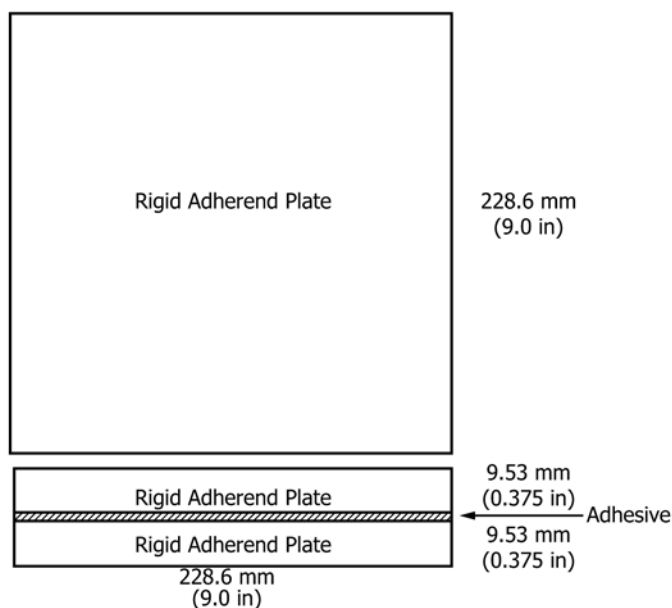


FIG. 1 Bonded Adherend Plates

7.3 After cure, saw-cut 25.4-mm (1.0-in.) wide test samples from the bonded panel, or if machined edges are preferred saw-cut the samples to 27.5-mm widths and machine the saw cut edges to a sample width of 25.4 mm (1.0 in.). Use an end mill to machine notches through the glue line in accordance with Fig. 2. Drill holes for inserting support pins and steel bushings for testing in accordance with Fig. 2. (Warning—Exercise caution during machining of the notches to minimize the buildup of stresses and thermal degradation of the adhesive in this region.) End milling or saw cutting has been found to carry metal from the metal plates and decrease the adhesive bondline thickness in a narrow region along the cut edge. Hand sanding using progressively finer grit sandpaper on a flat sanding block has been found to eliminate this edge effect. The use of 240, 320, 400 followed by 600 grit sandpaper to sand the cut edges, which determine the specimen width, has been effective in removing the metal spread into the adhesive bondline edge.

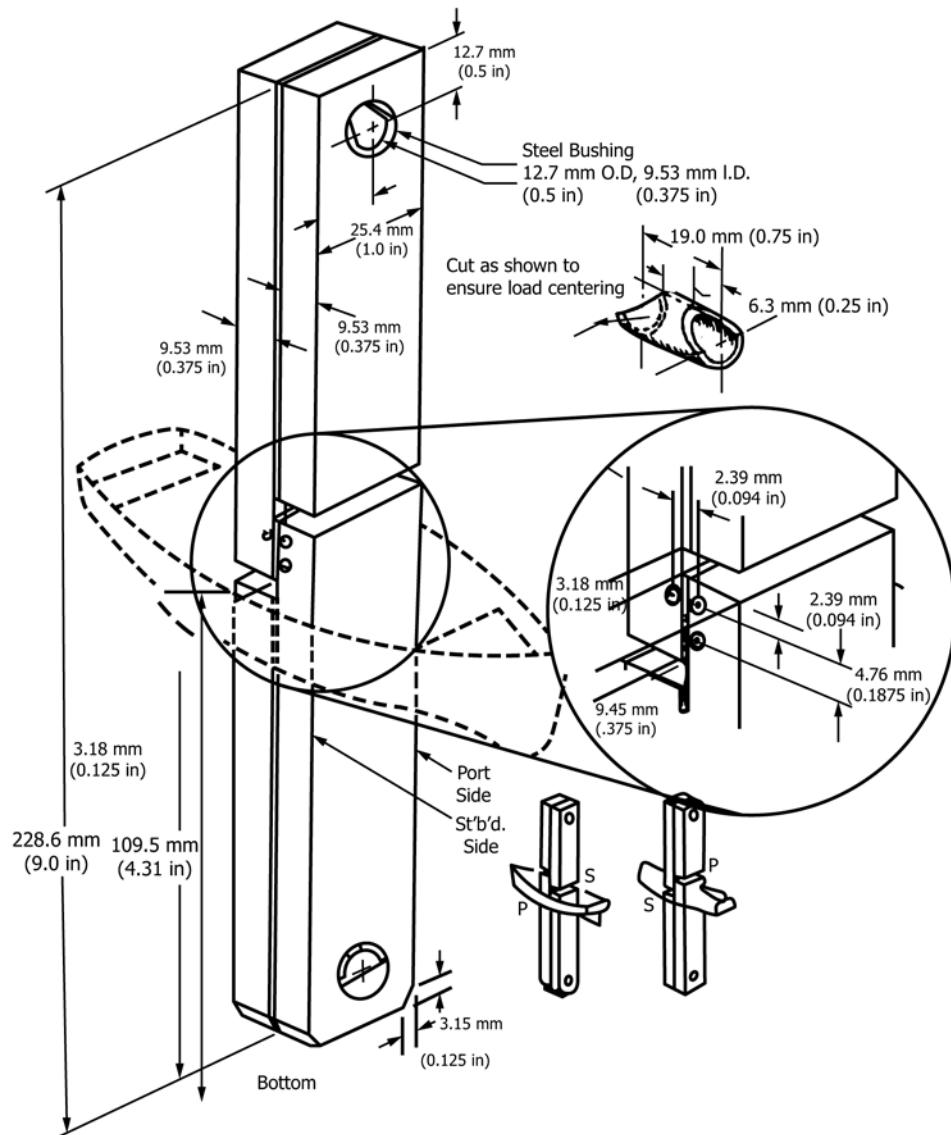
7.4 Using a vernier caliper or other suitable measuring device, measure the width of the overlap area to the nearest 0.03 mm (0.001 in.) for each test sample.

7.5 Using an optical comparator or other suitable measuring device, measure the overlap length to the nearest 0.03 mm (0.001 in.) for each test sample.

7.6 Using an optical comparator or other suitable measuring device, measure the adhesive thickness in the overlap region to the nearest 0.003 mm (0.0001 in.) for each test sample. Due to metal surface roughness start and end the measurement of bond thickness at the average of metal peak to valley distance.

8. Calibration of Extensometers

8.1 Connect the extensometers to an amplifier with two signal inputs and one signal output, which is the average of the two input signals. Connect the amplifier output to a strip-chart



NOTE 1—Asymmetry obviates “left-right” convention. Ship terms retain identity of test site when specimen is rotated or inverted.

FIG. 2 Thick Adherend Lap Shear Test Specimen

recorder that will be used during testing. Allow sufficient time for the amplifier to warm up to minimize signal drift.

8.2 Using a micrometer gage or similar device, adjust the amplifier gain such that 0.0254 mm (0.001 in.) of extensometer displacement is equal to 50.8 mm (2.0 in.) of chart travel.

8.3 Repeat the procedure for the remaining extensometer.

9. Conditioning

9.1 The accuracy of the results of strength tests of adhesive bonds depends on the conditions under which the bonding process is carried out. Unless otherwise agreed upon between the manufacturer and the purchaser, use the bonding conditions prescribed by the manufacturer of the adhesive. In order to ensure that complete information is available to the individual conducting the tests, obtain numerical values and other specific

information from the manufacturer of the adhesive for each of the following variables:

9.2 Procedure for preparation of surfaces prior to application of the adhesive, the cleaning and drying of adherend surfaces, and special surface treatments, such as sanding, that are not specifically limited by the pertinent test method,

9.2.1 Complete mixing directions for the adhesive,

9.2.2 Conditions for application of the adhesive, including the rate of spread or thickness of film, number of coats to be applied, whether to be applied to one or both surfaces, and the conditions of drying where more than one coat is required,

9.2.3 Assembly conditions before application of pressure, including the room temperature, relative humidity, and length of time,

9.2.4 Curing conditions, including the amount of pressure to be applied, the length of time under pressure, method of applying pressure (pressure bag, press platens, etc.), heat-up rate, and the temperature of the assembly when under pressure. Ensure that it is stated whether this temperature is that of the bondline or of the atmosphere at which the assembly is to be maintained,

9.2.5 Conditioning procedure before testing, unless a standard procedure is specified, including the length of time, temperature, and relative humidity.

9.3 A range may be prescribed for any variable by the manufacturer of the adhesive if it can be assumed by the test operator that any arbitrarily chosen value within such a range or any combination of such values for several variables will be acceptable to both the manufacturer and the purchaser of the adhesive.

10. Procedure

10.1 Connect two extensometers to the signal amplifier and the averaged signal output from the amplifier to the strip-chart recorder. Allow sufficient time for the amplifier to warm up to minimize signal drift.

10.2 Set the chart range such that the expected output signals during testing from the extensometer is between 15 and 85 % of full scale.

10.3 Using a solid bar of adherend material machined to the same overall dimensions and geometry as the test sample, attach one of the extensometers to each edge of the bar, in accordance with Fig. 3.

10.4 Mount the bar in the test machine by placing a steel bushing through each of the drilled holes of the solid bar and

inserting steel pins through the bushings and clevises attached to the test machine. Adjust the cross-head position of the test machine to zero, the initial load on the specimen. Load the bar to 4017 N (900 lbf) at a rate of 2455 N/min (550 lbf/min). Repeat the loading six times and record the load versus displacement for each loading.

10.5 For each of the six loadings extrapolate the load-displacement curve to 4464 N (1000 lbf) and record the displacement at this load. Average the results of the six tests to obtain the metal load-displacement values. For 2024-T3 aluminum, 0.00057 mm/1000 N (0.00010 in./1000 lbf) has been found to be a reasonable value for metal deformation at 22°C (72°F).

10.6 Using an adhesively bonded test sample, mount the extensometers on each edge of the sample such that the adhesive bondline is centered between the contact points of each extensometer and each extensometer is centered between the machined notches in the test specimen, in accordance with Fig. 4. Ensure that the single point on the extensometer is on the leg of the coupon that will be raised during testing.

10.7 Mount the test specimen in the test machine by placing a steel bushing through each of the drilled holes of the test sample and inserting steel pins through each bushing and clevis attached to the test machine. Adjust the cross-head position of the test machine to zero, the initial load on the specimen. Load the test sample to failure at a rate of 2455 N/min (550 lbf/min).

10.8 Test a minimum of three replicate samples.

11. Calculation (See Fig. 5)

11.1 Draw a tangent line to the initial, linearly rising portion of the load-displacement curve and label the line as “A.” Label

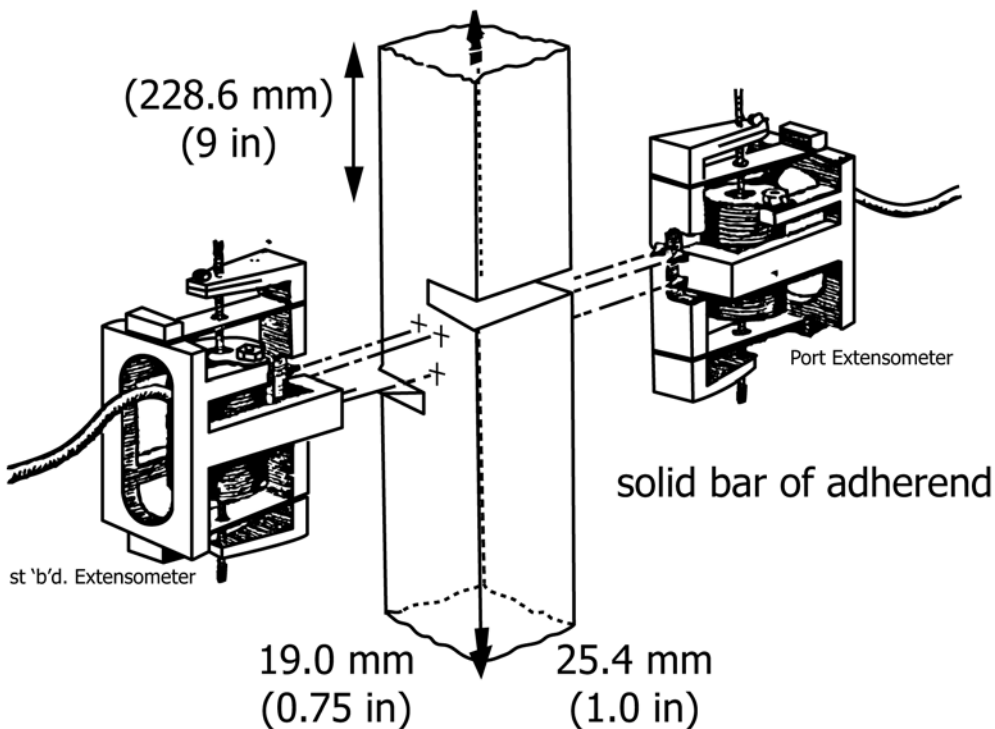


FIG. 3 Extensometer Positioning for Metal Deformation Determination

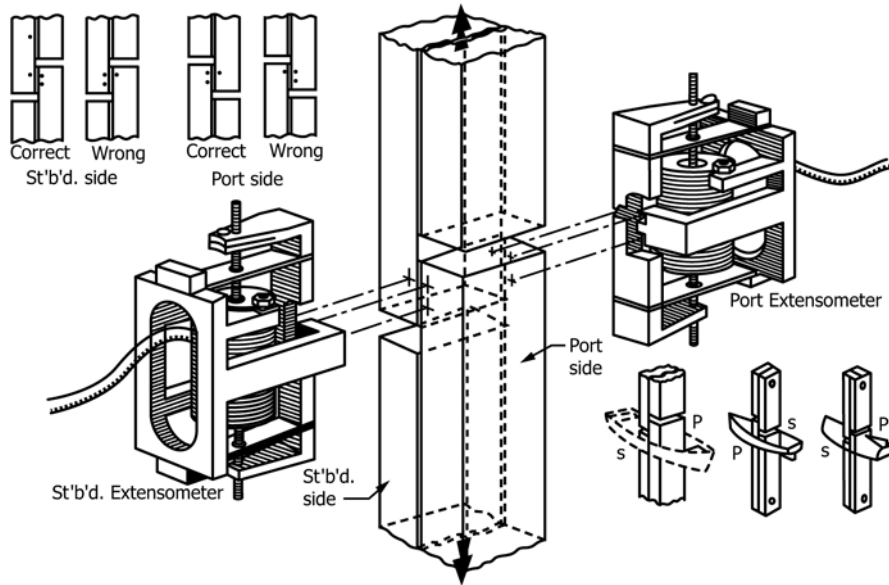


FIG. 4 Extensometer Positioning for Thick-Adherent Lap-Shear Specimen

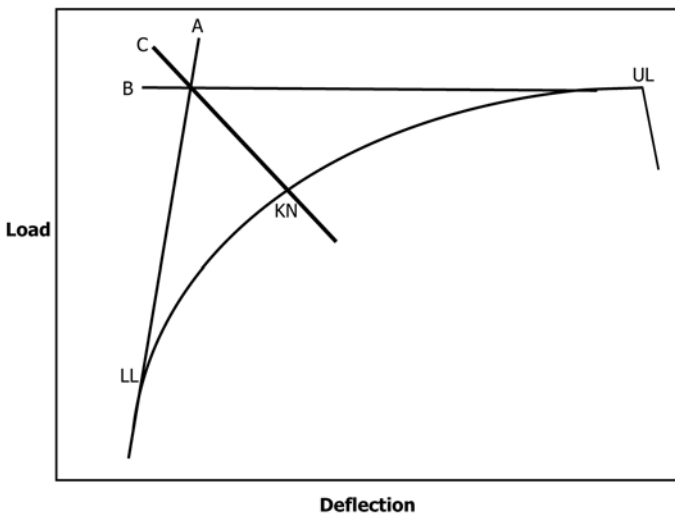


FIG. 5 Load-Displacement Curve

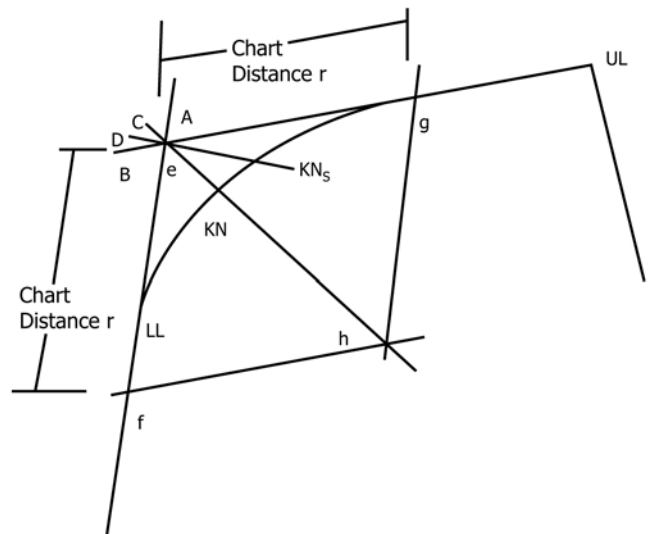


FIG. 6 Shear Stress-Strain Curve for Digital Data

the point at which the load-displacement curve diverges from the tangent as “LL” (linear limit). The calculated stress and strain at LL are δ_{LL} and γ_{LL} , respectively (refer to 11.8 and 11.9).

11.2 Draw a tangent line to the horizontal or yielded portion of the load-displacement curve and label the line as “B.”

11.3 Draw a line through the load-displacement curve, which bisects the angle between the two tangents and label this line “C.” Label the point on load displacement curve as “KN” (knee). If digital data is utilized (see Fig. 6) do not use the calculated slope of the two tangent lines “A” and “B” to determine the bisector of line “D,” instead use an equal chart distance from the intersection point “e” to define points “f” and “g” on lines “A” and “B” respectively. Use the same chart distance from point “f” and “g” to determine point “h.” Calculate a linear equation for a line through points “e” and “h” and label this line “C.” The point of intersection between line

“C” and the calculated shear stress-strain data defines the point “KN” (knee). For FM73 adhesive at 72°F the corrected shear strain at the knee (see 11.12) determined by tangent line slopes “KNs” has been shown to be roughly 6 % were the knee determined by using equal chart distances to determine the knee “KN” is 3 %. An example calculation using digital data is shown in Appendix X1.

11.4 Label the point at which the load drops as a result of adhesive failure as “UL” (ultimate load).

11.5 Calculate the corrected shear strain (γ_c) for at least two points along Line A by subtracting the corrected displacement of the adherend (d_m) obtained from testing of the solid bar, from the displacement measured on the test sample (d_a) at several loads along Line A and dividing by the thickness (t) of the adhesive layer.

$$d_m = \frac{p-t}{p} M \frac{L}{1000} \quad (1)$$

where:

- p = average point gap,
- t = glue-line thickness,
- M = metal displacement at 1000 N or lbf, and
- L = load at displacement d_a .

$$\gamma_i = \frac{d_a - d_m}{t} \quad (2)$$

11.6 Calculate the average shear stress in the bond (δ_i) at each of the corrected shear strains by dividing the load (F_i) by the product of the overlap length (l) and sample width at the overlap (w)

$$\delta_i = \frac{F_i}{l \times w} \quad (3)$$

11.7 Calculate the shear modulus (G_c) as the change in stress along Line A divided by the change in corrected shear strain along Line A:

$$G_c = \frac{\delta_1 - \delta_2}{\gamma_1 - \gamma_2} \quad (4)$$

11.8 Calculate the average shear stress in the bond at the linear limit LL (δ_{LL}) by dividing the load at LL by the product of the overlap length (l) and sample width at the overlap (w):

$$\delta_{LL} = \frac{F_{LL}}{l \times w} \quad (5)$$

11.9 Calculate the corrected shear strain (γ_{LL}) at LL by subtracting the displacement of the metal (d_m), obtained from testing of the solid bar, at a load equal to the load at LL from the displacement measured on the test sample (d_{LL}) at LL and dividing by the thickness (t) of the adhesive layer (see procedure described in 11.5):

$$\gamma_{LL} = \frac{d_{LL} - d_m}{t} \quad (6)$$

11.10 Calculate the average ultimate adhesive stress at ultimate (δ_{UL}) by dividing the load at UL by the product of the overlap length (l) and sample width at the overlap (w):

$$\delta_{UL} = \frac{F_{UL}}{l \times w} \quad (7)$$

11.11 Calculate the corrected shear strain (γ_{UL}) at UL by subtracting the displacement of the metal (d_m), obtained from testing of the solid bar, at a load equal to the load at UL from the displacement measured on the test sample (d_{UL}) at UL and dividing by the thickness (t) of the adhesive layer (see procedure described in 11.5):

$$\gamma_{UL} = \frac{d_{UL} - d_m}{t} \quad (8)$$

NOTE 4—The value of δ_{UL} and γ_{UL} as an adhesive property is only meaningful if the fracture mode of the specimen is cohesive in nature. All other calculations are unaffected by the adhesive failure mode.

11.12 Calculate the average shear stress in the bond (δ_{KN}) at KN by dividing the load at KN by the product of the overlap length (l) and sample width at the overlap (w):

$$\delta_{KN} = \frac{F_{KN}}{l \times w} \quad (9)$$

11.13 Calculate the corrected shear strain (γ_{KN}) at KN by subtracting the displacement of the metal (d_m), obtained from testing of the solid bar, at a load equal to the load at KN from the displacement measured on the test sample (d_{KN}) at KN and dividing by the thickness (t) of the adhesive layer (see procedure described in 11.5):

$$\gamma_{KN} = \frac{d_{KN} - d_m}{t} \quad (10)$$

12. Report

12.1 Report the following information:

12.1.1 Complete identification of the adhesive tested, including type, source, date of manufacture, manufacturer's code numbers, form, etc.,

12.1.2 Complete identification of the adherend material used, method of cleaning, and surface preparations,

12.1.3 Application and cure procedures, and other pertinent conditions used in preparing the specimen,

12.1.4 Adhesive thickness, and bond area dimension including average length and width of the overlap region,

12.1.5 Conditioning prior to testing,

12.1.6 Conditions during testing (that is, temperature, humidity, etc.),

12.1.7 The shear stress in the bond (δ_{LL}) and corrected shear strain (γ_{LL}) at the linear limit (LL),

12.1.8 The shear modulus (G_c),

12.1.9 The average shear stress in the bond (δ_{KN}) and corrected shear strain (γ_{KN}) at the knee (KN),

12.1.10 The average shear stress in the bond (δ_{UL}) and corrected shear strain (γ_{UL}) at sample fracture (UL),

12.1.11 The nature of sample fracture (adhesive or cohesive),

12.1.12 Do not use results obtained using specimen exhibiting the loss of interfacial adhesion to calculate δ_{UL} and γ_{UL} , and exercise caution when using the results to make other calculations that may be impacted by premature specimen failure, and

12.1.13 Determine an average value for shear stress (δ_i), corrected strain (ϵ_i), and modulus (G_c) for each group of specimen tested.

13. Precision and Bias

13.1 Laboratories willing to participate in an interlaboratory study for this standard have been difficult to identify. Subcommittee D14.80 will attempt to initiate an ILS in 2010 with a goal of having it being available by April 2012.

14. Keywords

14.1 modulus; strain; strength; yield

APPENDIX

(Nonmandatory Information)

X1. EXAMPLE CALCULATION OF KNEE SHEAR STRESS AND STRAIN USING DIGITAL DATA

X1.1 Using the average shear stress in the bond as the y-axis and the corrected shear strain as the x-axis plot the data such as shown in Fig. X1.1.

X1.2 The slope of a linear fit to the lower portion of this curve (line A in Fig. 6) is the shear modulus G (19783 psi in Fig. X1.1). The slope of a linear fit to the data above the knee (line B in Fig. 6) is designated U (823.64 psi in Fig. X1.1). To correct for the relative scale difference between the x-axis and the y-axis a scale factor is calculated by printing out a copy of the plot and measuring the number of units of stress per inch of chart along the y-axis and absolute strain per inch of chart along the x-axis, see Eq X1.1 and Eq X1.2.

$$S_x = V_x/L_x \tag{X1.1}$$

$$S_y = V_y/L_y \tag{X1.2}$$

where:

V_x = absolute strain measured for L_x distance along the x-axis,

V_y = number of units of stress for L_y distance along the y-axis,

S_x = scale factor for the x-axis, and

S_y = scale factor for the y-axis.

X1.3 The angle between the lower portion of the graph, line “A” shown in Fig. X1.2, and the x-axis is the inverse tangent of the shear modulus G times the x-axis scale factor divided by the y-axis scale factor, see Eq X1.3.

$$A1 = ATAN(G (S_x/S_y)) \tag{X1.3}$$

X1.4 The angle between the upper portion of the graph, line “B” shown in Fig. X1.2, and the x-axis is the inverse tangent of the slope U and x-axis scale factor divided by the y-axis scale factor, see Eq X1.4.

$$A2 = ATAN(U (S_x/S_y)) \tag{X1.4}$$

X1.5 If we move a distance r radially from the intersection point “e” at (X_o, Y_o) the coordinates for the point “h” at are (X_i, Y_i) are:

$$X_i = X_o + S_x(r(\cos(A2)) - r(\cos(A1))) \tag{X1.5}$$

$$Y_i = Y_o + S_y(r(\sin(A2)) - r(\sin(A1))) \tag{X1.6}$$

X1.6 Calculating the equation of a linear line between (X_o, Y_o) and (X_i, Y_i) and determining the intersection point with the recorded data produces the shear stress, δ_{KN} , and corrected shear strain, γ_{KN} , at the knee.

FM-73 @ 180°F

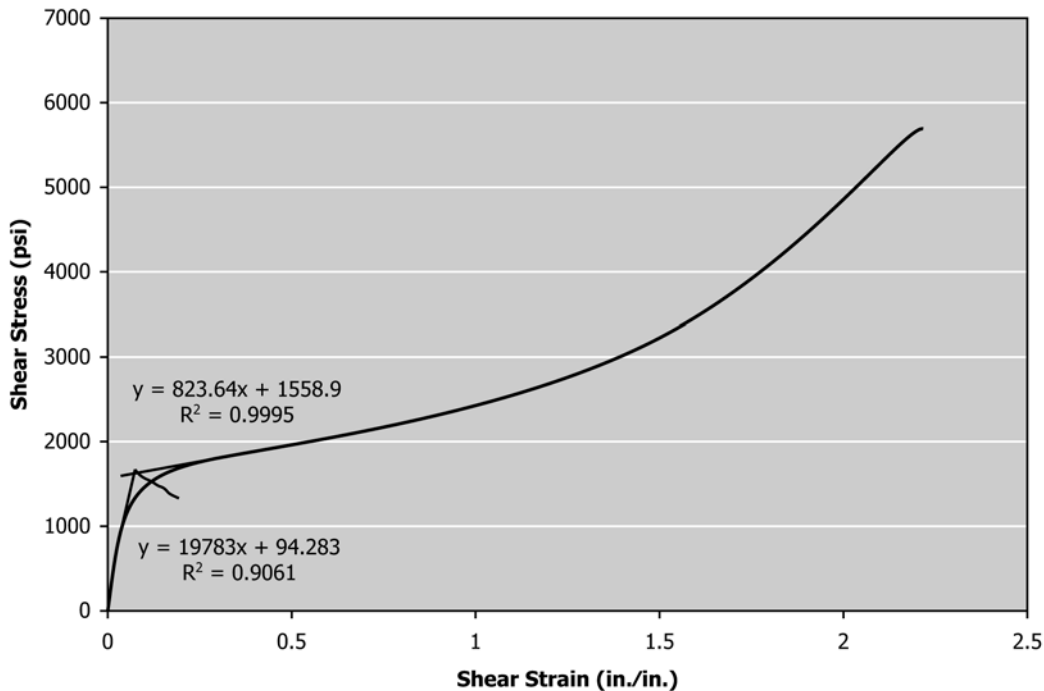


FIG. X1.1 Shear Stress-Strain Curve of FM-73 at 180°F

